

KLE COLLEGE OF ENGINEERING AND TECHNOLOGY DEPARTMENT OF ELECTRONICS AND COMMUNICATION ENGINEERING KLE COLLEGE OF ENGINEERING AND TECHNOLOGY CHIKODI

CONTROL SYSTEMS NOTES (18EC43) (As per Choice based Credit System (CBCS) Scheme)IVTH SEMESTER



DEPARTMENT OF ELECTRONICS AND COMMUNICATION ENGINEERING KLE COLLEGE OF ENGINEERING AND TECHNOLOGY CHIKODI

"Don't see others doing better than you, beat your own records every day, because success is a fight between you and yourself"

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	MODULE 1 Introduction to Control Systems. Types of Control Systems. Effects of feedback Systems. Mechanical Systems (Translational Systems). Problems to be solved on Translational Systems. Solved Problems on Translational Systems. Practice problems on Translational Systems. Assignment on Translational Systems. Rotational Systems Problems to be solved in Class on Rotational Systems. Solved Problems on Rotational Systems. Solved Problems on Rotational Systems.

MODULE-1

Syllabus:

- Introduction to Control Systems:
- > Types of Control Systems
- Effect of Feedback Systems
- Differential equation of Physical Systems
 - Mechanical Systems,
 - Electrical Systems, and
 - Analogous Systems.

Controls Systems Notes

Introduction to Control System :-Control System: - It is an arrangement of different physical Components such that it gives the desire Dutput for the given input by means of regulate or Control Rither direct or indirect method. Plant: It is defined as the postion of a system which is to be controlled or regulated. It is also called as Process. Controlles: It is an Element of the System itself or external to the System. It Controls the plant may be or the process. Input! The applied Signal or Exception Signal that is applied to a Control System to get a specified Supput is called. Output: The actual response that is Obtained from a input. Control system due to the application of the loput is termed ous output Control Sutput. System Types Of Control Systems:-* Open - loop Control System : Controlled output Control Signal > Reterence Controller Plant * Open - Loop Control System are Control Systems in which the output has no litect upon the control active.

In such systems, there is no measurement of the Output and no-subsequent use of that Output to generate any Control action. * It is kimple to Construct and Easy to maintain. Advaptages !-* It is less Espensive because of the use of minimum * The problem of instability does not Exist. * It is able to perform accurately once the calibration of the input its doze. Dis. advantages: * Disturbances, internal or External, Causes drifts in the desired Sutput. * Changer in Calibration Cause Errors in the system. * Re- Calibration of the System may be necessary from time to - time in swaler to main the required quality of the Dutput. Closed - Loop Control System: Controlled Control Signal Plant Reference + Controller output. roput Feedbaus DELWOOKS

* A cloped - loop Control System 14 and that rocasures its Sutput and adjusts its isput accordingly by using a feed back Signal. * Feed backs networks Consists of puggive elements like R, L, C Which its been used to feed backs the Obtained Output to the Popul. * The Controller Subsequently produces the necessary Control Signal, Which is then applied to the plant or process to reduce the Propos and bring the Dutput of the System to the depired value. In closed loop system the junch manity of the system depends on the difference between input and the * feed backs. * A cloped loop Constrol System is also called as feedborg Control Bystom. * Relatively more accurate and Expensive. Components may be used to Obtain an accurate Control of a given process. The influence of internal and External disturbances On the Output can be made almost in effective. * Transient response of the system (an be improved. * Steady - state Error can be reduced.

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Dis ad	Dis advantages:-			
 * It requires more Equipment and Components and it is costurer * These is a tendeney to Over Correct Errors, which * These is a tendeney to Over Correct Errors, which may create Iscultations in the System Dutput. This may Cause the System to drift to instability. 				
Companission between Open-loop and Cloped-Loop Control				
System.				
SUNO	Open-Loop Control System	Cloped Loop Control System		
1.	The Open-loop System are to imple to construct and cheap.	The closed Loop System are Consplicated to Construct and Costly.		
2.	It consumes less power	It consumes more power.		
з.	Any Change in the Output has no litteet On the input i.e., feed back, cloer not Exists	Changes. is the Dutput, affects the isput which is possible by use of feed backs.		
ч.	Highly sensitive to the disturbances	cless sensitive to the disturbances.		
5.	It is inaccurate and un- reliable	Highly accurate and reliable.		
6.	The open-loop systems are generally stable.	More care is required to design a stable System		
4.	Highly Sensitive to the Environmental Charges	Less servicitive to the Environmental changes.		

of an Open- Loop system :-Hpplication 15 Constrol action of a Celing - fan regulator: Actual Voltage specal Regulator Fas leaves Fan motor Controlled Ret Sutput. process controller Actuator 1/P BLOUS diagram of Fan Regulator A fan regulator in a combination of a Series Switch and a specid regulator. For a given speed setting on the fan regulator, the fan runs at a specific speed. To Obtain a different speed the setting On the regulator is to be changed This changes the Voltage applied to the motor the specific setting the the regulator is the reference input and the Variable Voltage applied to the fan motor is the Control Signal. The speed of the fan is the Controlled Output. the Output speed is not measured so the Control Scheme is Open- wep The block diagram is shown in the figure above. 25 Working of an automatic Washing machine: Control Controlled signal Ret Output Tank, Water input Clother Timer > Detergents, Actual Preset motor cleanliness time Process Controller Actuation of clothes. Bloug diagram of Automatic Washing Machine.

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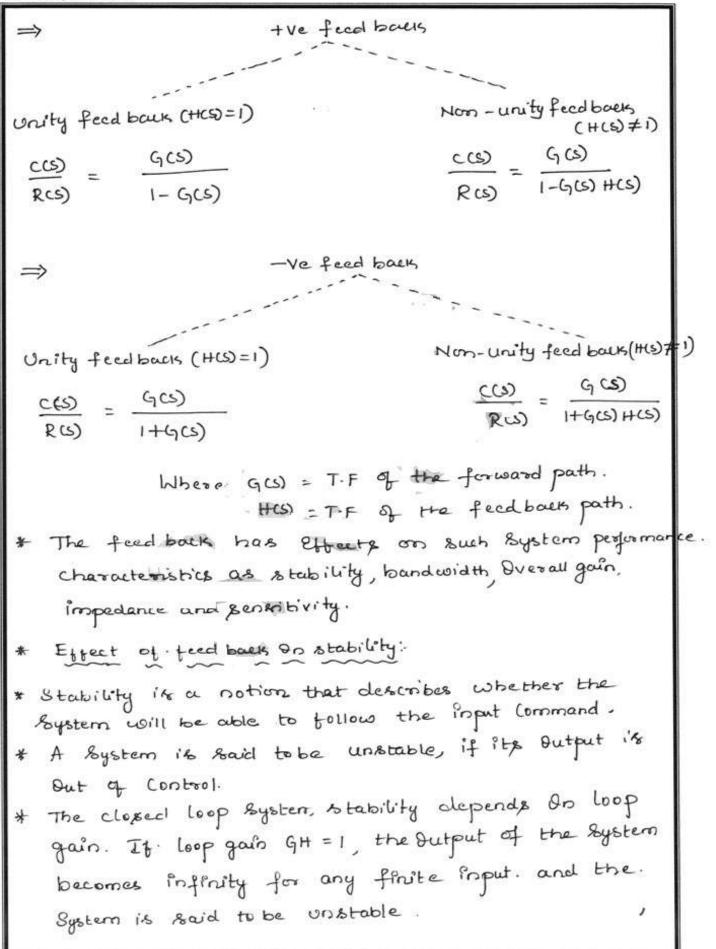
* An automatic Washing machine is one with present wash times. The different operations like sourcing, washing, rinking, wringing and drying are all Performed on a time basis After the clothes to be washed are kept inside * the tank, water and aletergent are added in Proper amounts The times and the relays act as the Controller. The tank, water, detergent and the motor (onstitute * the actuator. The clothes that are to be washed from the Process. These is no measurement of quality of * wash. If clean liness of clothes is the output Parameter, these its no mechanism to judge. it. The system is thus open-loop. The block cliagram in shown above. Applications of an closed loop Costool System:-1) Control Scheme: in the manual steering system of an Disturbone Sping actual auto mobile; Auto mobile detection Nervous Control direction Ret Hand and Signal > body and input Steering Depired Y- Error System wheel wheel direction process Actuator Control Unit Measured direction Eyes feed bouts writ (Automobile Steering Control)

* The driver of an automobile watches the direction or boading of the Vehicle with respect to a specified direction of the road. The Eye senses the deviation, if any, and the information is fed to the brain, through the nervous System. * The brain processes the signal and generates a Connective signal and it is transmitted to the hand and then to the steering wheel. The direction of the road is the reference input and. beading of the automobile is the controlled Poput. * The human lye acts as the Error detector. The brain and the nervous bystem arete as the Componiators and the Controller. The Corrective Solenal generated in the brain is the Control Sorgnal which is transmitted to the hands. * The human hands and the steering wheel from the actuators unit, The automobile body and the. wheels from the process. The scheme represents a manual feedback system and it is shown in the figure above. 25 The control scheme for mixing Cold and not water whale taking a shower. Hotwater Actual temp (ontro) 8-1000 Geference Shower + brain Nervous Signal Hand, Valves input + (optrolle water Pipes & Taps output System. Depired process temperature Actuator Controller feedback unit Sensed Temperature SKin (BLOWS Diagram of mixing not and cold water)

* One must have knowe idea of the Water temperature. he/she wants while taking a shower. The skin acts as a temperature sensor, which measures the temperature not quantitatively but qualitatively, and Conveys the information to the brain. These. it is compared with the water temperatures the. Person desires. the brain computes the differences in terms of 'too cold' or 'too hot' and activates the hand muscles to manipulate the hot and cold water Valves to reduce the temperature if it is too hot or increase the temperature if it is too cold. This corrective action is reciprocal Until the required. Water temperature is achieved. The block diagram is shown above.

Feedbacks Systeric. R(s) G(s) C(s) H(s)

* If Proves Signal ect) is Zero, Output is Controlled.
* If Proves Signal ect) is not Zero, Output is not Controlled.
For positive feedback, Porce Signal = rct) + cct)
For Negative feedback, Proce Signal = rct) - cct)
* The purpose of feedbacks is to reduce the Error between the reference input and the System.
Output.



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DM/M and DG/G denote the percentage . Change in Mand G respectively. $S_{G}^{M} = \frac{\partial M}{\partial G} \times \frac{G}{M} = \frac{1}{1+GH}$ This relation shows that the Sensitivity function Can be made arbitrarily Somall by Preseasing GH, Provided that the system remains stable. In an Open-loop System, the gain of the System will respond in a One-to-One fashion to the Vaniation in 'G'. In general, the sensitivity of the system gainz of a feedback System to parameter Variations depends on where the parameter is located. => Effects of feed backs (In - Brief) * Gain is reduced by a faster * These is reduced of parameter Variation by a factor 1+G(5)+(5) * There is improvement in sensitivity. * These may be reduction of stability. The disadvantages of reduction of gain and reduction of Stability can be over come by gain amplification and good design respectively.

* Feedbacks reduces the Effect of noise and disturbance On system performance. * Band width increases by the factor of 1+9(5) H(5) * The System becomes more accurate.

Mechanical Systems:-Mechanical System are broadly classified into two. groups. 1) Translational System. 2) Rotational System Translational System: In translational System, the motion of the body is along a straight line Rotational System: In Rotational System, the motion of the body is about it's Dionaxis. (Circular Path). * The Basic Elements of translational System are. 1 Mass (2) Spring Dashpot or friction. (Mass: Mass is the Energy (Kinchic) storage elements, where Energy can be stored and retrieved without loss. →x(+) (Displacement) i→U(+) (Velocity) - M - F(t) or F FM When a force FCt) is applied on the mass, it opposing force Fm and it is given by Produce an Fmaa Fm = Ma $F_{M} = M \frac{du(t)}{dt} = M \frac{d}{dt} \left(\frac{dxt}{dt} \right)^{2}$

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$$F_{M} = M \frac{d^{2} x(t)}{dt^{2}}$$
Where
 $x(t)$ and $u(t)$ are the displacement and the
velocity respectively.
M is the mass, the force due to acceleration
is given by
 $F_{M} = M \frac{d^{2} x(t)}{dt^{2}}$
At Equilibrium according to person is laws $F(t)=F_{M}$
 $\therefore F(t) = M \frac{d^{2} x(t)}{dt^{2}}$
 $M = M \frac{d^{2} x(t)}{dt^{2}}$
At Equilibrium according to person is laws $F(t)=F_{M}$
 $\therefore F(t) = M \frac{d^{2} x(t)}{dt^{2}}$
 $M = Mass is an incorrhial element it stores energy in
the form of Kinotic energy is given by.
 $K = W = \frac{1}{2} mu^{2} J = -0$
inductance:
 $I(t) = Henry$
 $V(t) = L \frac{d}{dt}^{2} - 3$
 $V(t) = L \frac{d}{dt^{2}} - 3$
 $W = \frac{1}{2} L I^{2} J - 9$$

ł

Two Systems are baid to be analogous if the Mathematical Equations of the two Systems are identical 1 FC+) = V(+) and M = L then x(t) = q(t) or u(t) = i(t)When force is compared with Voltage the Corresponding Electroic Circuit is social to be force - Voltage analogoury Chrcuit : Electrical System Mechanical F-V Qualogy System Voltage Force current Velocity Displacement charge inductance. Mans Capaci tance : $i(t) = C \frac{d V(t)}{dt}$ SC+2 but Uct) = d + (+) , $i(t) = C \frac{d^2 \phi(t)}{d^2 \phi(t)} - (t)$ i(+) $I_{\lambda J} = \frac{1}{2} (V^{2}(+) J - G)$ Comparing Equations () and () they are analogous if F(+) = i(+)M=C Then $x(t) = \phi(t)$ (+) = 0(+)

* Mass has only one displacement. * Counter force produced by the mass is proportional to second derivative of displacement. When force is compared with voltage inductance * is the electrical analog for Mass and When force. is compared with current Capacitance is Electrical analog for mass.

25 Spring: Spring is the low gy (Potenbial) Storage.
Element (where low gy can be stored and retrieved
Without low.
(i) When one End of Spring is (connected to the reference.
K
$$\rightarrow$$
 Scit)
Reference F_{K} .
For a linear spring (counter force produced by the
sepaing is proportional to net displacement of the spring.
 $F_{K} \propto (\chi(t) - 0)$
 $F_{K} \equiv K \chi(t)$
Where K is the (constant of Proportionality
Known as the spring Constant.
At Equilibrium, according to Nauton law.
 $F(t) = F_{K}$
 $F(t) = K \chi(t) = K (U(t) dt) = 0$
(apacitonce:
 C_{F}
 $V(t)$
 $V(t)$
 $V(t) = \frac{Q(t)}{C} = \frac{1}{C} \int i(t) dt - 2$

Controls Systems Notes

Comparing Equation
$$0 \neq 0$$
 They are analogous
if $F(t) = U(t)$
 $K = \frac{1}{c}$
 $\chi(t) = q(t)$
 $U(t) = i(t)$
Inductance:
 $U(t) = i(t)$
Inductance:
 $U(t) = \frac{1}{c} (\psi(t) - (z)) \qquad (y(t)) = \frac{d}{dt} \phi(t)$
Equations 0 and (a) are analogous if
 $F(t) = I(t)$
 $K = \frac{1}{c}$
 $\chi(t) = \phi(t)$
 $U(t) = U(t)$
iii When both Ends of the spring are free to move.
 $i \rightarrow \chi_2(t) \rightarrow \chi_1(t)$
 $i \rightarrow \chi_1(t) \rightarrow \chi_2(t)$
 $i \rightarrow \chi_1($

* If Droe End of the Spring its Lonsneeted to the reference
it has one displacement and if both ends are tree
to move it has two displacements.
* Counter force produced by spring its proportional
to pet displacement of the sporting
* When force its Compared With Voltage, Capacitance
is the electroical analog. When force is compared With Current, inductance is the electroical analog for
the Mechanical Element Spring.
34 Dashpot or Friching :-
* The froiction laists PD physical Systems Whenever Mechanical Surfaces roe Aperated in Schiding Contact.
* There are 3 types of troichor trong
as Coulomb friction force: This is the force of bliding friction between dry Susfaces, Coulomb froiction force
is kubstastially constant.
by Viscous friction force: It is the friction between moving buspaces separated by a Viscous fluid or
between a bolid body and a fluid medium. it is
Proportional to the velocity. It is pict
14 Stiction :- This is the force required to initiate motion between two Contacting Surfaces.

Viscous friction : Capeir When one End of the doshpot it connected to. the reference. BorD $\rightarrow \mathcal{L}(t)$ 11111111 $\rightarrow u(t)$ F(t) ь annan Piston FB Reference Cylinder BODD F(t) FB The Couster force produced by the Dasspot i's

The Counter force produced by the Dashpot 1's Proposional relative velocity between piston and the Cylinder.

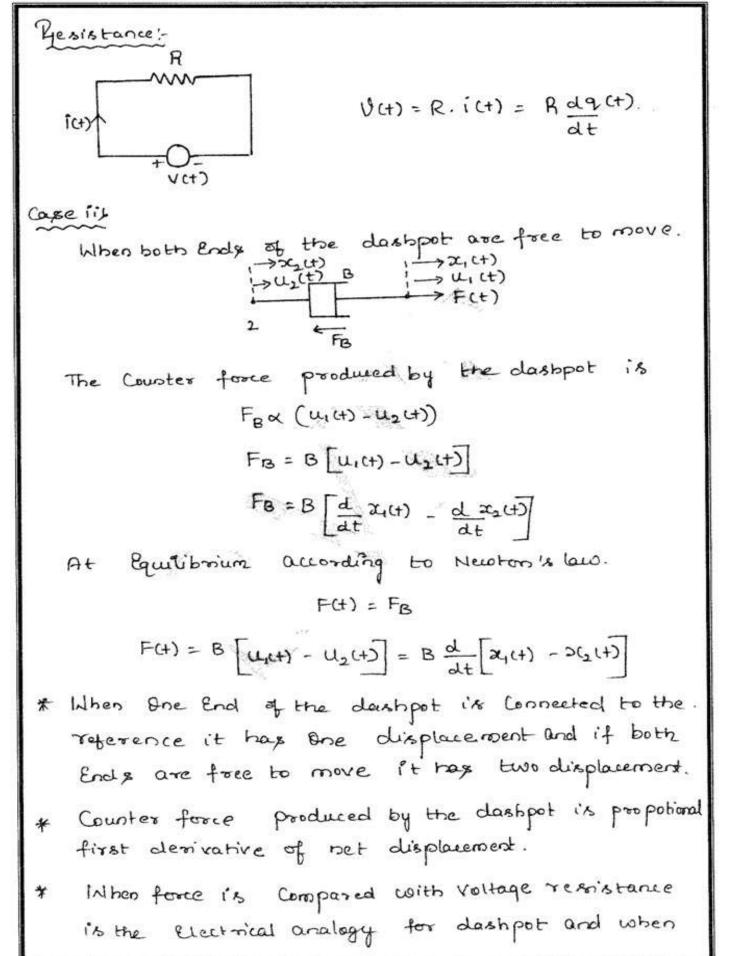
$$F_{B} \propto (u(t) - 0)$$

 $F_{B} = Bu(t) = B \frac{dx(t)}{dt}$

B is the constract of propotionality known as the viscous friction Co-Efficient. At Equilibrium according to Newton law [Fct) = FB]

$$F(t) = Bu(t) = B \frac{dx(t)}{dt} - 0$$

Controls Systems Notes



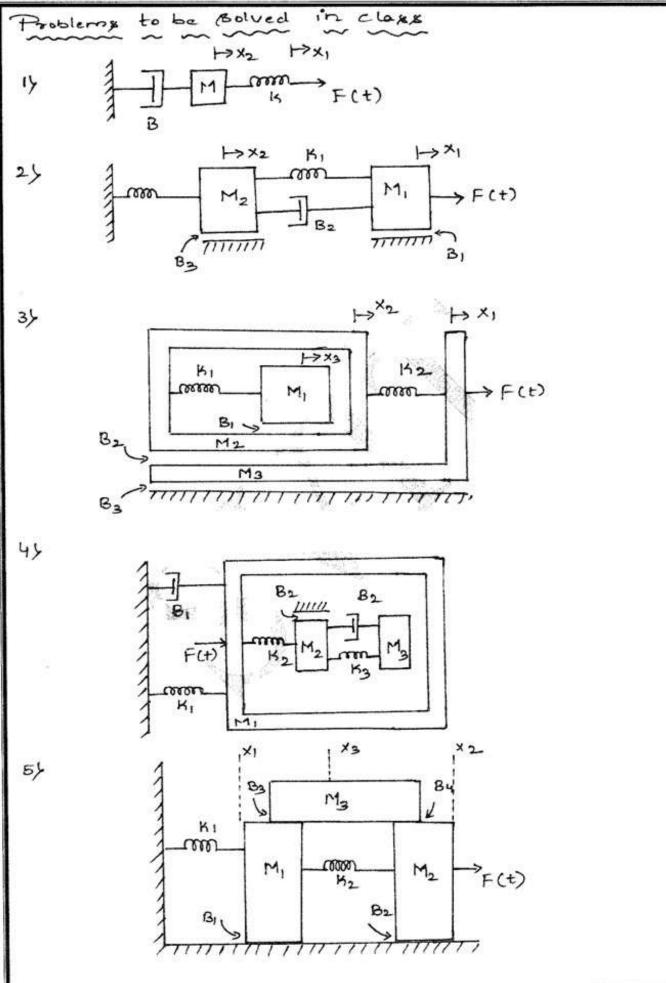
force is compared with current Conductance is the Electrical analogy for the Mechanical Element dashpot. $F(t) = B u(t) = B \frac{d x(t)}{dt}$ $V(t) = Ri(t) = R \frac{d}{dt} q(t)$ $\dot{L}(t) = \frac{\varphi(t)}{\varphi} = \varphi(\mathbf{u}(t)) = \varphi(\frac{\varphi(t)}{\varphi(t)})$ for Converting Mechanical System to. labulation Force-Voltage and Force - Current analogy. Electrical System Mechanical F-V Analogy F-I Analogy System Voltage V(+) Current I(+) Force F(+) Voltage V(+) Velocity [uc+) Current I(+) Displacement Secto Charge Qcto Magnetic Flux put Inductance L Capacitance C Mass M Acciprocal of Capacitonic Reciprocal of Spring Constant [K] inductance $\left|\frac{1}{c}\right|$ 1-Compliance [1] Gapacitance [c] Inductance [L] Dash pot Constant Resistance R] Conductance G] B

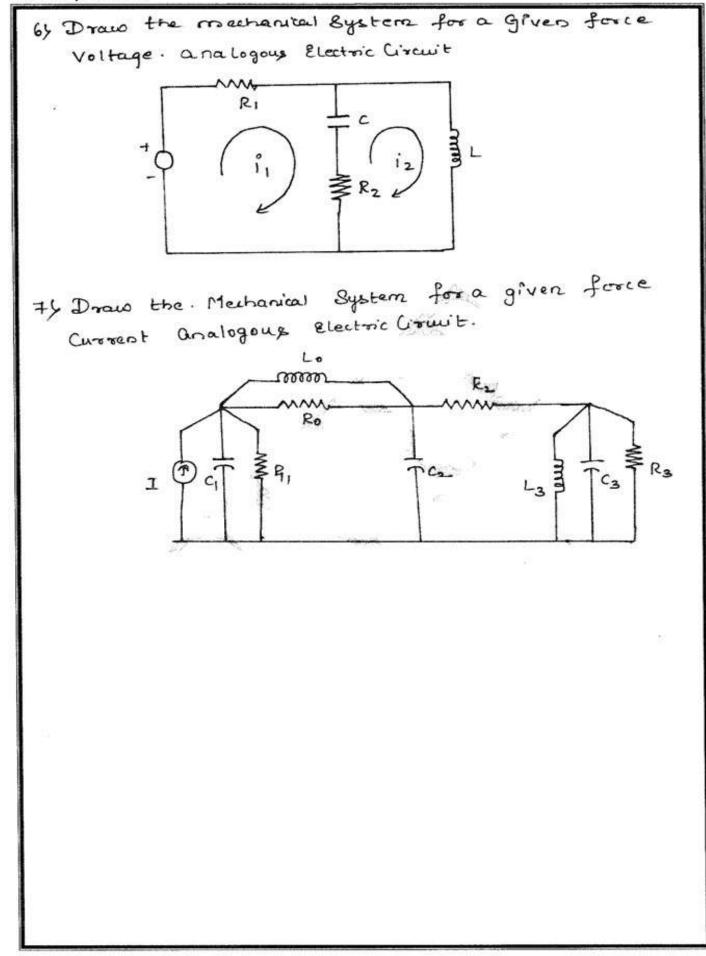
Things to remember F-V Analogy: $\Rightarrow \frac{dq(t)}{dt} = i(t)$ ⇒ q(t) = | i(t) dt F-I Analogy: $\Rightarrow de^{(+)} = V^{(+)}$ => \$ \$ (+) = {V(+) dt.

Note:

- * If the force directly acts On the mass, then the. number of displacement is Equal to number of masses in the system, provided there is no direct series Connected of two springs or two dashpot or dashpot and a spring.
- * If the force directly acts on the spring or the dashpot then the number of displacement in the. Mechanical System is Equal to Durbles of marses +1. Provided there is no direct series Connection of two spring or two dashpot and spring and a dashpot.

Controls Systems Notes

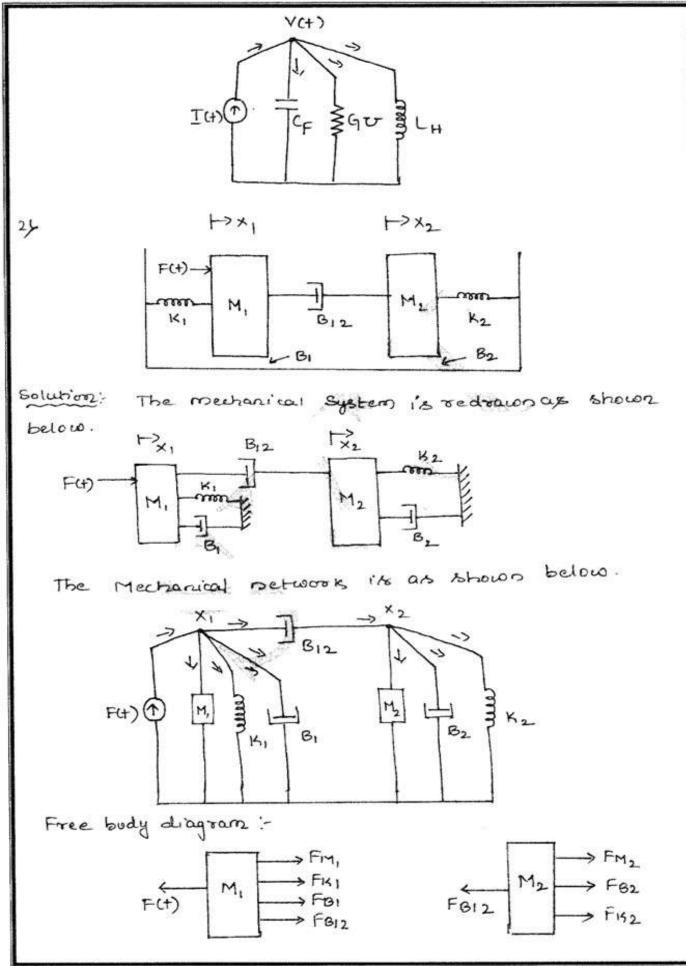




Problems Do Translational Systems: => Write differential Equations for the mechanical Systems shows below and also Electrical analogous Circuit based on force Voltage analogy or force Current (+>x(+) analogy. Dec 2009 K 14 → F(+) MT B Solution: The mechanical metworks for a given mechanical. system is as shown below x 47 F(+) 0 KAN M 7B mmminnin mmm The free body diagram is as follows FM $F_{k} \longleftarrow M \longrightarrow F(+)$ FRE The Equilibrium Equation of a system are FC+2 = FM + FB + FIS $F(t) = M \frac{d^2 x(t)}{dt^2} + B \frac{d x(t)}{dt} + K x(t) \rightarrow 0$

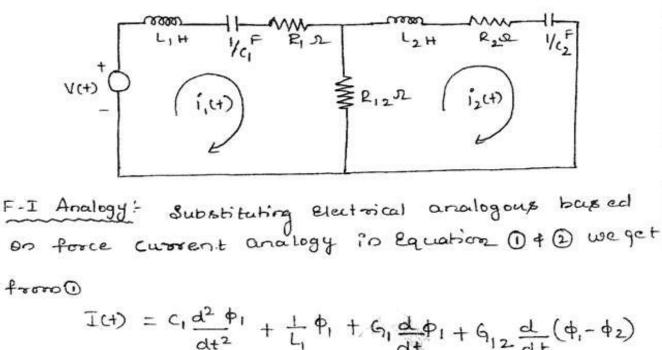
F-V Analogy: Substituting the Electrical analogous based on force voltage analogy is Equation ($V(t) = L \frac{d^2 q(t)}{dt^2} + R \frac{d q(t)}{dt} + \frac{1}{c} q(t)$ $i(t) = \frac{dq(t)}{dt} + \int i(t)dt = q(t)$ But $V(t) = L \frac{d}{dt} (t) + Ri(t) + \frac{d}{dt} \rightarrow @$ The Electrical Circuit Satisfyfrig Equation () is as shown below. LH(M) $(F-(+)) \vee (+) \begin{pmatrix} i \\ i \\ i \\ k \end{pmatrix} = \begin{pmatrix} i \\ i \\ k \end{pmatrix} = \begin{pmatrix} i \\ k \\ k \end{pmatrix} = \begin{pmatrix} i \\ k \\ k \\ k \end{pmatrix}$ F-I Analogy: Substituting the electrical analogous based on force Current analogy in Equation () we get. $I(t) = C \frac{d^2 \phi(t)}{dt^2} + G \frac{d \phi(t)}{dt} + L \phi(t)$ But $V(t) = \frac{d}{dt} \frac{\phi(t)}{\phi(t)} \phi(t) dt = \phi(t)$ IC+) = C d V(+) + G V(+) + L JV(+) dt - 3

The electrical Circuit Satisfying Equation (3) is



The Equilibrium Equations are given by .
At
$$x_1$$

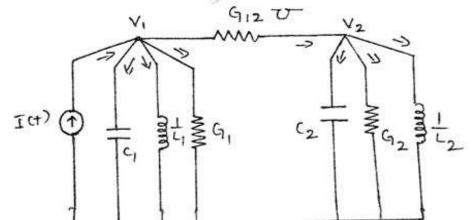
 $F(t) = FM_1 + FK_1 + FB_1 + FB_{12}$
 $F(t) = M_1 \frac{d^2 x_1}{dt^2} + K_1 x_1 + B_1 \frac{d x_1}{dt} + B_{12} \frac{d (x_1 - x_2)}{dt} \rightarrow 0$
At x_2
 $FB_{12} = FM_2 + FB_2 + FK_2$
 $B_{12} \frac{d (x_1 - x_2)}{dt} = M_2 \frac{d^2 x_2}{dt^2} + B_2 \frac{d x_2}{dt} + K_2 x_2 \rightarrow 0$
 $F-V$ Analogy:
Substituting electrical analogous based in Force.
Voltage analogy in Equations $0 + 0$
from 0
 $V(t) = L_1 \frac{d^2 G_1}{dt^2} + \frac{1}{C_1} + R_1 \frac{d q_1}{dt} + R_{12} \frac{d (q_1 - q_2)}{dt}$
 $V(t) = L_1 \frac{d d q_1}{dt^2} + \frac{1}{C_1} \int f_1 dt + R_1 f_1 + R_{12}(f_1 - f_2) \rightarrow 0$
from 2
 $R_{12} \frac{d (q_1 - q_2)}{dt} = L_2 \frac{d^2 q_2}{dt^2} + R_2 \frac{d q_2}{dt} + \frac{1}{C_2} \int f_2 dt \rightarrow 0$
The electrical Circuit Satistying Equations () and
() its as shown below.



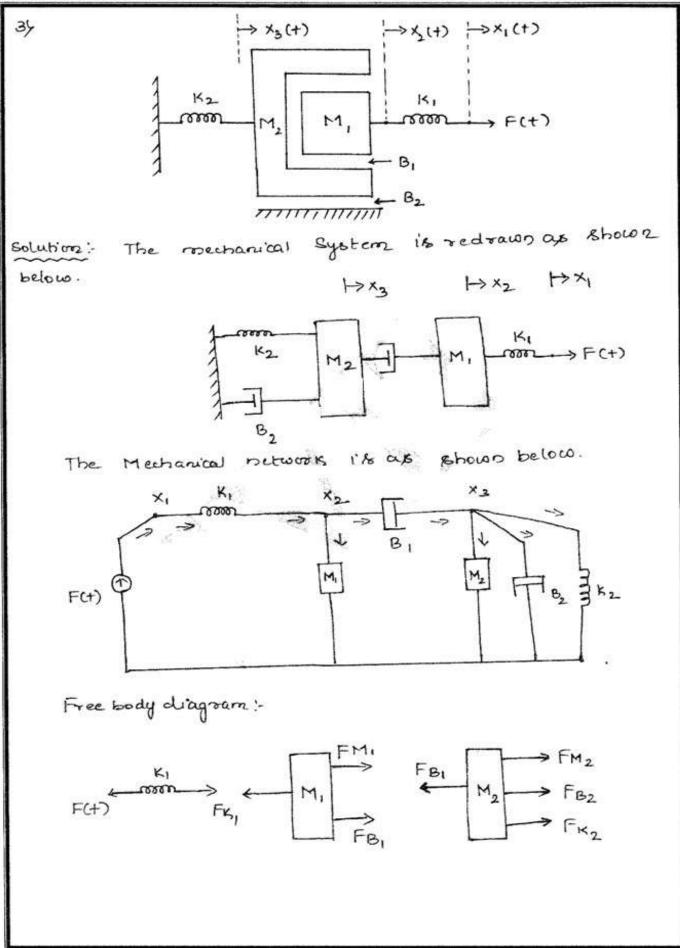
$$I_{c+} = c_1 \frac{d}{dt} V_1 + \frac{1}{L_1} \left[V_1 dt + G_1 V_1 + G_{12} (V_1 - V_2) \right] = \Im$$

$$G_{12} \frac{d}{dt} \begin{pmatrix} \phi_1 - \phi_2 \end{pmatrix} = C_2 \frac{d^2}{dt^2} \phi_2 + G_2 \frac{d}{dt} \phi_2 + \frac{1}{L_2} \phi_2$$

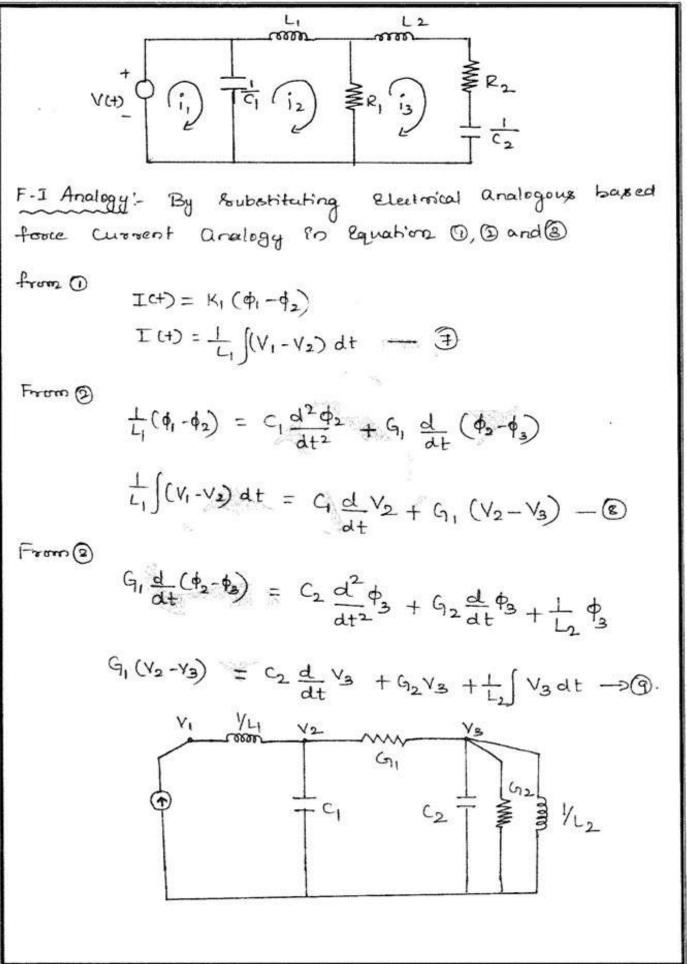
$$G_{12}(V_1-V_2) = C_2 \frac{d}{dt}V_2 + G_2V_2 + \frac{1}{L_2}\int V_2 dt \rightarrow 0$$



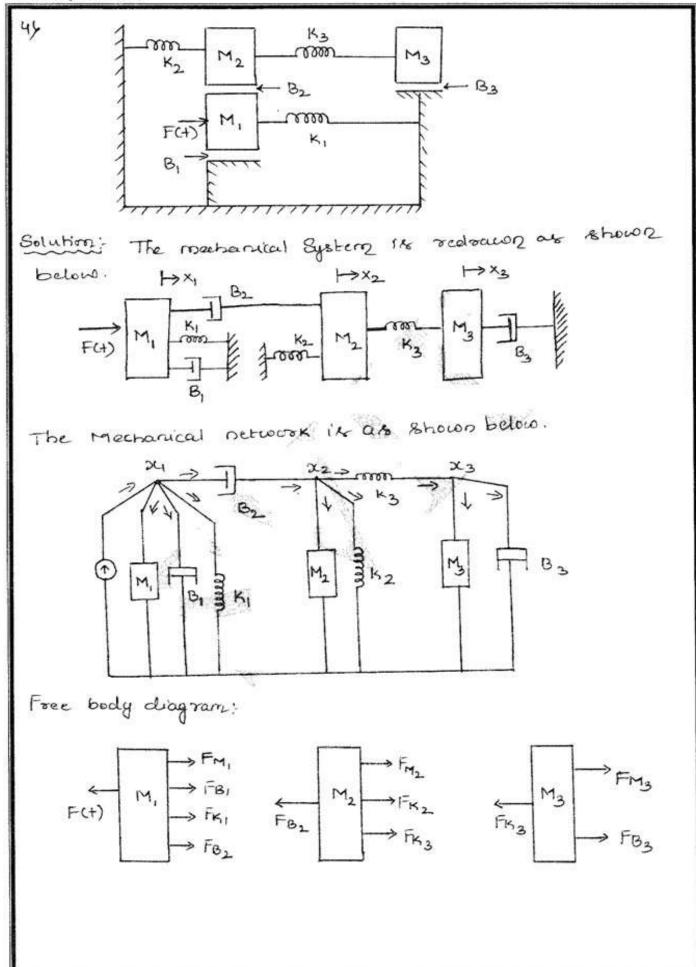
Controls Systems Notes



Equilibrium Equations for a given system are. At ∞_1 ; F(+) = K₁ ($\alpha_1 - \alpha_2$) $\longrightarrow \bigcirc$ At x_2 : $K_1(x_1-x_2) = M_1 \frac{d^2 x_2}{dt^2} + B_1 \frac{d}{dt} (x_2-x_3) \longrightarrow (2)$ At $x_3: B_1 \frac{d}{dt} (x_2 - x_3) = M_2 \frac{d^2}{dt^2} + B_2 \frac{d}{dt} x_3 + K_2 x_3 \longrightarrow 3$ F-V Analogy!- By Substituting electrical analogous based on force - Voltage analogy is Equation 0, 0+3 Fronc () $V(t) = \frac{1}{C_1} (q_1 - q_2)$ $V(t) = \frac{1}{C_1} \left(\begin{pmatrix} i_1 - i_2 \end{pmatrix} dt \rightarrow \bigcirc \right)$ from (2) $\frac{1}{c_1}(q_1 - q_2) = L_1 \frac{d^2 q_2}{d^2} + R_1 \frac{d(\dot{q}_2 - q_3)}{dt}$ $\underset{C_{1}}{\overset{\perp}{\int}}(i_{1}-i_{2})dt = L_{1} \frac{d}{dt}i_{2}^{i_{2}} + R_{1}(i_{2}-i_{2}) \rightarrow G$ From 3 $R_1 \frac{d}{dt} (q_2 - q_3) = L_2 \frac{d^2 q_3}{dt^2} + R_2 \frac{d}{dt} \frac{q_3}{dt} + \frac{1}{c_2} q_2$ $R_1(i_2 - i_3) = L_2 \frac{d}{dt}i_3 + R_2i_3 + \frac{1}{c_2} \int i_3 dt \rightarrow 0$

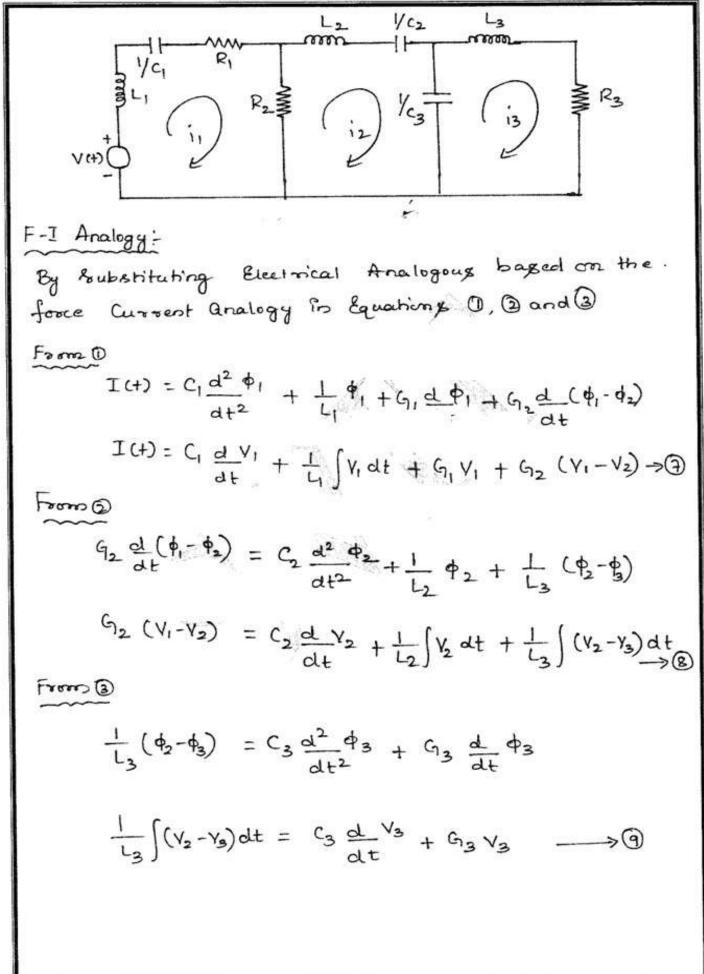


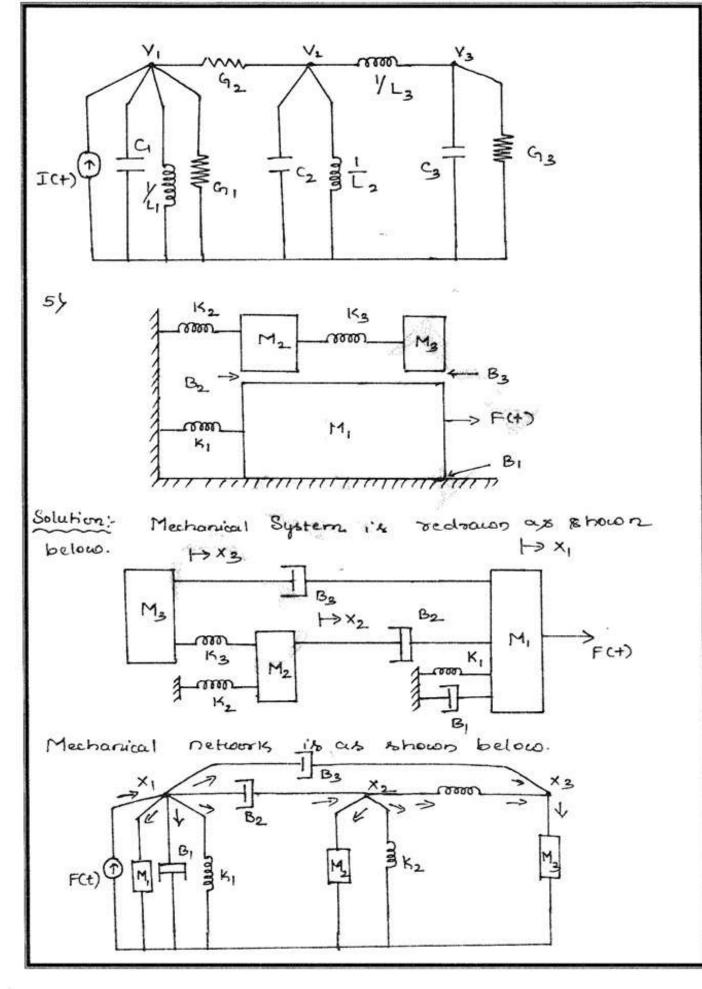
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Equilibrium Equations are given by At $24 = F(+) = F_{m_1} + F_{B_1} + F_{B_2}$ $F(t) = M_1 \frac{d^2 x_1}{dt^2} + B_1 \frac{d x_1}{dt} + K_1 x_1 + B_2 \frac{d (x_1 - x_2)}{dt} \rightarrow 0$ At x_2 ; $F_{B_2} = F_{M_2} + F_{K_2} + F_{K_3}$ $B_2 \frac{d}{dt} (x_1 - x_2) = M_2 \frac{d^2 x_2}{dt^2} + K_2 x_2 + K_3 (x_2 - x_3) \longrightarrow 2$ At x3; $Fk_3 = Fm_3 + FB_3$ $K_3(x_2-x_3) = M_3 \frac{d^2 x_3}{dt^2} + B_3 \frac{d x_3}{dt} \longrightarrow 3$ F-V Analogy :-By Substituting Electrical analogous based on force Voltage analogy in Equation (), () and (3) From 1: V(t) = $L_1 \frac{d^2 q'}{dt^2} + R_1 \frac{d''}{dt} + \frac{L}{q} q_1 + R_2 \frac{d}{dt} (q_1 - q_2)$ $V(t) = L_1 \frac{d_1}{dt} + R_1 i_1 + \frac{1}{c_1} \left(i_1 dt + R_2 (i_1 - i_2) \rightarrow \Theta \right)$ From 2: $R_2 \frac{d}{dt} \begin{pmatrix} q_1 - q_2 \end{pmatrix} = L_2 \frac{d^2}{d^2} \frac{q_2}{d^2} + \frac{1}{c_2} \frac{q_2}{d^2} + \frac{1}{c_3} \begin{pmatrix} q_2 - q_3 \end{pmatrix}$ $R_{2}(i_{1}-i_{2}) = L_{2}\frac{d}{dt}i_{2} + \frac{1}{c_{2}}\int_{2}^{1}i_{2}dt + \frac{1}{c_{3}}\int_{-3}^{1}(i_{2}-i_{3})dt + \frac{1}{c_{3}}\int_{-3}^{1}(i_{2}-i_{3})dt$ From 3 !- $\frac{1}{c_{2}}(q_{2},q_{3}) = L_{3}\frac{d^{2}q_{3}}{dt^{2}} + R_{3}\frac{d}{dt}q_{3}$ $\frac{1}{c_3}\int (i_2 - i_3)dt = \frac{1}{dt} \frac{d}{dt} + R_3 i_3 \longrightarrow \textcircled{6}$

Controls Systems Notes





Free body diagrom:

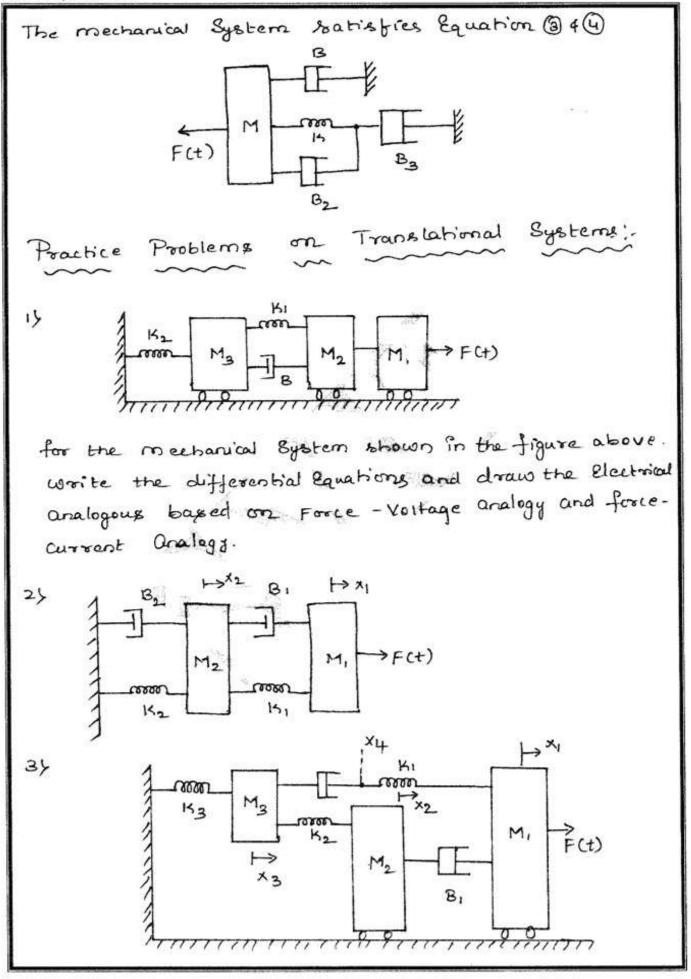
$$free body diagrom:$$

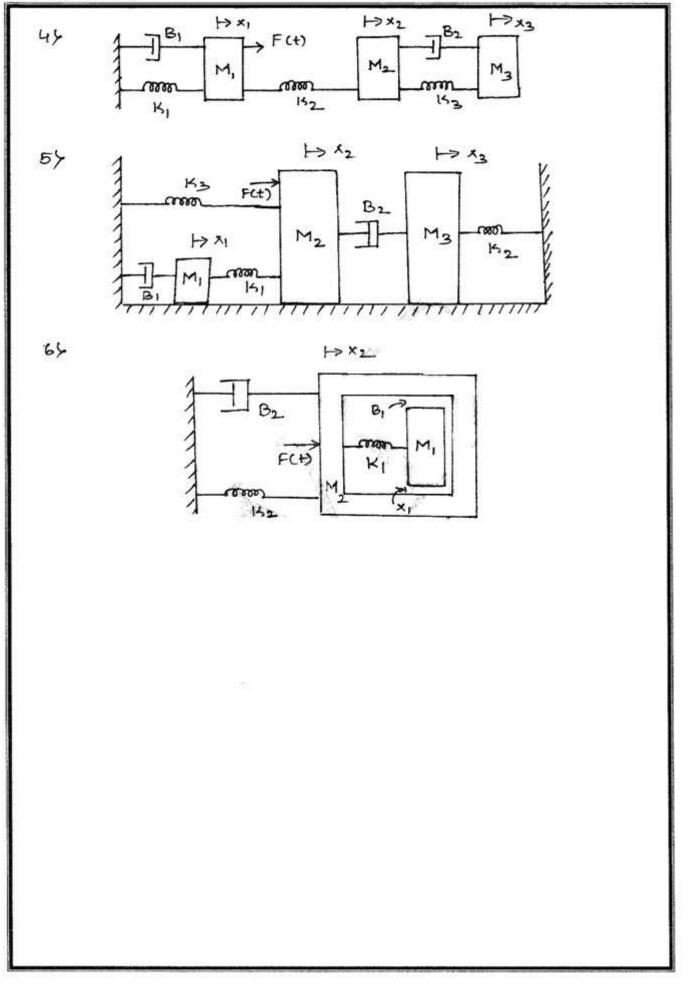
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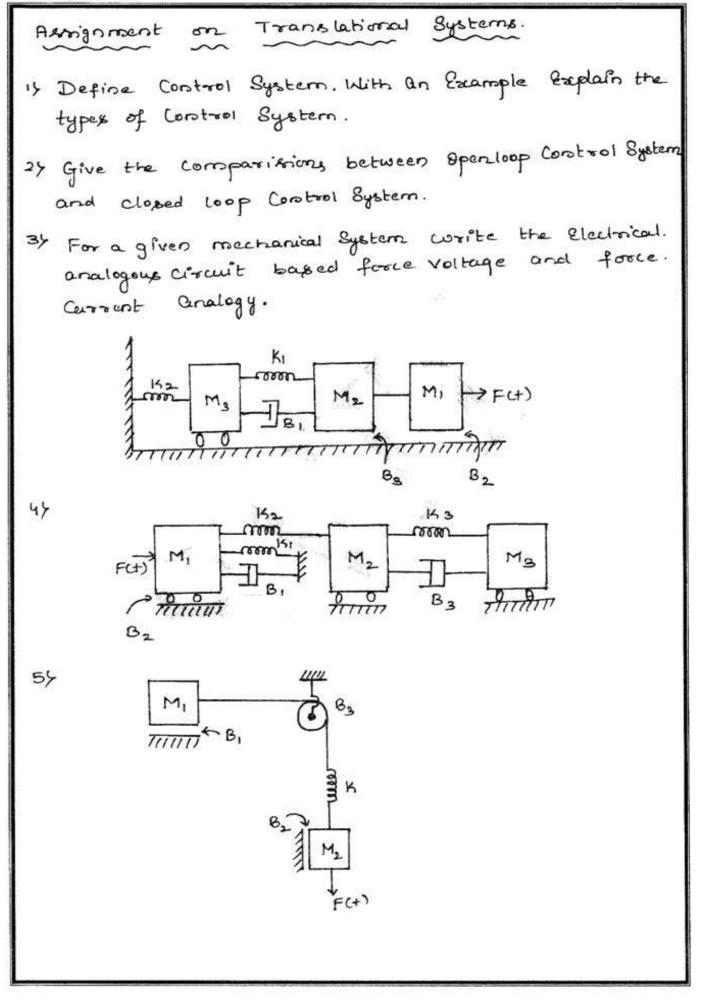
From
$$z_{3}$$
:
 $G_{2} \frac{d}{dt} (\phi_{1} - \phi_{2}) = C_{2} \frac{d^{2}}{dt^{2}} \phi_{2} + \frac{1}{L_{2}} \phi_{2} + \frac{1}{L_{3}} (\phi_{2} - \phi_{3})$
 $G_{2} (V_{1} - V_{2}) = C_{2} \frac{d}{dt} V_{2} + \frac{1}{L_{2}} \int V_{2} dt + \frac{1}{L_{3}} \int (V_{2} - V_{3}) dt \rightarrow 0$
From z_{3} :
 $\frac{1}{L_{3}} (\phi_{2} - \phi_{3}) + G_{3} \frac{d}{dt} (\phi_{1} - \phi_{3}) = G_{3} \frac{d}{dt^{2}} \phi_{3}$
 $\frac{1}{L_{3}} \int (V_{2} - V_{3}) dt + G_{3} (V_{1} - V_{3}) = G_{3} \frac{d}{dt} V_{3} \rightarrow 0$
(From z_{3} :
 $\frac{1}{L_{3}} \int (V_{2} - V_{3}) dt + G_{3} (V_{1} - V_{3}) = G_{3} \frac{d}{dt} V_{3} \rightarrow 0$
(From z_{3} :
 $\frac{1}{L_{3}} \int (V_{2} - V_{3}) dt + G_{3} (V_{1} - V_{3}) = G_{3} \frac{d}{dt} V_{3} \rightarrow 0$
(From z_{3} :
 $\frac{1}{L_{3}} \int (V_{2} - V_{3}) dt + G_{3} (V_{1} - V_{3}) = G_{3} \frac{d}{dt} V_{3} \rightarrow 0$
(From z_{3} :
 $\frac{V_{1}}{V_{2}} \int C_{2} \frac{1}{V_{2}} \int V_{3} \frac{1}{L_{1}} \int C_{2} \frac{1}{V_{2}} \int V_{3} \frac{1}{L_{2}} \int C_{3} \frac{1}{L_{1}} \int C_{3} \frac{1}{L_{1}} \int C_{3} \frac{1}{L_{1}} \int C_{3} \frac{1}{L_{2}} \int C_{3} \frac{1}{L_{1}} \int C_{3}$

=> Draw the force voltage analogous Mechanical System for the Electrical Circuit shown is the figure . P1 0000 RE in R3 i, Solution By applying ISVL to the given limit from Loop i: - $E = R_1 i_1 + L \frac{d}{dL_1} + R_2 (i_1 - i_2) + \frac{1}{c} \int (i_1 - i_2) dt$ $E = R_1 \frac{dq'}{dt} + L \frac{d^2 q_1}{dt^2} + R_2 \frac{d}{dt} \frac{(q_1 - q_2)}{dt} + \frac{1}{C} \frac{(q_1 - q_2)}{2} = 0$ form loop 12:- $R_{3i_2} + \frac{1}{c} \left[(i_2 - i_1) dt + R_2 (i_2 - i_1) = 0 \right]$ $R_{3} \frac{d}{d_{1}} + \frac{1}{c} (q_{2} - q_{1}) + R_{2} \frac{d}{d_{1}} (q_{2} - q_{1}) = 0 \longrightarrow \textcircled{2}$ By Substituting the Mechanical Elements in Equation 04@ based on F-V analogy. Form 1: $F(+) = B_{1} \frac{d^{2}x_{1}}{dt} + M \frac{d^{2}x_{1}}{dt^{2}} + B_{2} \frac{d(x_{1}-x_{2})}{dt} + K(q_{1}-q_{2}) = -33$ From 2 :- $B_3 \frac{d}{dt} x_2 + K(x_2 - x_1) + B_2 \frac{d}{dt}(x_2 - x_1) = 0 \longrightarrow (4)$

Controls Systems Notes



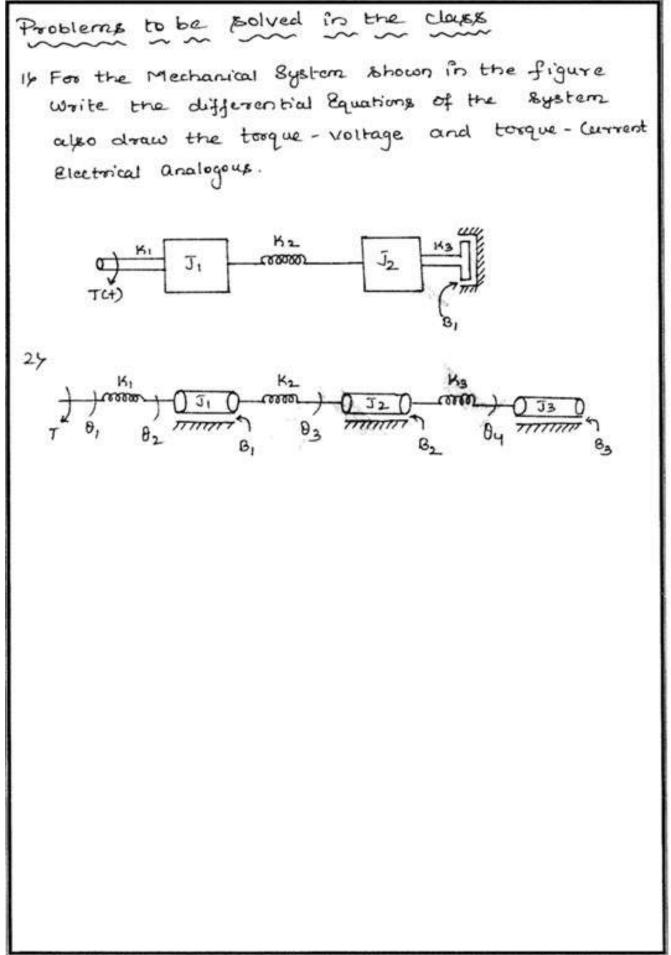




6) Draw the force voltage analogous mechanical System for the Electrical Circuit shows in figure writing the loop Equation for the Electric Circuit then transforming them to these mechanical analog 0000 m L2 2, LI Z P2 11 FY R, LI CI C2

Rotational Systems:-
The basic mechanical Elements of a rotational ystem are
as Moment of Inertia (J)
by Rotational Spring (K)
cy Rotational Dashport (B)
* Tabulation for Converting Rotational Systems to Electrical analogous (Torque -> voltage and Torque -> current)
Mechanical Electrical Analogous
Elements (T-V Analogy) (T-I Analogy) Torque [T(t)] Voltage [V(t)] (urrent [I(t)]
Angular Velocity [uct)] Current [I(t)] Voltage [V(t)]
Angular Displacement Charge [9(t)] Magnetic Flux [p(t)]
Moment of Inestia Inductance [L] Capacitance [c]
[J] I I I I I I I I I I I I I I I I I I I
Rotational Dushpot Resistance Conductance [B] [R] [G]

Controls Systems Notes



Note: Angular Velouity (
$$\mathbf{w}$$
) = $\frac{d}{dt} \Theta(t)$
Angular Acceleration (α) = $\frac{d^2}{dt^2}$
Problems on Gorahimal Systems!-
 \Rightarrow For the rolebanical System bholos in figure. Write
the differential Equations of the System also draw
the torque Voltage and torque Current electrical
Analogous.
 I_{i}
 I_{i}

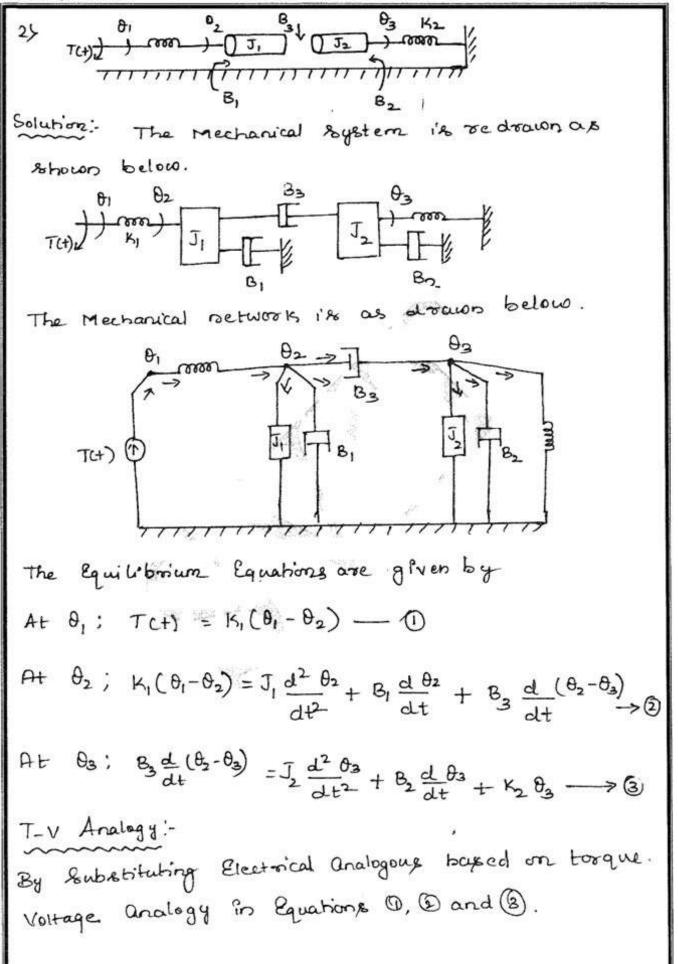
T-v Analogy
By Substituting Electrical analogous based on torque.
Voltage analogy is Equation (0, (2) and (3).
From (0),

$$v(t) = L_1 \frac{d^2}{dt^2} + R_1 \frac{d}{dt} R_1 + R_4 \frac{d}{dt} (R_1 - R_2)$$

 $V(t) = L_1 \frac{d}{dt^2} + R_1 \frac{d}{dt} R_1 + R_4 (\tilde{r}_1 - \tilde{r}_2) - (4)$
From (3),
 $R_1 \frac{d}{dt} (R_1 - R_2) = L_2 \frac{d^2}{dt^2} + R_2 \frac{d}{dt} R_2 + \frac{1}{C_1} (R_2 - R_3)$
 $R_4 (\tilde{r}_1 - \tilde{r}_2) = L_2 \frac{d}{dt} \tilde{r}_2 + R_2 \tilde{r}_2 + \frac{1}{C_1} (\tilde{r}_2 - \tilde{r}_3) \frac{dt}{dt} - (3)$
From (3),
 $\frac{1}{C_1} (R_2 - R_3) = L_3 \frac{d^2}{dt^2} + R_3 \frac{d}{dt} R_3 + \frac{1}{C_2} R_2$
 $\frac{1}{C_1} (\tilde{r}_2 - \tilde{r}_3) \frac{dt}{dt} = L_3 \frac{d\tilde{r}_3}{dt} + R_3 \tilde{r}_3 + \frac{1}{C_2} \int_{13}^{13} dt - (4)$
Electrical Concurt Satisfies Equation (4), (3) and (5)
 $L_2 \frac{R_2}{R_3} + \frac{1}{C_1} \frac{\tilde{r}_3}{\tilde{r}_3} + \frac{1}{C_2} \int_{13}^{13} dt - (4)$

T-I Analogy:
By Substituting electroical analogous based on booque.
Current analogy is Equation (), () and ()
From ()
I(t) =
$$C_1 \frac{d^2 f_1}{dt^2} + G_1 \frac{d}{dt} \frac{f_1}{t} + G_1 \frac{d}{dt} (f_1 - f_2)$$

I(t) = $C_1 \frac{dV_1}{dt} + G_1 V_1 + G_1 (V_1 - V_2)$ (F)
From ()
 $G_1 \frac{d}{dt} (f_1 - f_2) = C_2 \frac{d^2 f_2}{dt^2} + G_2 \frac{d}{dt} \frac{f_2}{t^2} + \frac{1}{L_1} (f_2 - f_3)$
 $G_1 \frac{d}{dt} (f_1 - f_2) = C_2 \frac{d}{dt} V_2 + G_2 \frac{d}{dt} \frac{f_2}{t^2} + \frac{1}{L_1} (f_2 - f_3)$
 $G_1 \frac{d}{dt} (V_1 - V_2) = C_2 \frac{d}{dt} V_2 + G_3 \frac{d}{dt} \frac{f_3}{t^2} + \frac{1}{L_1} \int (V_2 - V_3) dt - (8)$
From ()
From ()
 $I_1 (f_2 - f_3) = C_3 \frac{d^2 f_3}{dt^2} + G_3 \frac{d}{dt} \frac{f_3}{t} + \frac{1}{L_2} \frac{f_3}{t}$
 $I_1 \int (V_2 - V_3) dt = C_3 \frac{d^2 f_3}{dt^2} + G_3 \frac{d}{dt} \frac{f_3}{t} + \frac{1}{L_2} \int V_3 dt - (9)$
Electrical (eruit Scatiblying Equation $f_1 \in and 9$
 V_1 V_2 V_2 V_3 V_3 V_4 V



From (i)
$$V(t) = \frac{1}{C_1} (q_1 - q_2)$$

 $V(t) = \frac{1}{C_1} \int (i_1 - i_2) dt = 0$
From (i) $\frac{1}{C_1} (q_1 - q_2) = L_1 \frac{d^2 q_1 2}{dt^2} + R_1 \frac{d q_1 2}{dt} + R_2 \frac{d t}{dt} (q_2 - q_3)$
 $\frac{1}{C_1} \int (i_1 - i_2) dt = L_1 \frac{d i_2}{dt^2} + R_1 i_2 + R_3 (i_2 - i_3) = 0$
From (i) $R_3 \frac{d (q_2 - q_3)}{dt} = L_1 \frac{d i_2}{dt^2} + R_2 \frac{d q_3}{dt} + \frac{1}{C_2} q_3$
 $R_3 (i_2 - i_3) = L_1 \frac{d i_3}{dt^2} + R_2 i_3 + \frac{1}{C_2} \int i_3 dt \rightarrow 0$
Electrical Concent Ratiofying Equation (i) (i) and (i) $R_1 \frac{L_2}{L} \frac{R_2}{L}$
 $V(t) (i_1) = \frac{N_1}{L_1} \frac{I_2}{L_1} \frac{R_2}{L_2} \frac{I_3}{L_2} \frac{N_2}{L_2}$
T-I Analogy:
By Substituting Electrical Analogous based on torque
Curve ent Analogy in Equation (i) (i) and (i).

From () エ(+)= 亡((+, - +2) $I(t) = \frac{1}{1} \left((V_1 - V_2) dt - \overline{P} \right)$ From 2 $\frac{1}{L_1}(\phi_1 - \phi_2) = c_1 \frac{d^2}{d^2} \phi_2 + G_1 \frac{d}{d^2} \phi_2 + G_3 \frac{d}{d^4}(\phi_2 - \phi_3)$ $\frac{1}{L_1}\int (V_1-V_2) dt = C_1 \frac{d}{dt}V_2 + C_1 V_2 + C_3 (V_2-V_3) \longrightarrow (B)$ From (2) $G_{3} \frac{d}{dt} (\phi_{2} - \phi_{3}) = G_{2} \frac{d^{2}}{dt^{2}} \phi_{3} + G_{2} \frac{d}{dt} \phi_{3} + \frac{1}{L_{2}} \phi_{3}$ 58 $G_3(V_2 - V_3) = G_2 \frac{d}{d+} V_3 + G_2 V_3 + \frac{1}{L_1} \int V_3 dt \longrightarrow @$ Electroical Circuit Soatisfies Equation (1), (8) and (9). V3 V2 1/61 V. 503 s_T ¥6, 口十

Assignment on Rotational Systems. 4 Write the differential Equations describing the behavior of the mechanical system shown Pr the figure. Also draw an analogous electrical Crowit. (T-V and T-I) 142 14, T(+) 000 10001 J2 J3 -) JI אות B3 אות אות יזרחות) 131 B5 By B2 12

Module no 1 Questions

- Q-1: Explain with examples open loop and closed loop control systems. List merits and demerits of both. Jun. 2014, 10 Marks
- Q-2: Compare open loop and closed loop control systems and give one practical example of each. Jun. 2013, 6 Marks
- Q-3: Compare linear and non-linear control systems. Dec. 2012, 4 Marks
- Q-4: Define control system. Draw the basic block diagram of a control loop giving all the relevant details. **Dec. 2012, 4 Marks**
- Q-5: Derive the electrical analogous quantities for the mechanical quantities using force-voltage analogy. Dec. 2010, 5 Marks
- Q-6: Derive the mathematical model for an armature controlled DC motor. Dec. 2010, 5

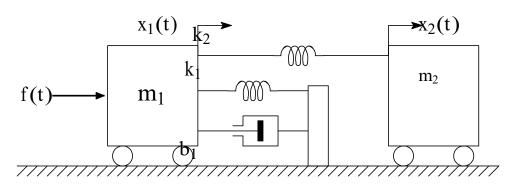
Marks

Q-7: For what purpose feedback is used in control system? Mention the effects of feedback on (i) stability (ii) overall gain (iii) disturbance and (iv) sensitivity of control systems.

10 Marks

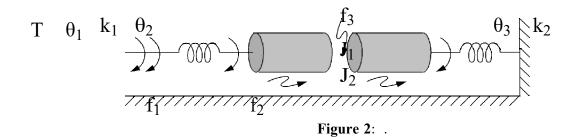
Q-8: For the system shown in Figure.1 write mechanical network and obtain its mathematical mod Jul. 2013, 6 Marks



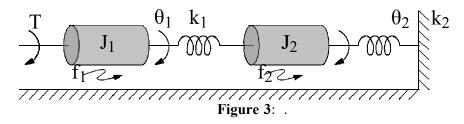


Jul. 2005,

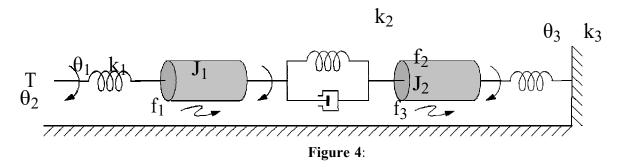
Q-9: For the system shown in Figure.2 write its mechanical network and obtain mathematical model and electrical analogous based on force-current analogy. Jul. 2013, 8 Marks



Q-10: Draw the electrical network based on torque-current analogy give all the performance equations for Figure. 3 Jul. 2014, 10 Marks



Q-11: For the rotational system shown in Figure.4. (i) Draw the mechanical network. (ii) Write the differential equations. (iii) Obtain torque to voltage analogy. **Dec. 2012, 08 Marks**



Q-12: For the mechanical system shown in Figure.5.

(i) Draw the free-body diagram and mechanical network. Write the differential equa- tions describing behaviour of the system.

(ii) Draw the electrical network based on force-voltage analogy and write the analogous electrical equations.

(iii) Draw the electrical network based on force-current analogy and write the analogous electrical equations.

Q-13: For the mechanical system shown in Figure.6.

(i) Draw the free-body diagram and mechanical network. Write the differential equa- tions describing behaviour of the system.

(ii) Draw the electrical network based on force-voltage analogy and write the analogous electrical equations.

(iii) Draw the electrical network based on force-current analogy and write the analogous electrical equations.

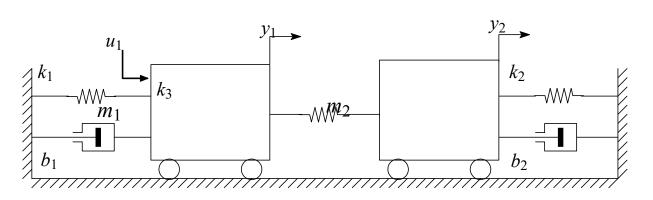


Figure 5: Mechanical system: u_i is force (N), y_i is displacement (m), m_i is mass (Kg), k_i is spring constant (N/m), b_i is viscous friction coefficient (N-s/m).

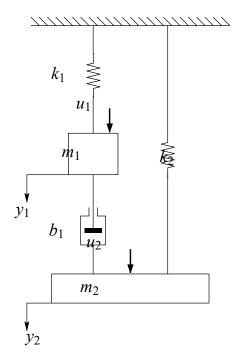


Figure 6: Mechanical system: u_i is force (N), y_i is displacement (m), m_i is mass (Kg), k_i is spring constant (N/m), b_i is viscous friction coefficient (N-s/m).

Q-14: For the mechanical system shown in Figure.7.

- (i) Draw the free-body diagram and mechanical network. Write the differential equa- tions describing behaviour of the system.
- (ii) Draw the electrical network based on force-voltage analogy and write the analogous electrical equations.
- (iii) Draw the electrical network based on force-current analogy and write the analogous electrical equations.

Q-15: For the mechanical system shown in Figure.8.

- (i) Draw the free-body diagram and mechanical network. Write the differential equa- tions describing behaviour of the system.
- (ii) Draw the electrical network based on force-voltage analogy and write the analogous electrical equations.

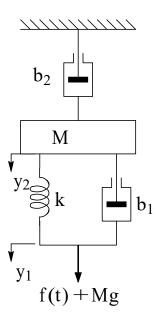


Figure 7: Mechanical system: f(t) is force (N), y_i is displacement (m), m_i is mass (Kg), k_i is spring constant (N/m), b_i is viscous friction coefficient (N-s/m), g is acceleration due to gravity (N-s²/m).

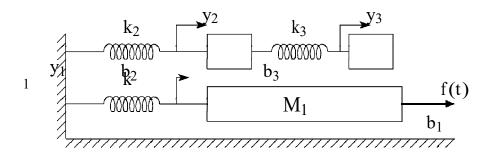


Figure 8: Mechanical system: f(t) is force (N), y_i is displacement (m), m_i is mass (Kg), k_i is spring constant (N/m), b_i is viscous friction coefficient (N-s/m)).

(iii) Draw the electrical network based on force-current analogy and write the analogous electrical equations.

Q-16: For the mechanical system shown in Figure.9.

- (i) Draw the free-body diagram and mechanical network. Write the differential equa- tions describing behaviour of the system.
- (ii) Draw the electrical network based on force-voltage analogy and write the analogous electrical equations.
- (iii) Draw the electrical network based on force-current analogy and write the analogous electrical equations.

Q-17: For the mechanical system shown in Figure.10.

- (i) Draw the free-body diagram and mechanical network. Write the differential equa- tions describing behaviour of the system.
- (ii) Draw the electrical network based on force-voltage analogy and write the analogous electrical equations.
- (iii) Draw the electrical network based on force-current analogy and write the analogous electrical equations.

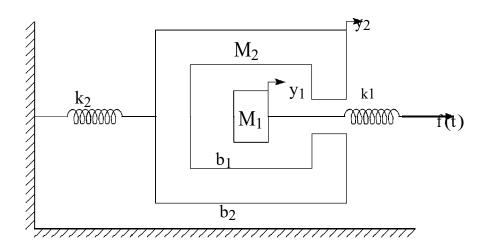


Figure 9: Mechanical system: f(t) is force (N), y_i is displacement (m), m_i is mass (Kg), k_i is spring constant (N/m), b_i is viscous friction coefficient (N-s/m).

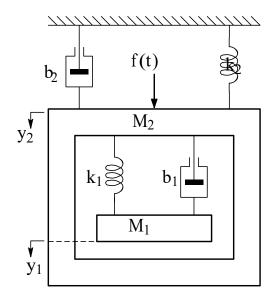


Figure 10: Mechanical system: f(t) is force (N), y_i is displacement (m), m_i is mass (Kg), k_i is spring constant (N/m), b_i is viscous friction coefficient (N-s/m).

Q-18: For the mechanical system shown in Figure.11.

- (i) Draw the free-body diagram and mechanical network. Write the differential equa- tions describing behaviour of the system.
- (ii) Draw the electrical network based on torque-voltage analogy and write the analo- gous electrical equations.
- (iii) Draw the electrical network based on torque-current analogy and write the analo- gous electrical equations.

Q-19: For the mechanical system shown in Figure.12.

- (i) Draw the free-body diagram and mechanical network. Write the differential equa- tions describing behaviour of the system.
- (ii) Draw the electrical network based on torque-voltage analogy and write the analo- gous electrical equations.
- (iii) Draw the electrical network based on torque-current analogy and write the analo- gous electrical equations.

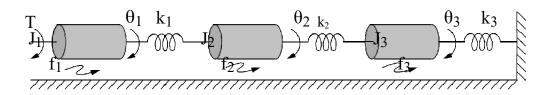


Figure 11: Mechanical system: T is torque (Nm), y_i is displacement (m), J_i is moment of inertia (Kgm²), k_i is torsional spring constant (Nm/rad), f_i is viscous friction coefficient (Nm-s/rad).

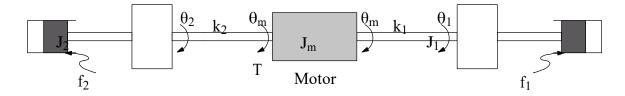


Figure 12: Mechanical system: T is torque (Nm), y_i is displacement (m), J_i is moment of inertia (Kgm²), k_i is torsional spring constant (Nm/rad), f_i is viscous friction coefficient (Nm-s/rad).

Q-20: Figure.13 shows a motor-load system coupled through a gear train with gear ratio $n = N_1/N_2$. The motor torque is $T_m(t)$ and $T_L(t)$ represents a load torque.

- (i) Draw the free-body diagram and mechanical network. Write the differential equa- tions describing behaviour of the system.
- (ii) Draw the electrical network based on torque-voltage analogy and write the analo- gous electrical equations.
- (iii) Draw the electrical network based on torque-current analogy and write the analo- gous electrical equations.
- **Q-21:** Figure.14 shows the diagram of a print-wheel system with belts and pulleys. The belts are modeled as linear springs with spring constants k_1 and k_2 .
- (i) Draw the free-body diagram and mechanical network. Write the differential equa- tions describing behaviour of the system.
- (ii) Draw the electrical network based on torque-voltage analogy and write the analo- gous electrical equations.
- (iii) Draw the electrical network based on torque-current analogy and write the analo- gous electrical equations.

Q-22: For the mechanical system shown in Figure.15.

- (i) Draw the free-body diagram and mechanical network. Write the differential equa- tions describing behaviour of the system.
- (ii) Draw the electrical network based on torque-voltage analogy and write the analo- gous electrical equations.
- (iii) Draw the electrical network based on torque-current analogy and write the analo- gous electrical equations.

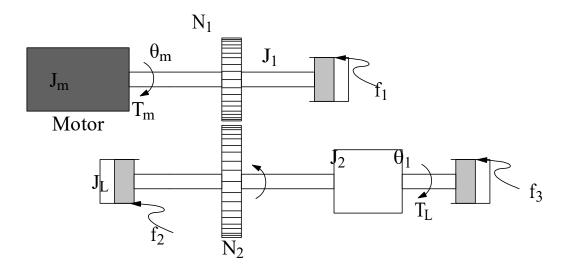


Figure 13: Mechanical system: T_m is motor torque (Nm), y_i is displacement (m), J_i is moment of inertia (Kgm²), k_i is torsional spring constant (Nm/rad), f_i is viscous friction coefficient (Nm-s/rad).

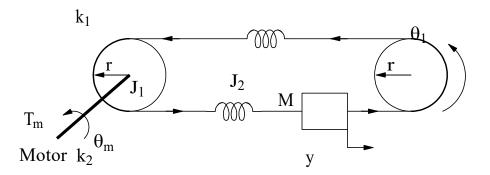


Figure 14: Mechanical system: T_m is motor torque (Nm), y_i is displacement (m), J_i is moment of inertia (Kgm²), k_i is torsional spring constant (Nm/rad), y is displacement (m).

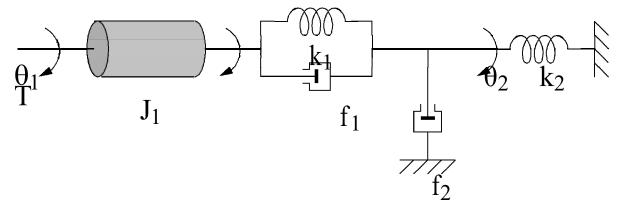
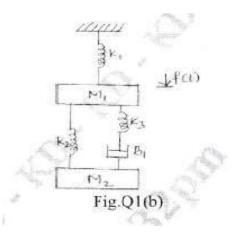


Figure 15: Mechanical system: T is torque (Nm), y_i is displacement (m), J_i is moment of inertia M(Kgm²), k_i is torsional spring constant (Nm/rad), f_i is viscous friction coefficient (Nm-s/rad).

Q:23. For the mechanicrrl system shorrn in Fig.Ql(b) the analogotts clectrical netu'ork based on F-V analogy.

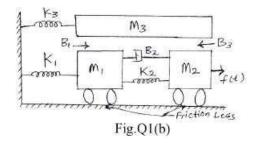


Q:24. For the mechanical system shown in Fig.Ql(b):

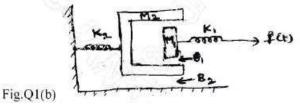
i) Draw the mechanical network.

ii) Obtain equations of motion.

iii) Draw an electrical network based on force current analogy.

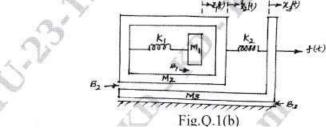


Q:25. Write the differential eil\iailons fbr the mechanical s)steni shoun in Fig.QI(b) and ohtain F-V and F-I analogous electrica i networks. (05 Nlarks)I

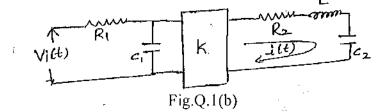


Q:26.

b. For the mechanical system shown in Fig.Q.1(b), write i) The mechanical network ii) the equations of motion and iii) the force-current analogous electrical network. (08 Marks)



Find the transfer function $\frac{l(s)}{Ui(s)}$ for the circuit shown in Fig.Q.1(b) and K is the gain of an ideal amplifier. (06 Marks)



Q:27



KLE COLLEGE OF ENGINEERING AND TECHNOLOGY DEPARTMENT OF ELECTRONICS AND COMMUNICATION ENGINEERING KLE COLLEGE OF ENGINEERING AND TECHNOLOGY CHIKODI

CONTROL SYSTEMS NOTES (18EC43) (As per Choice based Credit System (CBCS) Scheme) IVTH SEMESTER



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"Don't see others doing better than you, beat your own records every day, because success is a fight between you and yourself"

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MODULE-2

Syllabus:

- Transfer functions
- Block diagrams and
- ➤ signal flow graphs
- Block diagram algebra and
- ➢ Signal Flow graphs.

Transfer function:
* Transfer function gives the mathematical Equivalent
model for a system.
Defination I:
The Transfer function is defined as the value of
laplace Transferring of the Dutput to the Laplace transform
of the Poput Considering all initial Conditions Equals
to zero;

$$Y(t) \rightarrow System \rightarrow C(t)$$

 $V(t) \rightarrow System \rightarrow U(t)$
 $V(t) \rightarrow V(t)$
 $V(t) \rightarrow V(t)$

> Impulse. response in also Called as System suponse, natural response, free force response, Input response.

Properties of Tranxfer function (T.F):

- * The transfer function of a system its the laplace transform of it's impulse response for Zero instial Conductions.
- * The transfer function Can be determined from System input - Dutput poir by taking ratio of laploce of output to laplace of Popul.
- * The System differential Equation Can be Obtained from transfer function by replacing S- Vaniable with linear differential operator D, defined by D= d
- * The transfer function is independent of the Popula to the bystem.
- * The System polys/ Zeros Caobe found out from transfer function
- * Stability Can be determined from the characteristic Equation
- * The transfer functions is defined Only for linear time invariant functions, it is not defined non-linear System.

SL. NO	F-C+)	F(S) = L[F(+)]
לו	S(t) impulse response at t=0	1
25	Uct) Unit step at t=0	1/s
46	u(t-T) unit stepat t = T	-1 e-ST
45	t	1/52
54	12/2	1/53
65	ť	<u>n1</u> S ⁿ⁺¹
Ŧŷ	e ^{-at}	
85	eat	<u> </u>
95	t.e ^{-at}	$\frac{1}{(s+\alpha)^2}$
105	t · eat	$\frac{1}{(s-\alpha)^2}$
117	t ⁿ e ^{-at}	$\frac{n!}{(s+a)^{n+1}}$
125	Sin wt	<u>_w</u> <u>s²+w²</u>
135	COB WOt	$\frac{s}{s^2 + \omega^2}$
143	e ^{at} Sin cot	$\omega/(s+\alpha)^2+\omega^2$
154	e cosuot	(S+x)/(S+x)2+W2

$$\begin{bmatrix} 14 & Sin h a't & \frac{a'}{S^2 - a'^2} \\ 17 & Cosh a't & \frac{5}{S^2 - a'^2} \\ 18 & \frac{1}{a^2} (at - 1 + e^{-\alpha t}) & \frac{1}{S^2(S + \alpha)} \\ \end{bmatrix}$$

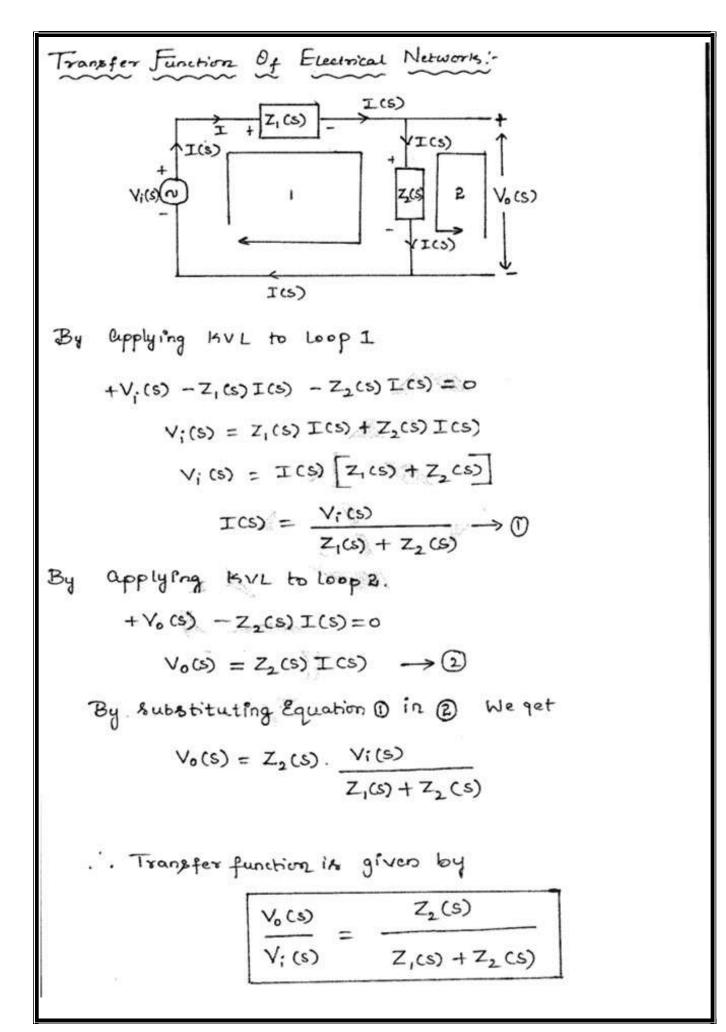
$$\begin{bmatrix} Laplace transform & of Revistance 'R' & L_T[R] = R \\ Laplace transform & of Capacitance 'C' & L_T[C] = \frac{1}{Sc} \\ Laplace transform & of Inductance 'L' & L_T[L] = SL \\ L[F(t)] = F(S). \\ \end{bmatrix}$$

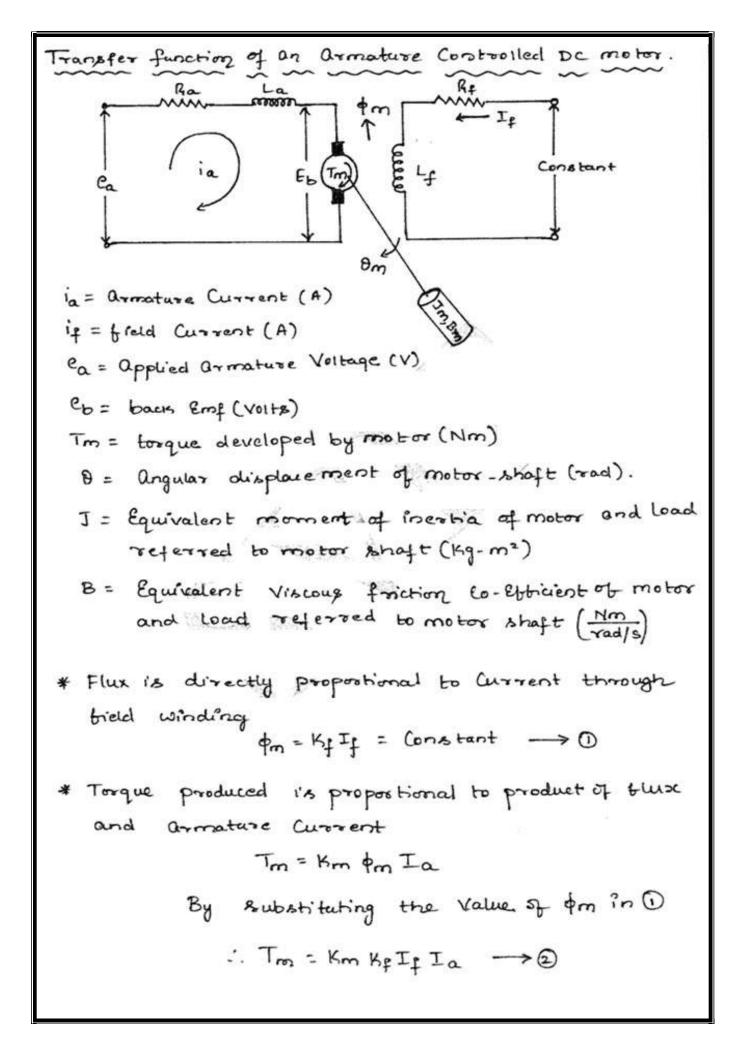
$$\begin{bmatrix} L & L[F(t)] = F(S). \\ L & L & L[At^2] = S(S) \\ L & L & L & L(At^2) = S(S) \\ L & L & L & L(At^2) = S(S) \\ L & L & L(At^2) = S(S) \\ L & L & L & L(At^2) = S(S) \\ L & L & L(At^2) \\ L & L(At^2) = S(S) \\ L & L & L(At^2) \\ L & L(A$$

* Laplace transform of linear lombination.

$$L\left[af_{1}(t) + bf_{2}(t)\right] = aF_{1}(s) + bF_{2}(s)$$
Where $f_{1}(t), f_{2}(t)$ are functions of time and atbase
constants.
Scale change. $f\left[\frac{t}{a}\right] \Rightarrow aF(as); a > 0$
Real translation $f(t-to) \Rightarrow e^{-st_{0}}F(s)$
(complex translation $e^{-at}f(t) \Rightarrow F(s+a)$
Multiplication by t $t^{n}f(t) \Rightarrow (-1)^{n} \frac{d^{n}F(s)}{ds^{n}}$
Multiplication by t $t^{n}f(t) \Rightarrow \int_{s\to\infty}^{\infty} F(s) ds$
initial value theorem $\lim_{t\to\infty} f(t) = \lim_{s\to\infty} sF(s)$
 $t = Final Value theorem \lim_{t\to\infty} f(t) = \lim_{s\to\infty} sF(s)$
 $t = \int_{at}^{t} f(t) = \int_{s\to\infty}^{t} F(s) ds = \int_{s\to\infty}^{t} F(s) ds$
 $t = \int_{at}^{t} f(t) = \int_{s\to\infty}^{t} (t) = \int_{s\to\infty}^{t} (t) = 0.$
(i) $L \frac{d}{dt}f(t) = [sF(s) - F(o)] \quad (F(o) = 0.$
(ii) $L \frac{d^{2}f(t)}{dt^{2}} = [s^{2}F(s) - sF(o) - sF^{1}(o) - F^{11}(o)]$
Where $f(o), f^{1}(o), f^{11}(o)$ are the values of $f(t), \frac{d}{dt}f(t), \frac{d^{2}f(t)}{dt+f(t)}, \frac{d^{2}f(t)}{dt+f(t)}, \frac{d^{2}f(t)}{dt+f(t)}, \frac{d^{2}f(t)}{dt+f(t)} = 0$

* If the Laplace transform of
$$f(t)$$
 is $F(t)$, then
(i) $L \int f(t) = \left[\frac{F(t)}{s} + \frac{t^{-1}(0)}{s}\right]$
(ii) $L \int \int f(t) = \left[\frac{F(t)}{s^2} + \frac{t^{-1}(0)}{s^2} + \frac{t^2(0)}{s}\right]$
(iii) $L \int \int \int f(t) = \left[\frac{F(t)}{s^3} + \frac{t^{-1}(0)}{s^3} + \frac{t^{-2}(0)}{s^2} + \frac{t^{-3}(0)}{s}\right]$
Where $f^{-1}(0)$, $f^{-2}(0)$, $f^{-3}(0)$ One the Values of $f(t)$,
 $\int \int f(t)$, $\int \int f(t) = --- \alpha t t = 10$.





* Baus Emf "Eb" is directly propertional to Shaft
Velocity "Wm", as flux
$$d_m$$
 is constant.
* We know that velocity $Wm = \frac{d}{dt} \frac{g_m(t)}{dt} = W_m(s) = S \theta_m(s)$
Baus Emf Eb(s) = Kb Wm(s) = KbS $\theta_m(s) \rightarrow (3)$
* By applying Kvk to armature circuit We get
 $e_a = Eb + I_a(Ra) + La \frac{dia}{dt}$
By taking leplace transform
Ea(s) = Eb(s) + Ia(s) Ra + Lars Ia(s)
Ea(s) = Eb(s) + Ia(s) Ra + Lars Ia(s)
Ea(s) = Eb(s) + Ia(s) Ra + sLa
 $Ia(s) = Ea(s) - Eb(s)$
* By substituting liquation (1) in (2)
 $Im = Km K_F I_F \left\{ \frac{Ea(s) - Eb(s)}{Ra - sLa} \right\} \rightarrow (9)$
* The differential Equation is given by.
 $Im = Jm \frac{d^2 Bm}{dt^2} + Bm \frac{d}{dt} Bm(s) - (6)$
Substituting liquation (2) in (3)
 $Im = \left\{ Jms^2 + Bms \right\} Bm(s) - (6)$
Km Kg If $\left\{ \frac{Ea(s) - Eb(s)}{Ra - sLa} \right\} = \left\{ Jms^2 + Bms \right\} Bm(s)$

$$\frac{k_{m} K_{f} I_{f} E_{a}(s) - K_{m} K_{f} I_{f} E_{b}(s)}{R_{fa} - SL_{a}} = \left\{ Ims^{2} + Bms^{2} \right\} \theta_{m}(s) \longrightarrow \mathfrak{F}$$
By substituting (a) In (f)
$$\frac{K_{m} K_{f} I_{f} E_{a}(s) - K_{m} K_{f} I_{f} K_{b} S \theta_{m}(s)}{R_{a} - SL_{a}} = \left\{ Jms^{2} + Bms^{2} \right\} \theta_{m}(s)$$

$$\frac{K_{m} K_{f} I_{f} E_{a}(s)}{R_{a} - SL_{a}} - \frac{K_{m} K_{f} I_{f} K_{b} S \theta_{m}(s)}{R_{a} - SL_{a}} = \left\{ Jms^{2} + Bms^{2} \right\} \theta_{m}(s)$$

$$\frac{K_{m} K_{f} I_{f} E_{a}(s)}{R_{a} - SL_{a}} - \frac{K_{m} K_{f} I_{f} K_{b} S \theta_{m}(s)}{R_{a} - SL_{a}} = \left\{ Jms^{2} + Bms^{2} \right\} \theta_{m}(s)$$

$$\frac{K_{m} K_{f} I_{f} E_{a}(s)}{R_{a} - SL_{a}} - \frac{K_{m} K_{f} I_{f} K_{b} S \theta_{m}(s)}{R_{a} - SL_{a}} + \left(Jms^{2} + Bms^{2} \right) \theta_{m}(s)$$

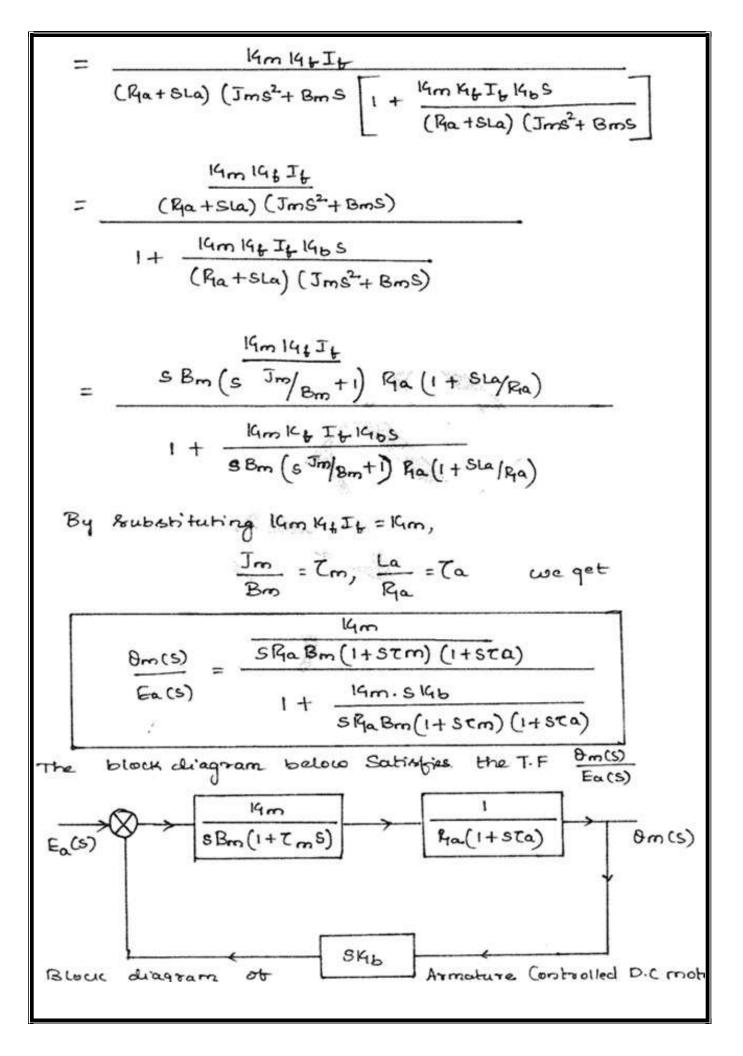
$$\frac{K_{m} K_{f} I_{f} E_{a}(s)}{R_{a} - SL_{a}} = \frac{K_{m} K_{f} I_{f} K_{b} S}{R_{a} - SL_{a}} + Jms^{2} + Bms^{2} \theta_{m}(s)$$

$$\frac{\theta_{m}(s)}{E_{a}(s)} = \frac{K_{m} K_{f} I_{f} K_{b} S}{R_{a} - SL_{a}} + Jms^{2} + Bms^{2} \theta_{m}(s)$$

$$\frac{\theta_{m}(s)}{E_{a}(s)} = \frac{K_{m} K_{f} I_{f} K_{b} S}{R_{a} + SL_{a}} + Jms^{2} + Bms^{2} R_{m} S$$

$$\frac{\theta_{m}(s)}{R_{a} + SL_{a}} = \frac{K_{m} K_{f} I_{f} K_{b} S}{R_{a} + SL_{a}} + Jms^{2} + Rms^{2} R_{m} S$$

$$\frac{\theta_{m}(s)}{E_{a}(s)} = \frac{K_{m} K_{f} I_{f} K_{b} S}{K_{m} K_{f} I_{f} K_{b} S + Jms^{2} (R_{a} + SL_{a}) + Bms^{2} (R_{a} + SL_{a})}{R_{a} + SL_{a}}$$



$$T_{m} = k_{1}^{r} \oint I_{a} = k_{1}^{r} l_{4} I_{4} I_{4} I_{4}$$

$$T_{m} = l_{4m}^{r} l_{4} I_{4} \longrightarrow (2)$$
Where $l_{4m} = l_{1}^{r} I_{a} = (constant)$
By applying leve to field Current
$$L_{f} \frac{d}{dt} + P_{4} P_{4} I_{4} = c_{f} \longrightarrow (3)$$
By taking laplace transform.
$$E_{f}(s) = L_{f}(s) I_{f}(s) + P_{4} I_{f}(s)$$

$$E_{f}(s) = [L_{f}(s) + R_{f}] I_{f}(s)$$

$$I_{f}(s) = \frac{E_{f}(s)}{s L_{f} + R_{f}} \longrightarrow (3)$$
By applying laplace transforms for Equation (2)
$$T_{m}(s) = l_{4m}^{r} l_{4f} I_{f}(s)$$

$$B_{4} \text{ substituting the Value of } I_{f}(s)$$

$$T_{m}(s) = l_{4m}^{r} l_{4f} \frac{E_{f}(s)}{s L_{f} + R_{f}} \longrightarrow (3)$$

$$T_{m}(s) = l_{4m}^{r} l_{4f} \frac{E_{f}(s)}{s L_{f} + R_{f}} \longrightarrow (5)$$

$$T_{m}(s) = l_{4m}^{r} l_{4f} \frac{E_{f}(s)}{s L_{f} + R_{f}} \longrightarrow (5)$$

$$T_{m}(s) = l_{4m}^{r} l_{4f} \frac{E_{f}(s)}{s L_{f} + R_{f}} \longrightarrow (5)$$

$$T_{m}(s) = l_{4m}^{r} l_{4f} \frac{E_{f}(s)}{s L_{f} + R_{f}} \longrightarrow (5)$$

By taking laplace transform for Equation (c)

$$T_{m}(s) = Jm s^{2} \ \theta m (s) + Bm s \ \theta m (s)$$

$$Tm (s) = (Jm s^{2} + Bm s) \ \theta m (s) = -(7)$$
By Aubstituting Equation (f) in (f) we get

$$(s^{2} Jm + sBm) \ \theta m (s) = \frac{l(m l(t_{1} E_{1}(s))}{(st_{1} + R_{1}t)}$$
Here Eq(s) is the input and $\theta m (s)$ is the $\theta utput$.
The transfer function is $q^{2} uer by = \frac{\theta m (s)}{E_{1}(s)}$

$$\frac{\partial m (s)}{E_{1}(s)} = \frac{l(m l(t_{1} + C_{1}(s)))}{(s^{2} Jm + sBm)(st_{1} + R_{1}t)}$$

$$= \frac{l(m l(t_{1} + C_{1}(s)))}{Bm} \left[\frac{s^{2} Jm + sBm}{Bm} \right] \left[\frac{st_{1}}{R_{1}} + 1 \right]$$

$$= \frac{l(m l(t_{1} + C_{1}(s)))}{Bm} \left[\frac{s m}{Bm} + 1 \right] \left[\frac{st_{1}}{R_{1}t} + 1 \right]$$
Substitute $\frac{Jm}{Bm} = Tm$ and $\frac{t_{1}}{R_{1}t} = T_{1}$

$$\therefore \frac{\theta m (s)}{E_{1}(s)} = \frac{l(m l(t_{1} + C_{1}(s)))}{s R_{1}R_{1}R_{1}} \left[\frac{st_{1}}{R_{1}t} + 1 \right]$$

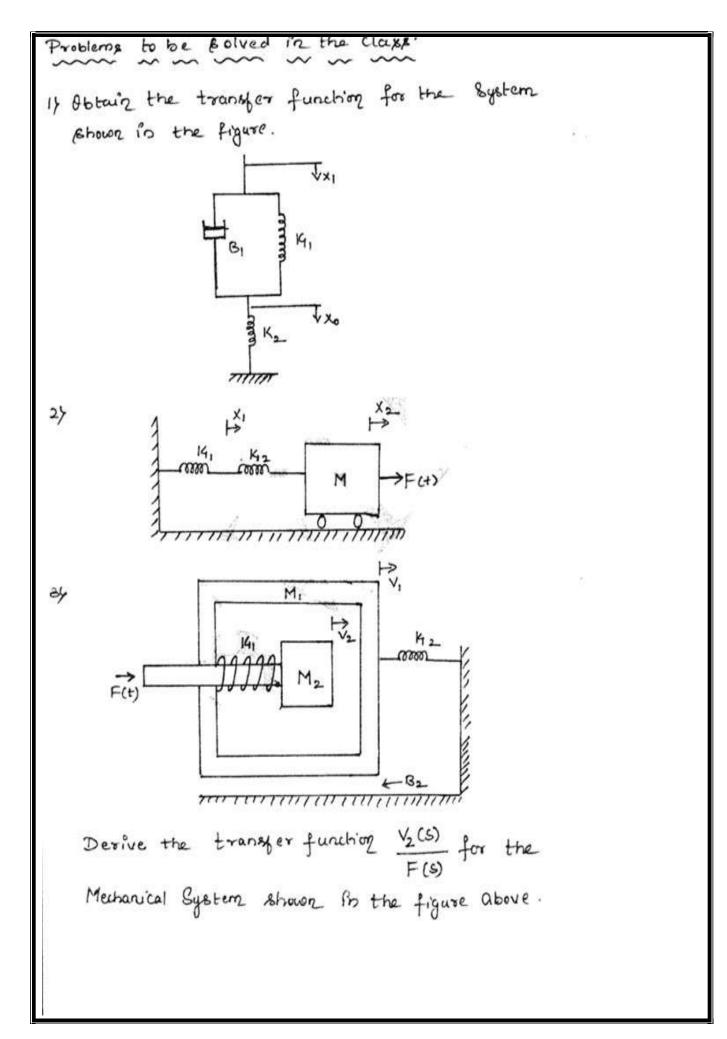
$$T.F = \frac{\theta_{m}(s)}{E_{F}(s)} = \frac{14t}{R_{1F}[1+st_{F}]} \cdot \frac{14m}{B_{m}[1+st_{m}]} \cdot \frac{1}{s}$$

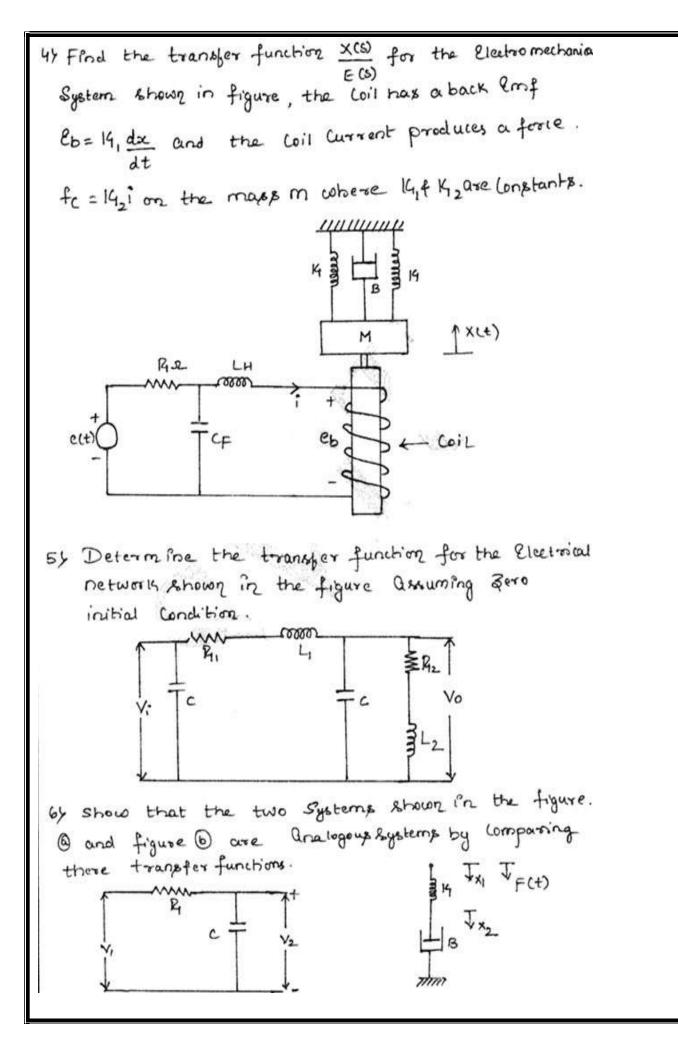
$$The blow diagram leatistics the transfer function
$$\frac{\theta_{m}(s)}{E_{F}(s)}$$

$$E_{F}(s)$$

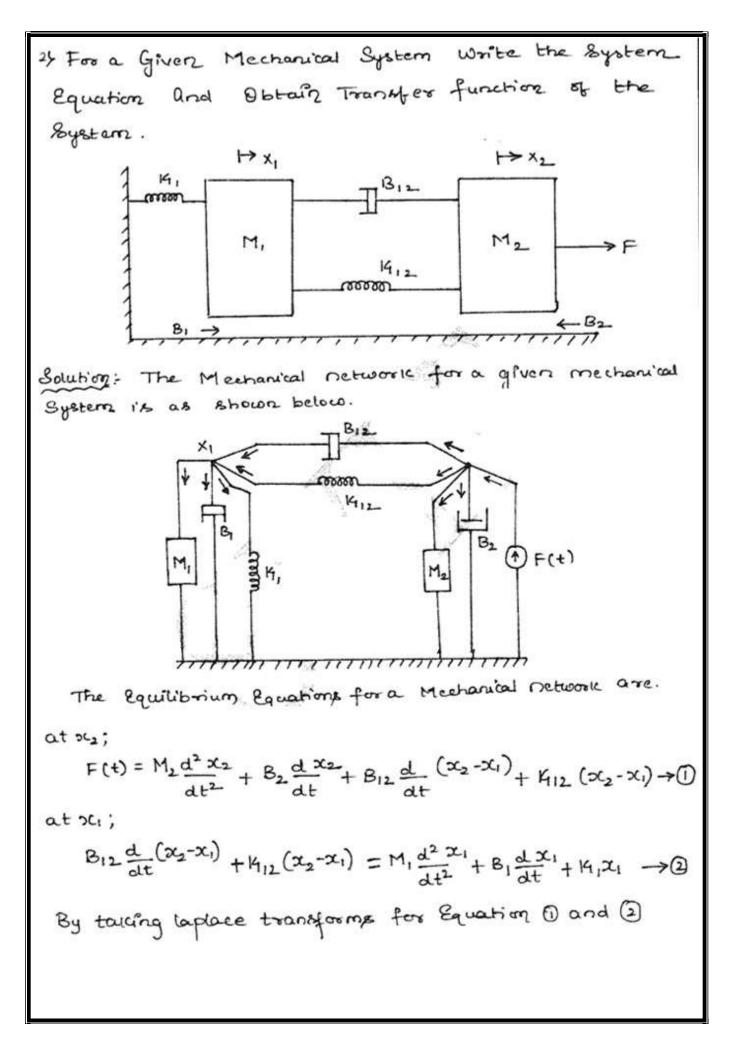
$$R_{1F}(1+st_{F}) = \frac{14m}{B_{m}(1+st_{m})} \cdot \frac{1}{s} = \frac{1}{\theta_{m}(s)}$$

$$Blow diagram of field (potrolled DC motor)$$$$





Froblems
$$\theta_{2}$$
 Transfer function for the hystem Shown
is the figure.
But $\forall x_{1}$
 $\exists x_{1}$
 $\exists x_{2}$
 $\exists x_{1}$
 $\exists x_{2}$
 $\exists x_{2}$
 $\exists x_{3}$
 $\exists x_{4}$
 $\exists x_{5}$
 $\exists x_{1}$
 $\exists x_{1}$
 $\exists x_{1}$
 $\exists x_{1}$
 $\exists x_{2}$
 $\exists x_{1}$
 $\exists x_{2}$
 $\exists x_{2}$
 $\exists x_{2}$
 $\exists x_{2}$
 $\exists x_{3}$
 $\exists x_{4}$
 $\exists x_{1} - \exists x_{2} = dx_{2} + \exists x_{2} + \exists x_{2}$



$$\begin{split} \overline{Fex} \underline{X_{3}} : \\ F(\underline{s}) &= M_{2} \underline{S^{2}} \underline{X}(\underline{s}) + \underline{B_{3}} \underline{S} \underline{X}(\underline{s}) + B_{12} \underline{S} (\underline{X}_{3}(\underline{s}) - \underline{X}_{1}(\underline{s})) + \underline{K}_{12} (\underline{X}_{3}(\underline{s}) - \underline{X}_{1}(\underline{s})) \\ F(\underline{s}) &= \left[M_{3} \underline{s^{2}} + \underline{B_{3}} \underline{S} \right] \underline{X}_{2}(\underline{s}) + \left[\underline{X}_{2}(\underline{s}) - \underline{X}_{1}(\underline{s}) \right] \left[\underline{S} \underline{B}_{12} + \underline{K}_{12} \right] \\ \overline{F(\underline{s})} &= \left(M_{3} \underline{s^{2}} + \underline{B_{3}} \underline{S} \right) \underline{X}_{2}(\underline{s}) + (\underline{S} \underline{B}_{12} + \underline{K}_{12}) \underline{X}_{3}(\underline{s}) - (\underline{S} \underline{B}_{12} + \underline{K}_{12}) \underline{X}_{3}(\underline{s}) \right] \rightarrow \underline{S}} \\ \overline{Fex} \underline{X}_{1} : \\ \overline{F(\underline{s})} &= \left(M_{3} \underline{s^{2}} + \underline{B_{3}} \underline{S} \right) \underline{X}_{2}(\underline{s}) + (\underline{S} \underline{B}_{12} + \underline{K}_{12}) \underline{X}_{3}(\underline{s}) - (\underline{S} \underline{B}_{12} + \underline{K}_{12}) \underline{X}_{3}(\underline{s}) \right] \rightarrow \underline{S}} \\ \overline{Fex} \underline{X}_{1} : \\ B_{12} \underline{S} (\underline{X}_{2}(\underline{s}) - \underline{X}_{1}(\underline{s})) + \underline{K}_{112} (\underline{X}_{3}(\underline{s}) - \underline{X}_{1}(\underline{s})) = M_{1} \underline{s^{2}} \underline{X}_{1}(\underline{s}) + \underline{B}_{1} \underline{S} \underline{X}_{1}(\underline{s}) + \underline{K}_{1} \underline{X}_{1}(\underline{s}) \\ - \underline{K}_{1}(\underline{s}) \\ B_{12} \underline{S} + \underline{K}_{12} \\ \underline{K}_{12}(\underline{s}) = M_{1} \underline{S^{2}} \underline{X}_{1}(\underline{s}) + \underline{B}_{1} \underline{S} \underline{X}_{1}(\underline{s}) + \underline{K}_{1} \underline{X}_{1}(\underline{s}) \\ - \underline{K}_{1}(\underline{s}) \\ B_{12} \underline{S} + \underline{K}_{12} \\ B_{12} \underline{S} \\ - \underline{K}_{1}(\underline{s}) \\ B_{12} \underline{S} + \underline{K}_{1} \\ B_{12} \underline{S} + \underline{K}_{12} \\ B_{12} \underline{S} \\ - \underline{K}_{12} \\ B_{12} \underline{S} + \underline{K}_{12} \\ B_{12} \underline{S} \\ - \underline{K}_{12} \\ - \underline{K}_{12} \\ B_{12} \\ B_{12} \underline{S} + \underline{K}_{11} \\ B_{12} \underline{S} \\ - \underline{K}_{12} \\$$

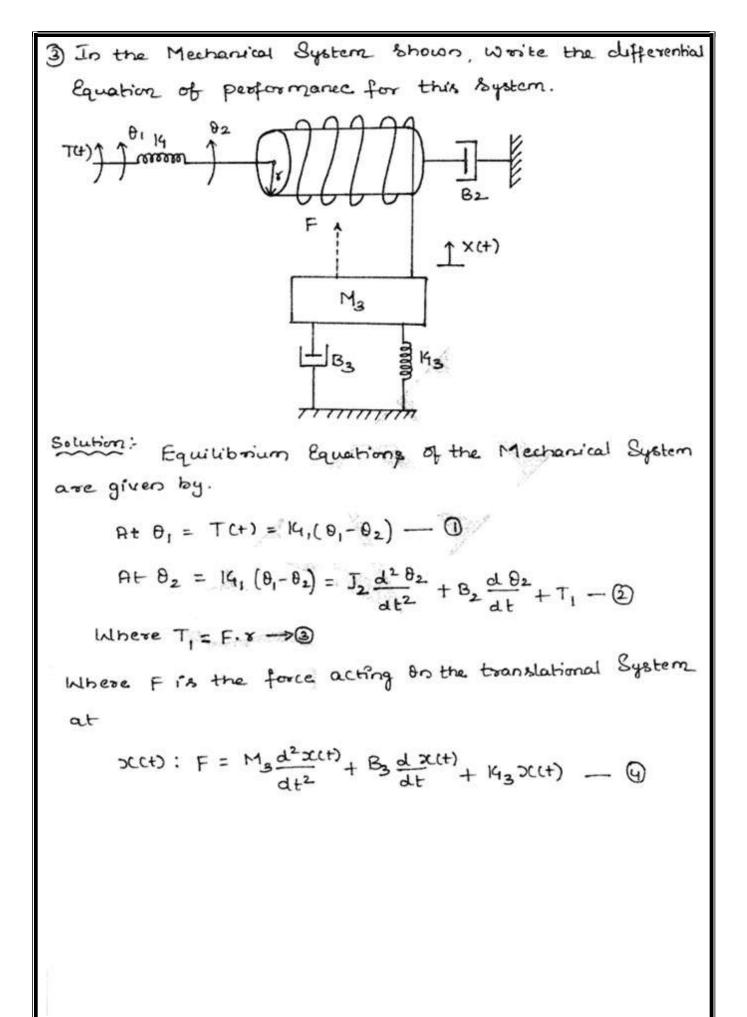
$$\frac{\left[\frac{M_{1}S^{2}+SB_{1}+K_{1}+SB_{12}+K_{12}\right]\left[SB_{12}+K_{12}\right]}{\left[SB_{12}+K_{12}\right]} - \left[SB_{12}+K_{12}\right]}{\left[SB_{12}+K_{12}\right]}$$

$$\frac{F(S)}{X_{1}(S)} = \frac{\left[\frac{M_{1}S^{2}+SB_{1}+K_{1}+SB_{12}+K_{12}\right]\left[SB_{12}+K_{12}\right] + \left(M_{2}S^{2}+B_{2}S\right)}{\left[SB_{12}+K_{12}\right]} - \left[SB_{12}+K_{12}\right]}$$

$$\frac{F(S)}{X_{1}(S)} = \frac{\left[\frac{M_{1}S^{2}+SB_{1}+K_{1}+SB_{12}+K_{12}\right]\left(SB_{12}+K_{12}\right) + \left(M_{2}S^{2}+B_{2}S\right)}{\left(M_{1}S^{2}+SB_{1}+K_{1}+SB_{12}+K_{12}\right) - \left(SB_{12}+K_{12}\right)^{2}\right]}$$

$$SB_{12}+K_{12}$$

$$\frac{SB_{12}+K_{12}}{\left[\frac{M_{2}S^{2}+SB_{1}+K_{1}+SB_{12}+K_{12}\right]\left(SB_{12}+K_{12}\right) + \left(M_{2}S^{2}+B_{2}S\right)}{\left(M_{1}S^{2}+SB_{1}+K_{1}+SB_{12}+K_{12}\right)\left(SB_{12}+K_{12}\right) + \left(M_{2}S^{2}+B_{2}S\right)}{\left(M_{1}S^{2}+SB_{1}+K_{1}+SB_{12}+K_{12}\right) - \left(SB_{12}+K_{12}\right)^{2}\right]}$$



4) Find the transfer function for the system shown
in the figure 0 and (2) hence show that they are.
Onalogous to Each Other.

$$g_{1,2}$$

 $f_{1,2}$
 f

$$\begin{array}{c} + \underbrace{ Z_{1} \\ F_{1}(S) \\ T(S) \\ Z_{2} \\ F_{1} \times \frac{1}{Sc_{1}} \\ F_{1} \times \frac{1}{Sc_{2}} \\ F_{2} = \frac{R_{12} + \frac{1}{Sc_{2}} = \frac{R_{12}Sc_{2}}{Sc_{2}} + 1 \\ T(S) = \frac{C_{1}(S)}{Z_{1} + Z_{2}} \\ F_{0}(S) = T_{2}T(S) \\ F_{0}(S) = \frac{Z_{2}}{Z_{1} + Z_{2}} \\ \frac{F_{0}(S)}{F_{1}(S)} = \frac{Z_{1}}{Z_{1} + Z_{2}} \\ \frac{F_{0}(S)}{F_{1}(S)} = \frac{\left(\frac{R_{2}Sc_{2} + 1}{Sc_{2}}\right)}{\frac{R_{1}}{R_{1}Sc_{1}} + \frac{R_{2}Sc_{2} + 1}{Sc_{2}}} \\ = \frac{(R_{1}Sc_{1} + 1)(R_{12}Sc_{2} + 1)}{R_{1}Sc_{2} + (R_{1}Sc_{1} + 1)(R_{2}Sc_{2} + 1)} \\ = \frac{C_{1}(R_{1}S + \frac{1}{c_{1}})C_{2}(R_{2}S + \frac{1}{c_{2}})}{R_{1}Sc_{2} + R_{1}R_{2}S^{2}c_{1}c_{2} + R_{1}Sc_{1} + R_{2}Sc_{2} + 1} \\ = \frac{C_{1}C_{2}(R_{1}R_{2}S^{2} + \frac{R_{1}S}{c_{1}} + \frac{R_{1}S}{c_{2}} + \frac{R_{1}S}{c_{1}} + \frac{R_{2}S}{c_{2}} + \frac{1}{c_{1}c_{2}}} \end{array}$$

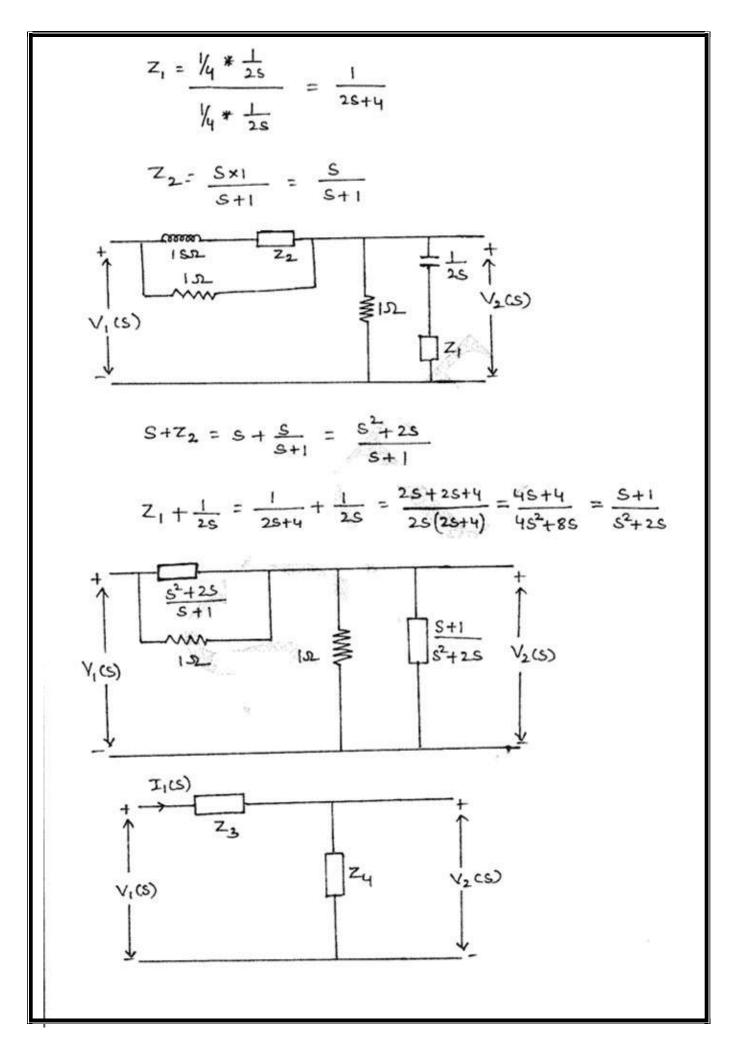
$$\begin{array}{c} \hline \hline E_{0}(S) \\ \hline E_{1}(S) \end{array} = \frac{\left(P_{1}S + \frac{1}{c_{1}}\right) \left(P_{1}S + \frac{1}{c_{2}}\right)}{P_{1}P_{1}S^{2}S^{2} + S\left(\frac{P_{11}}{c_{2}} + \frac{P_{11}}{c_{1}} + \frac{P_{12}}{c_{1}}\right) + \frac{1}{c_{1}}C_{2}} \rightarrow \textcircled{6} \end{array}$$
Equation (§) and (§) are mathematically scinitian bance.
the two System with be analogous.

$$\begin{array}{c} F_{1} = B_{1} \\ C_{1} = \frac{1}{K_{1}} \end{array}, \quad C_{2} = \frac{1}{K_{12}} \end{array}$$
57 For the two port network shown in figure obtain the transfer function.
(i) $V_{2}(S)$ (ii) $V_{1}(S)$

$$\begin{array}{c} V_{1}(S) \\ V_{1}(S) \end{array} \qquad \begin{array}{c} F_{1} = B_{2} \\ C_{1} = \frac{1}{K_{1}} \end{array}, \quad C_{2} = \frac{1}{K_{12}} \end{array}$$
58 For the two port networks shown in figure obtain the transfer function.
(i) $V_{2}(S)$ (ii) $V_{1}(S)$

$$\begin{array}{c} T_{1} = 00000 \\ T_{1}(S) \end{array}$$

$$\begin{array}{c} T_{1} = 0000 \\ T_{1}(S) \end{array}$$



$$Z_{3} = \frac{\left(\frac{S^{2}+2S}{S+1}\right) \times 1}{\frac{S^{2}+2S}{S+1} + 1} = \frac{S^{2}+2S}{S^{2}+3S+1}$$

$$Z_{4} = \frac{1\times \frac{(S+1)}{S^{2}+2S}}{1+\frac{S+1}{S^{2}+2S}}$$

$$Z_{4} = \frac{S+1}{\frac{S^{2}+2S+1+S}{S^{2}+2S+1+S}} = \frac{S+1}{S^{2}+3S+1}$$

$$I_{1}(S) = \frac{V_{1}(S)}{Z_{3}+Z_{4}}$$

$$V_{2}(S) = I_{1}(S) + Z_{4}$$

$$V_{2}(S) = \frac{V_{1}(S)}{Z_{3}+Z_{4}} \cdot Z_{4}$$

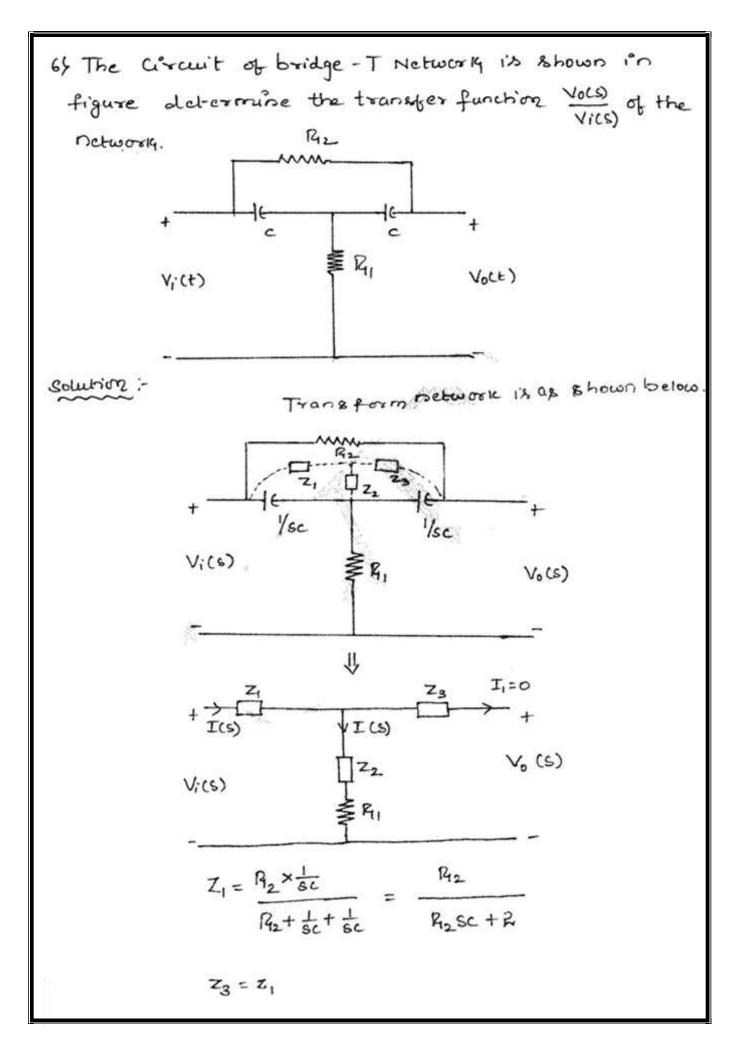
$$\frac{V_{2}(S)}{V_{1}(S)} = \frac{Z_{4}}{Z_{3}+Z_{4}}$$

$$\frac{V_{1}(S)}{I_{1}(S)} = Z_{3}+Z_{4}$$

$$Z_{3}+Z_{4} = \frac{S^{2}+2S}{S^{2}+3S+1} + \frac{S+1}{S^{2}+3S+1}$$

$$Z_{3}+Z_{4} = I$$

$$\cdots \qquad \frac{V_{2}(S)}{V_{1}(S)} = \frac{Z_{4}}{I} = \frac{S+1}{S^{2}+3S+1} \quad \text{f} \qquad \frac{V_{1}(S)}{I_{1}(S)} = I$$



$$Z_{2} = \frac{\frac{1}{8c} \times \frac{1}{8c}}{R_{12} + \frac{1}{8c} + \frac{1}{5c}} = \frac{1}{sc(R_{12}s(t+2))}$$

$$I(s) = \frac{Vi(s)}{Z_{1} + Z_{2} + R_{1}}$$

$$V_{0}(s) = I(s)(Z_{3} + R_{1})$$

$$V_{0}(s) = \frac{Vi(s)}{Z_{1} + Z_{2} + R_{1}} \cdot Z_{2} + R_{1}$$

$$\frac{V_{0}(s)}{V_{i}(s)} = \frac{Z_{2} + R_{1}}{Z_{1} + Z_{2} + R_{1}}$$

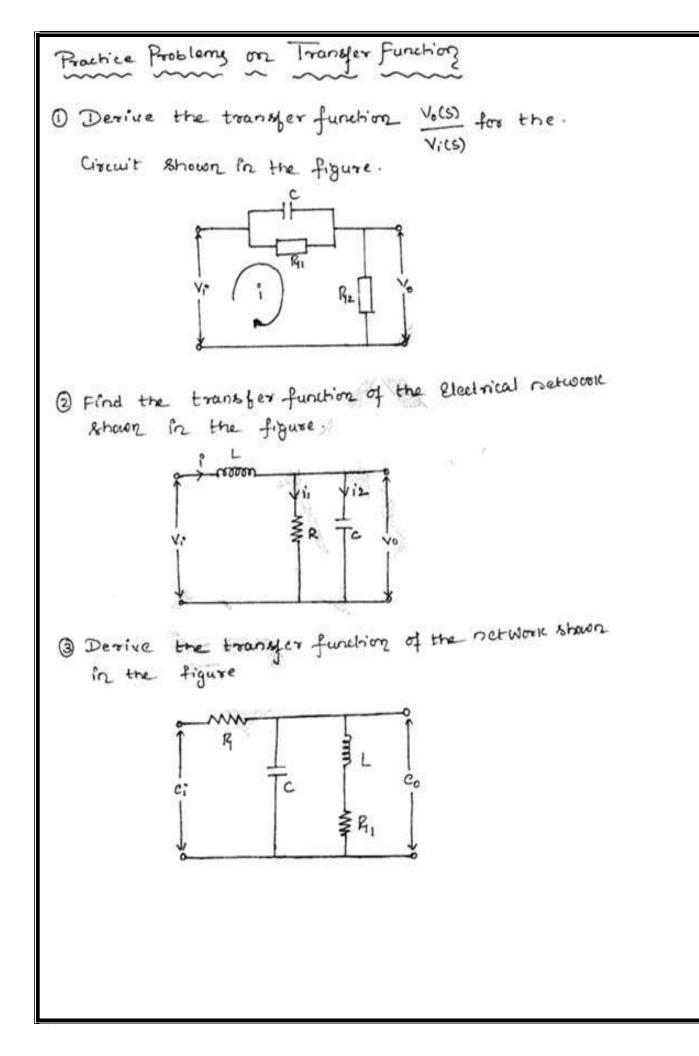
$$Z_{1} + Z_{2} = \frac{R_{12}}{R_{1} + R_{2}sc} + \frac{1}{sc(2 + R_{2}sc)}$$

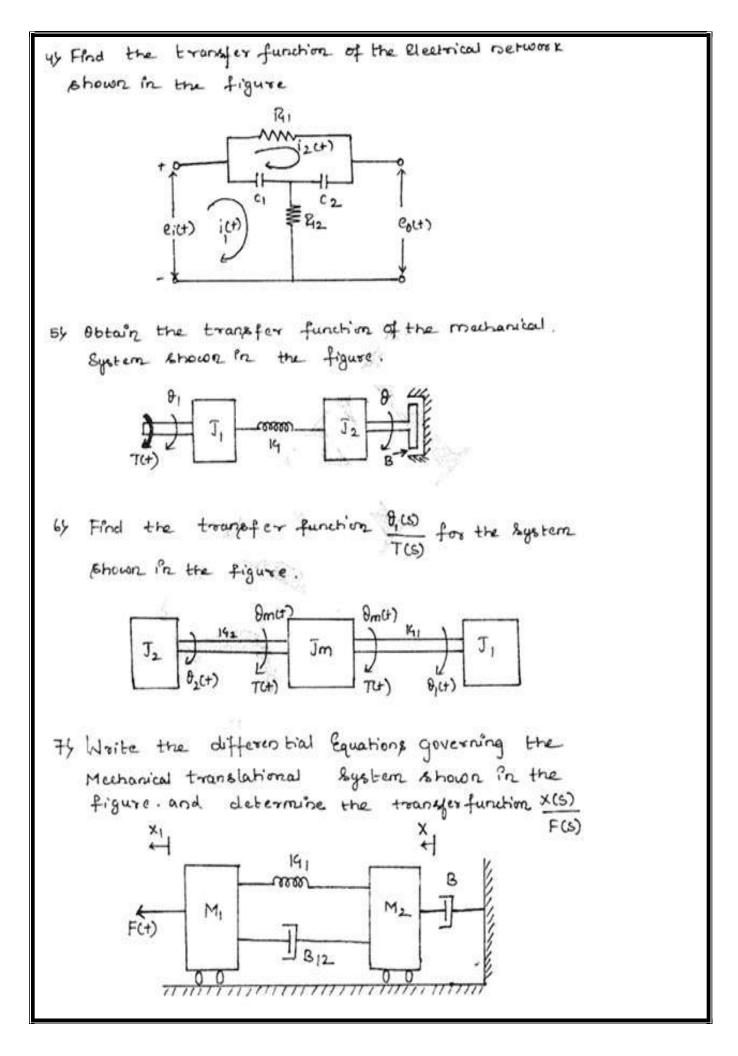
$$= \frac{R_{1}sc + 1}{sc(2 + R_{2}sc)}$$

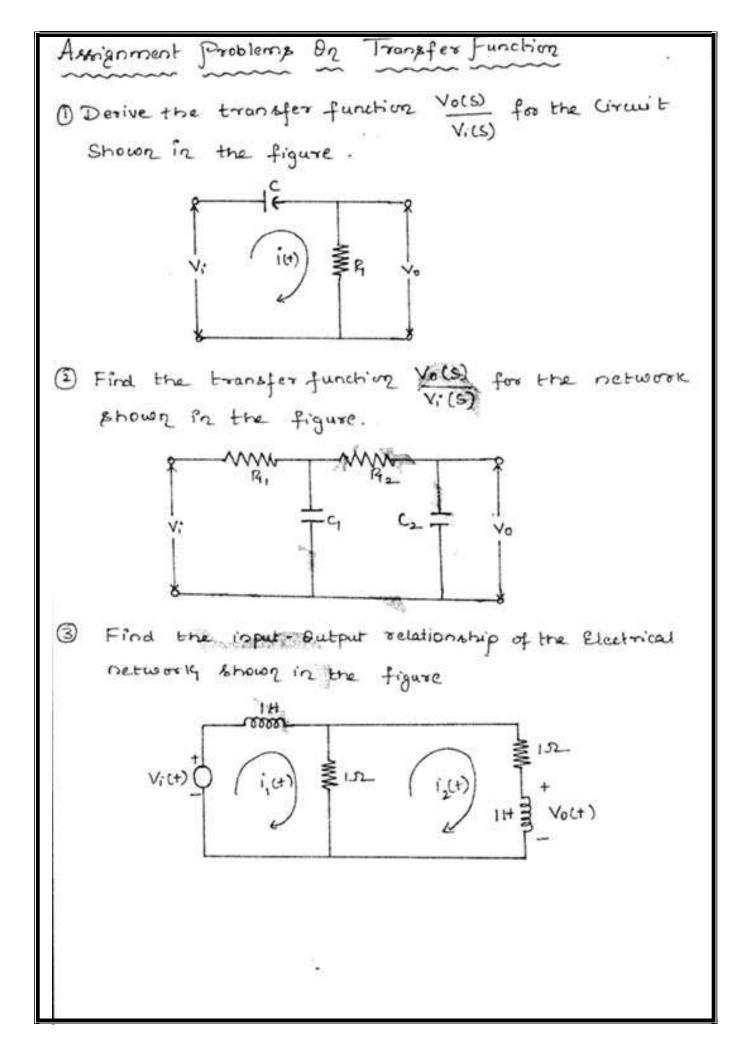
$$\frac{V_{0}(s)}{V_{i}(s)} = \frac{1}{sc(2 + R_{2}sc)} + R_{1}$$

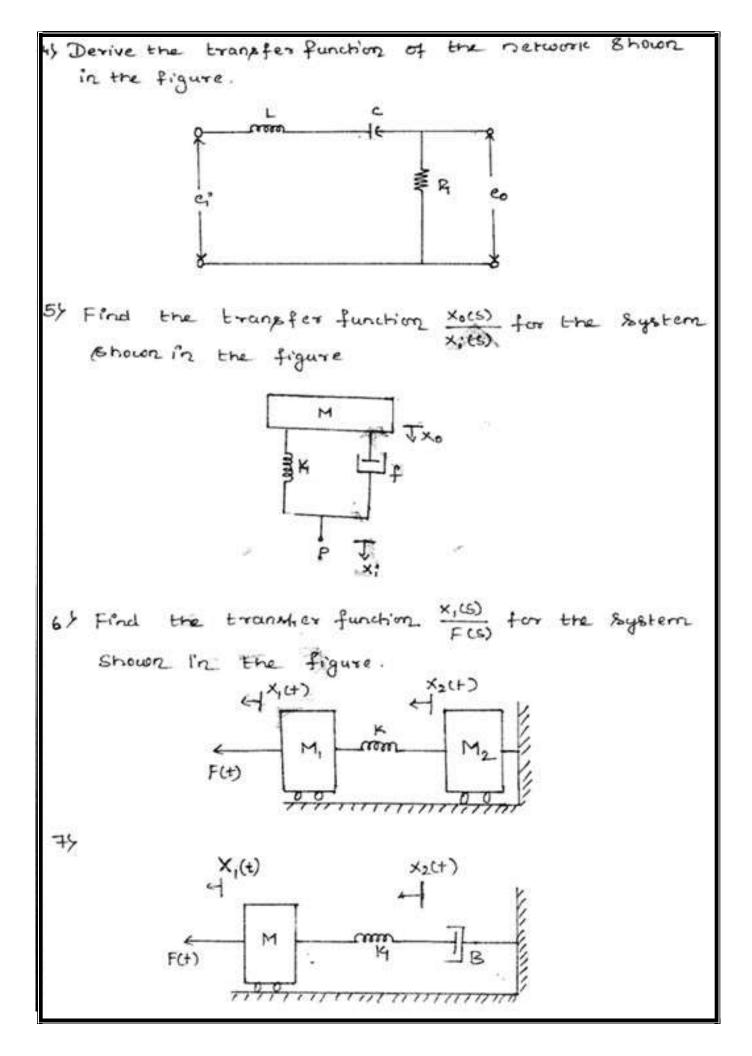
$$\frac{V_{0}(s)}{V_{i}(s)} = \frac{1 + R_{1}sc(2 + R_{2}sc)}{V_{i}(s)} + R_{1}$$

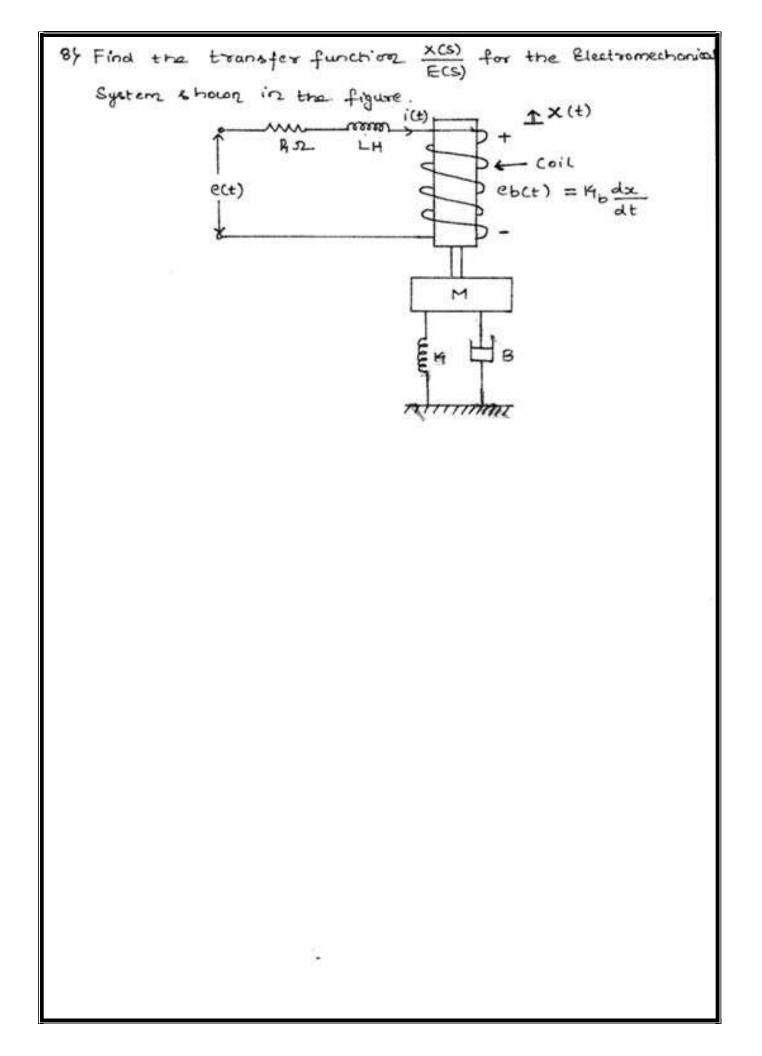
$$\frac{V_{0}(s)}{V_{i}(s)} = \frac{1 + R_{1}sc(2 + R_{2}sc)}{R_{2}sc + 1 + R_{1}sc(2 + R_{2}sc)}$$











Blour Diagram Algebra * In this section we discuss development of block diagram for the Systems. * A system consists of number of components, the function of Each component is represented by a block. * All the blocks are inter connected by lines with arrows indicating the flow of Signals from the Sutput of one block to the input of another. * Such a block diagram giver an overall idea of the inter-relationships that Exist among Various Components. System \rightarrow (t)x (t) -Butput Input Laplace - Transform + c(s) R(S) * Let up consider a System with transfer function G(S) * The System Can be represented by a block as shown in_ the figure above. The input signal into the block is R(s) Which is the Laplace transform of Input Signal r(t) The Butput Signal of the block is C(S) Which is the * laplace transform of the Dutput Signal c(t).

* The flow of leignal is unidirectional from the input to the Output. The Output ((s) is Equal to the convolution of the input signal and transfer function G(s),

i.e, C(S) = G(S) R(S)

⇒ Block Diagram Transformation
Is Feedback Control System (Eliminating feedback loop).
Comparator ing print
Freedback path
R(S) + (ES) (G(S)) (G(S)) (CS) Controlled Dutput
Righterence i/p B(S) (G(S)) (S) (S) (S) (S)
feedback path
The Entrope Arignal E(S) is given by
E(S) = R(S) → B(S) - 0
W.K.T (G(S) =
$$\frac{K(S)}{E(S)}$$

or $B(S) = H(S) \cdot C(S)$
By Substrictuting the Value of $E(S) + B(S)$ in 0
 $\frac{C(S)}{G(S)} = R(S) - H(S) - C(S)$
C(S) = $R(S) + R(S) - G(S) + R(S)$
C(S) [1 + $G(S) + R(S)$] = $G(S) R(S)$
 $\frac{C(S)}{S(S)} = \frac{G(S)}{S(S)} + \frac{G$

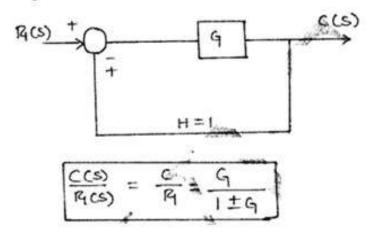
$$R_{(S)} \xrightarrow{G} C(S)$$

 $I \pm GH$

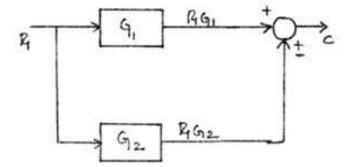
((s) is 14 nows as the overall transfer function or R(s)

croked loop transfer function.

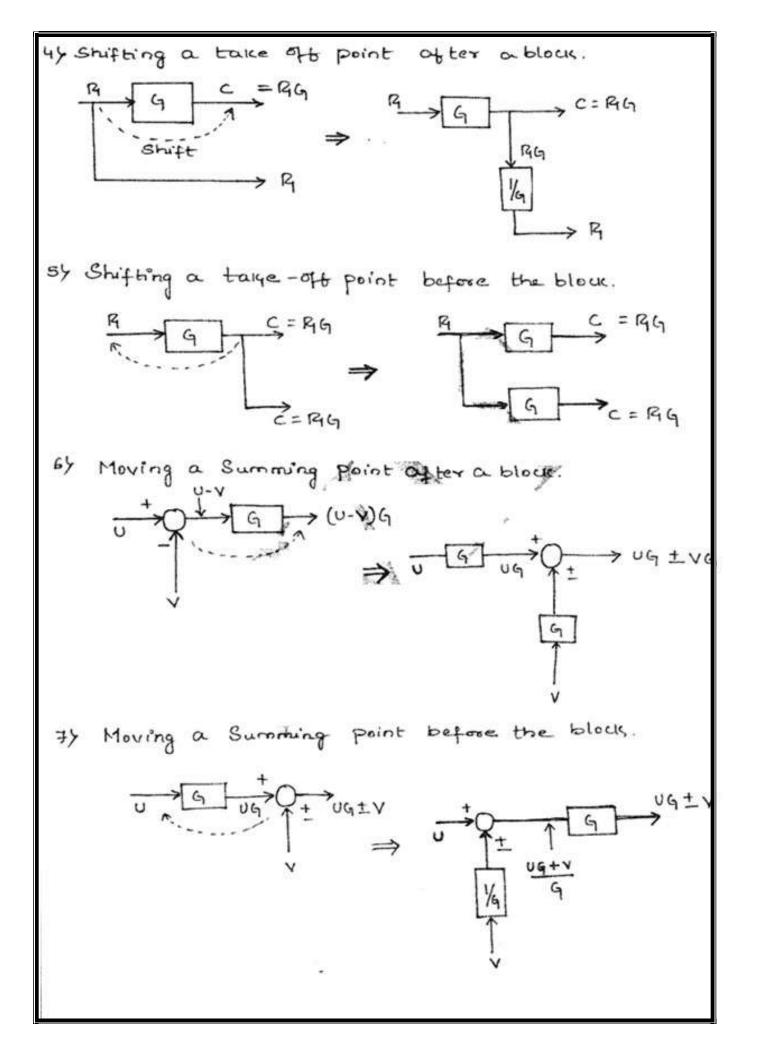
If H=1, then it is said to be Unity feed back Control System, shown in the figure.

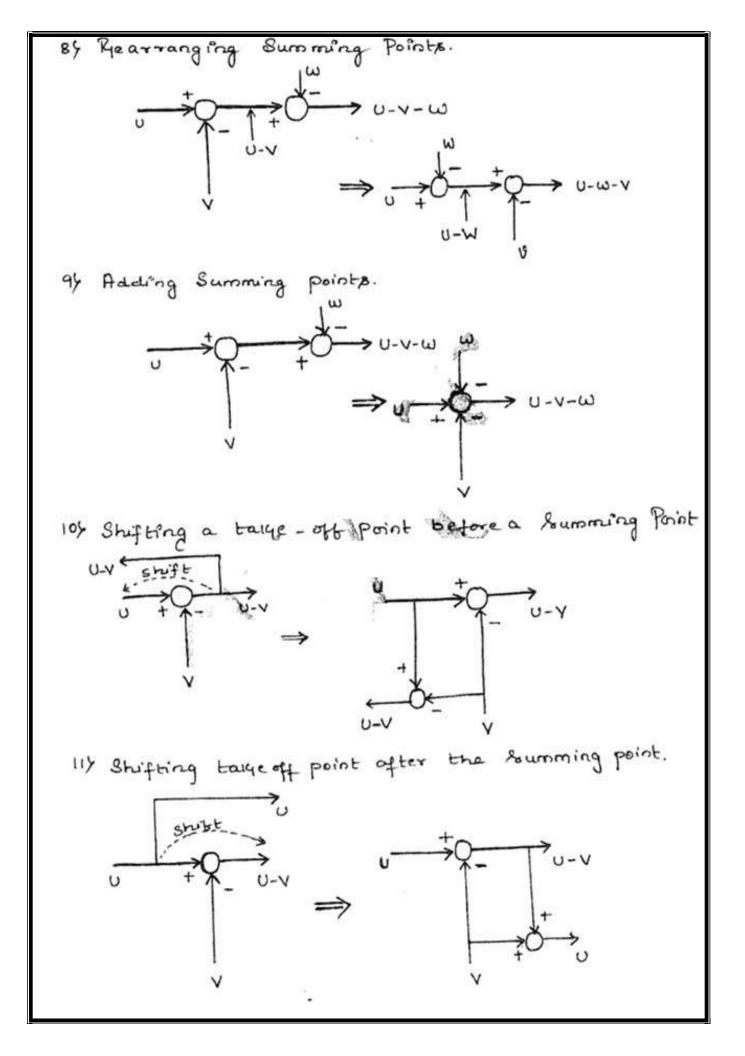


24 Blocks in cascade $R \rightarrow G_1 \qquad RG_1 \qquad G_2 \rightarrow C$ $C = R G_1 G_2$ $\frac{C}{R} = G_1 G_2$ $\frac{G_1 G_2}{R}$ $\frac{G_1 G_2}{C}$ 34 Blocks in parallel. C = C = C = C = C = C



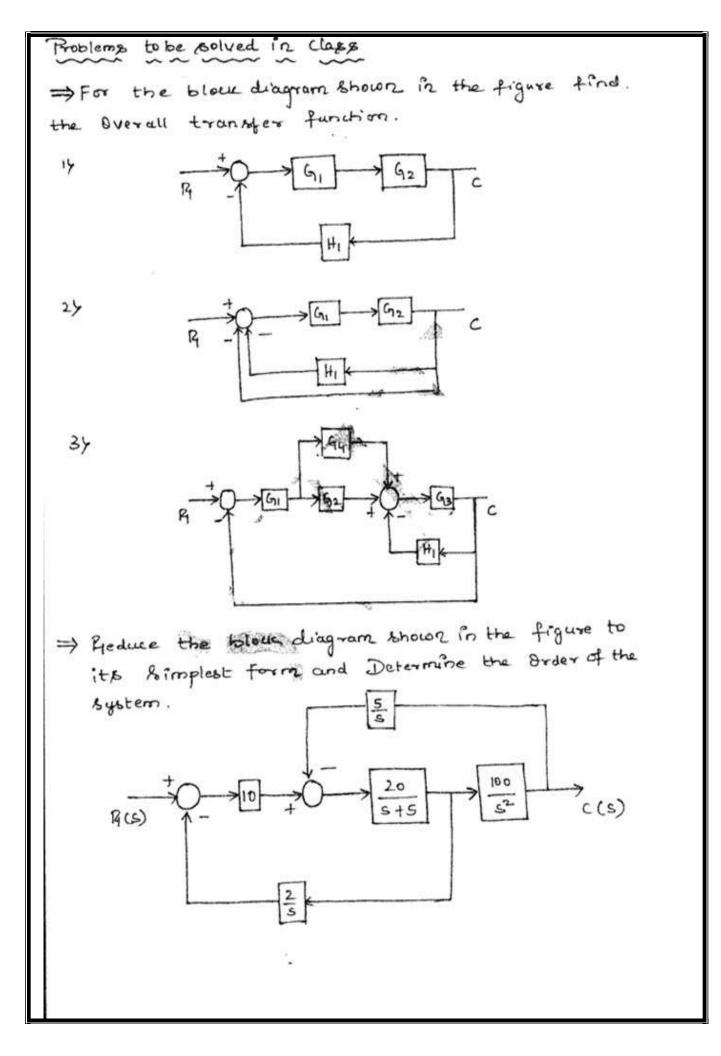
 $C = RG_{1} \pm RG_{2} = R(G_{1} \pm G_{2})$ $\frac{C}{R} = G_{1} \pm G_{2}$ $\xrightarrow{G_{1} \pm G_{2}} = C$ R



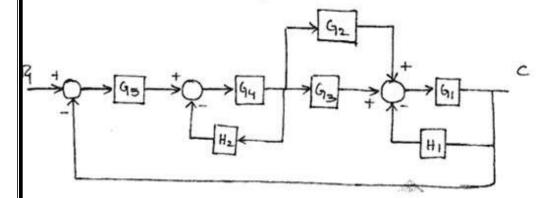


* Stepp for reduction of Complicated Block diagram:
steps: Combine all Cascade blacks.
steps: combine all parallel blocks.
steps: Eliminate all minor feed back loops.
stepu: Shift Summing points to the left and take of
Printse to the right of the major loop.
steps: Repeat steps 1 to 4 until the Canonical form has
been achieved for a particular input.
steps: Repeat steps 100 for Bach input as
required
* Superposition of Multiple Inputs:
-> Some time it is necessary to Evaluate System
Respondence ober several inputs are simultaneous! applied at different points of the System.
-> When multiple inputs are present in a linear
System, Each input is treated independently of the
others.
For Example:-
Determine the transfer functions for the blouchings.
shown below.
$ \xrightarrow{+} \bigcirc \xrightarrow{+} \longrightarrow \xrightarrow{+} \xrightarrow{+} \longrightarrow \xrightarrow{+} \longrightarrow \xrightarrow{+} \longrightarrow \xrightarrow{+} \xrightarrow{+} \longrightarrow \xrightarrow{+} \xrightarrow{+} \xrightarrow{+} \xrightarrow{+} \xrightarrow{+} \xrightarrow{+} \xrightarrow{+} \xrightarrow{+}$
•
p

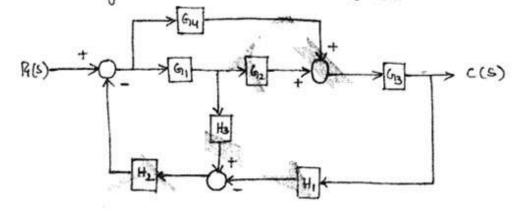
Consider any one input ata nime, Step1: Put U=0, Block diagram its redrawn as shown below. Step 2: 6,62 C H=1 Steps: The Dutput Equation C is given by $\frac{c}{R} = \frac{G_1 G_2}{1 + G_1 G_2}$ By convidening Second Priput (A) Stepu:-Put R = 0 Blour diagram is redrand as shown below. Steps: -> Gat > C & Similar . GI The output transfer function is given by $\frac{C}{U} = \frac{G_2}{1+G_1G_1}$ The total output is given by $C = \begin{bmatrix} c_{11} & c_{12} \\ \hline 1 + c_{11} & c_{12} \end{bmatrix} R_{1} + \begin{bmatrix} c_{12} \\ \hline 1 + c_{12} & c_{11} \end{bmatrix} U$



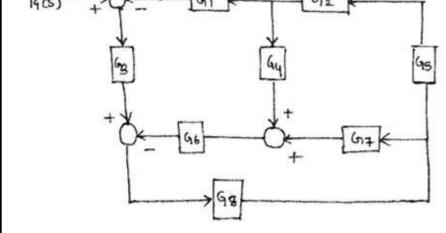
=> Find the Overall transfers functions for the block diagram bhown below. Upling block diagram reductions technique.

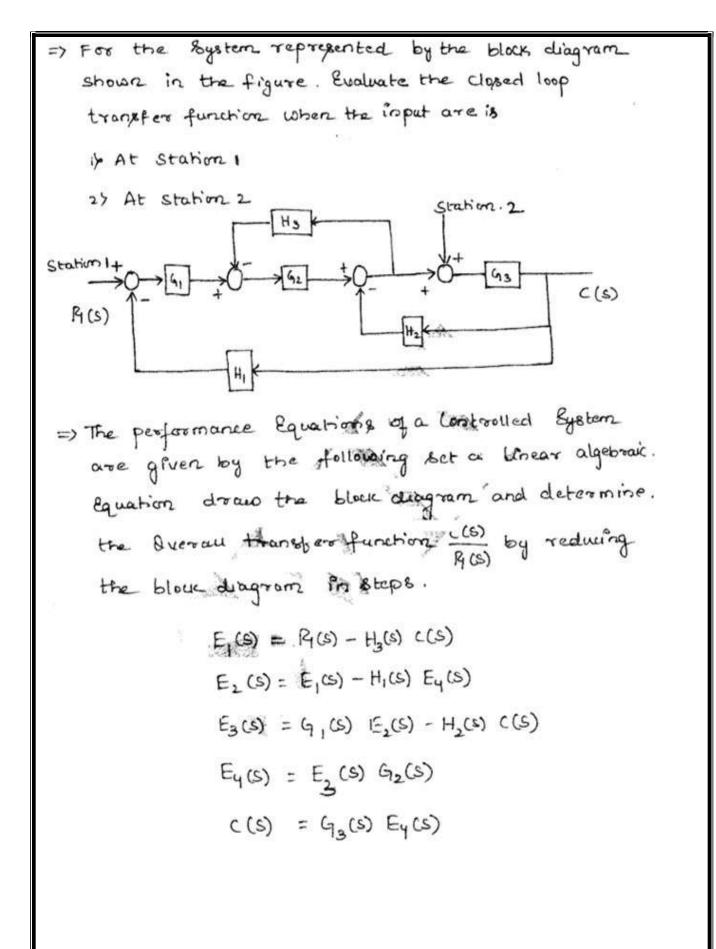


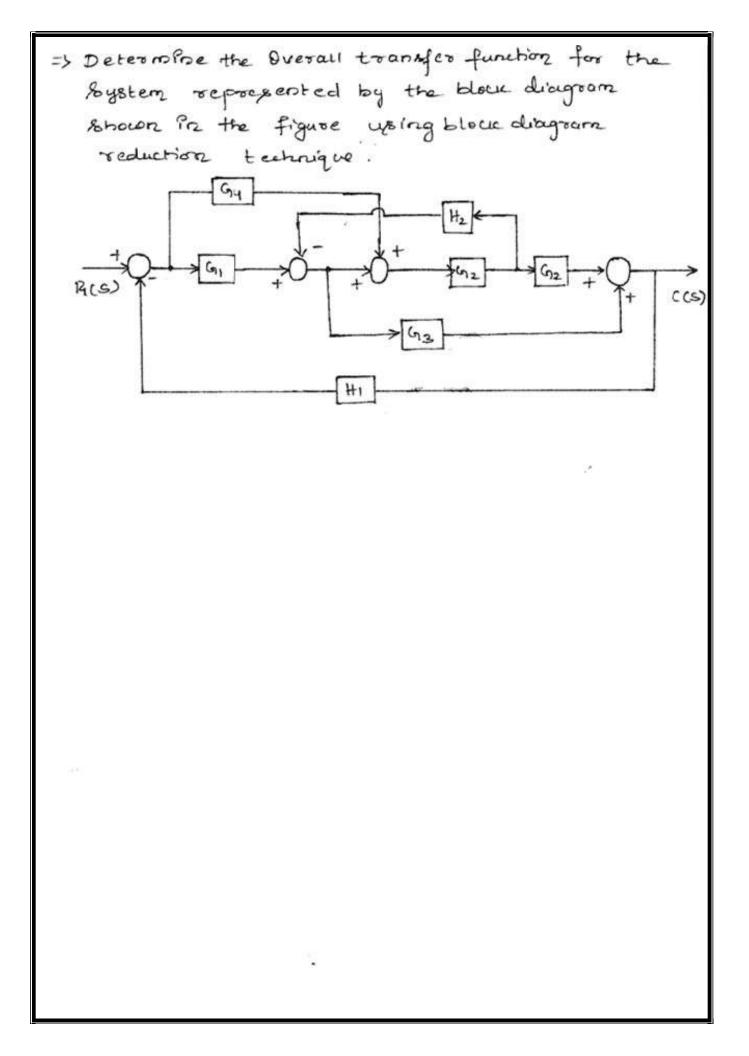
= Determine the Overall transfer function for the. block diagram shown in the figure.

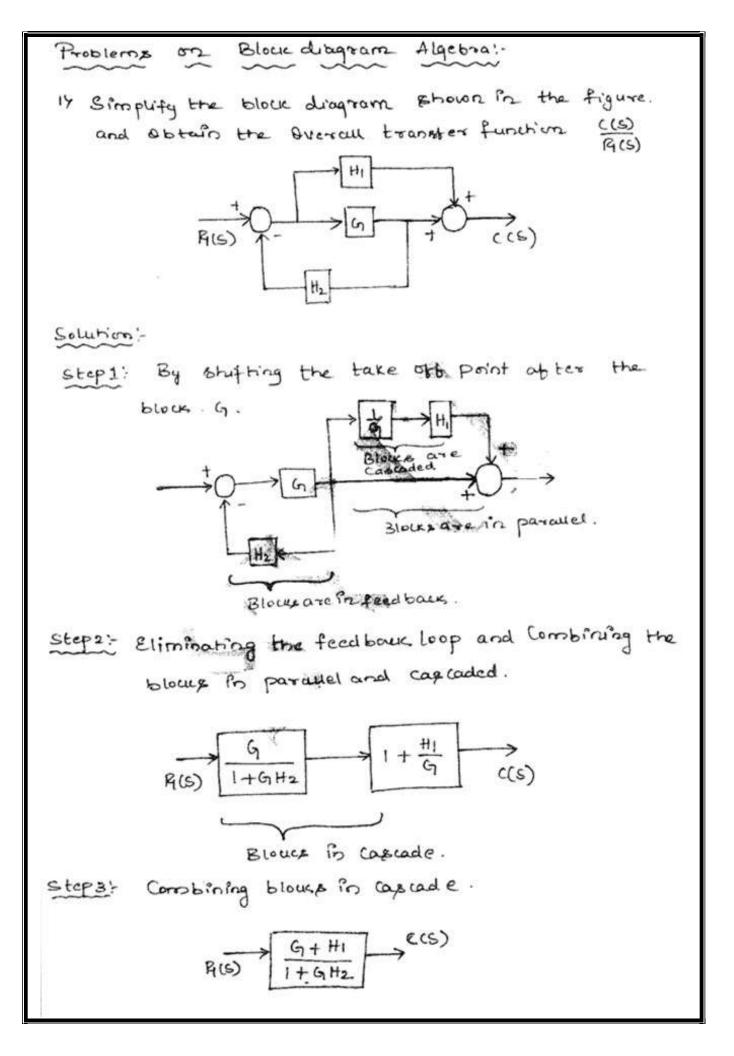


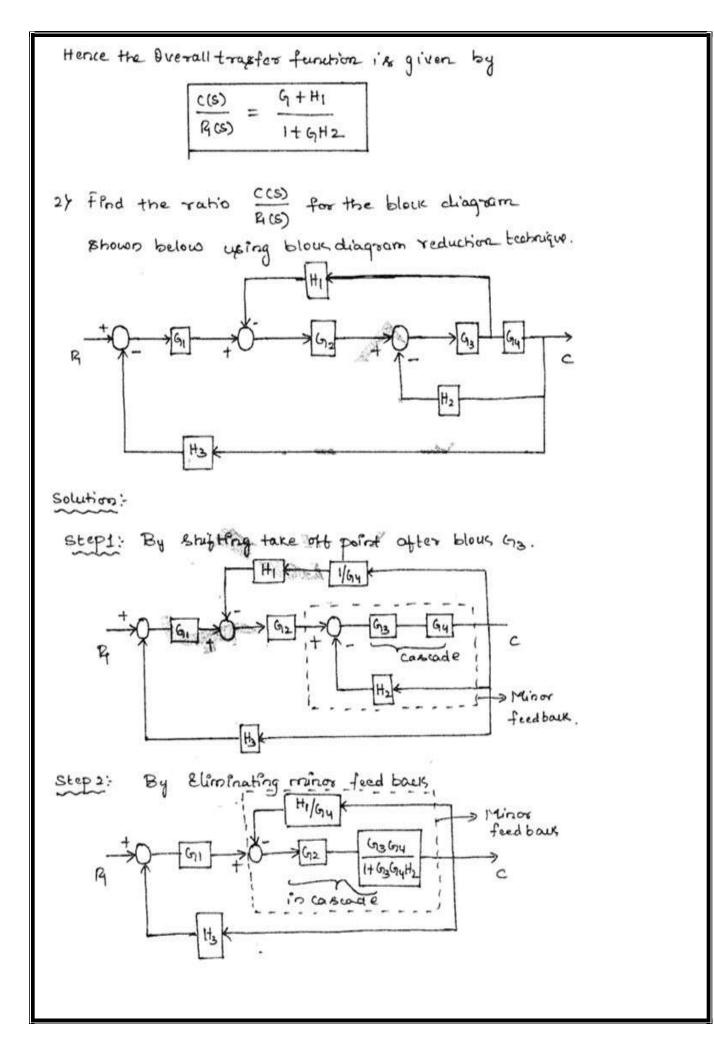
=> Find the Overall transfer function $\frac{C(S)}{R(S)}$ for the. Block diagram shown in the figure. $\int C(S)$ R(S) $\rightarrow O$ G_1 G_2 G_3

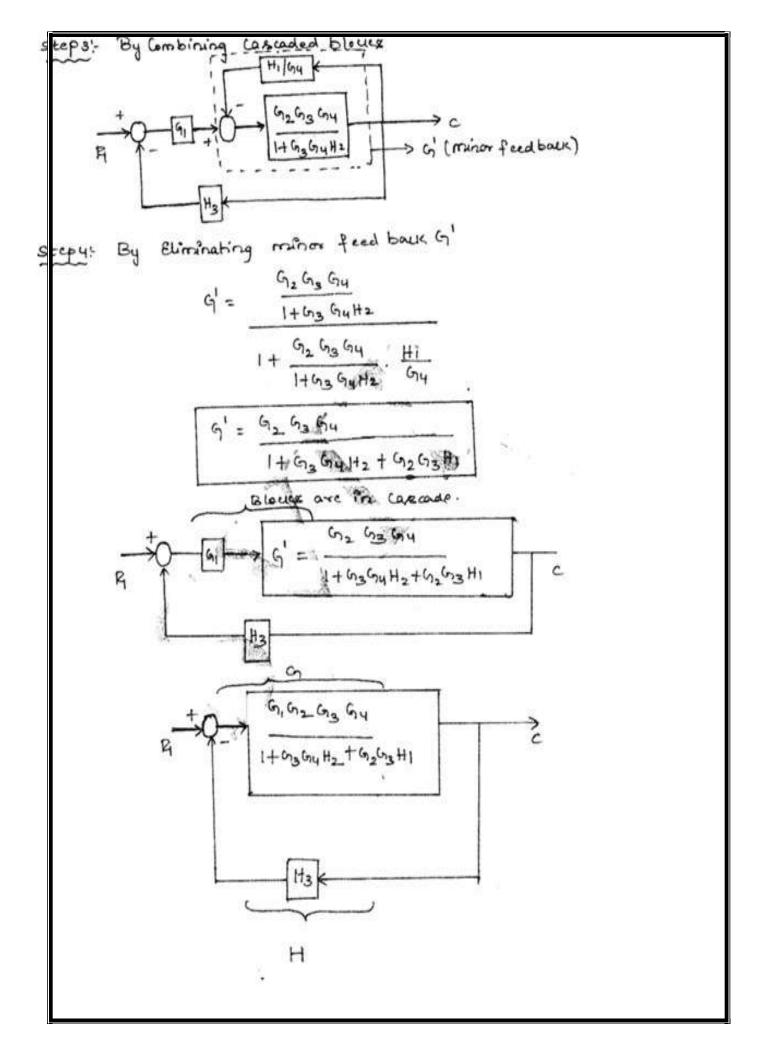


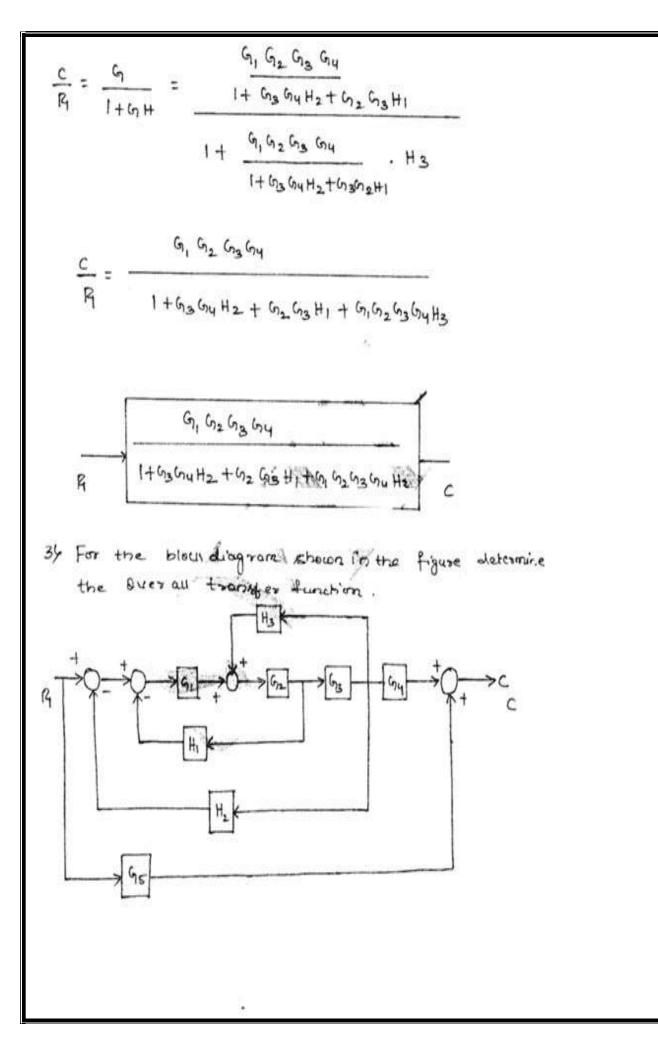


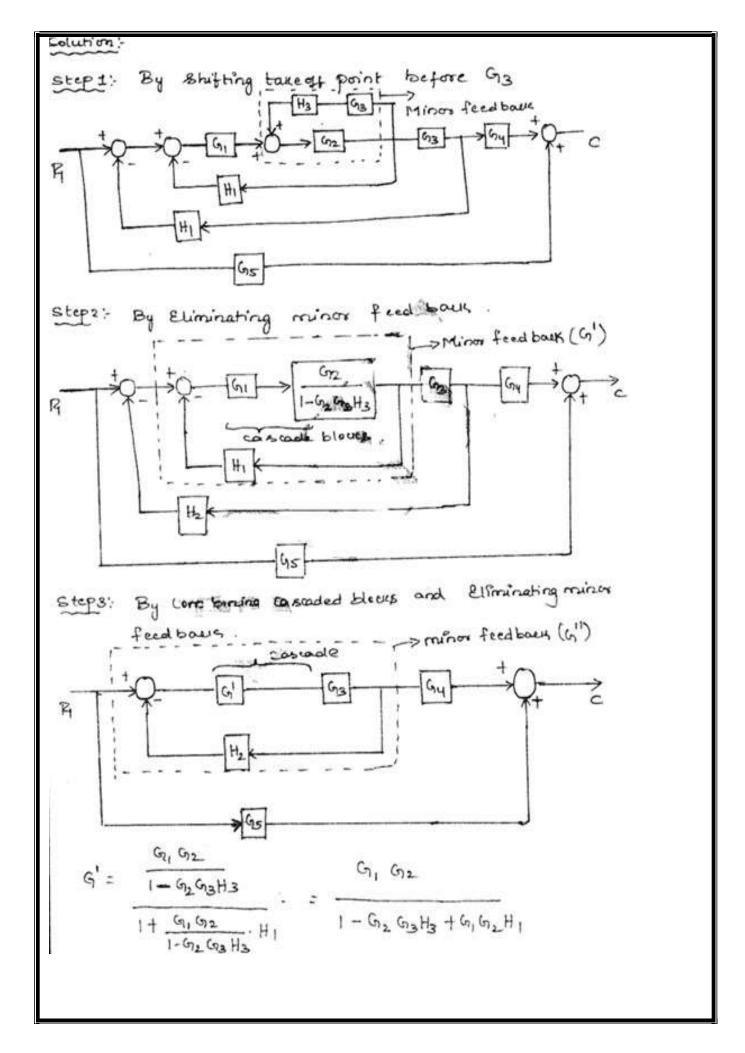


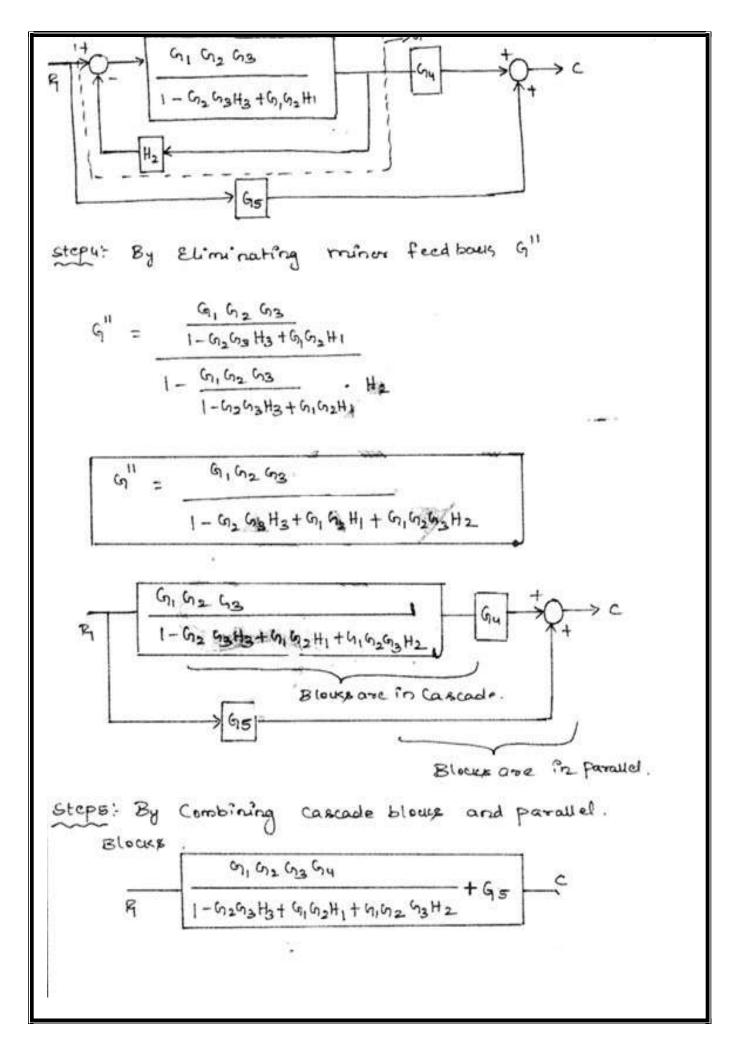


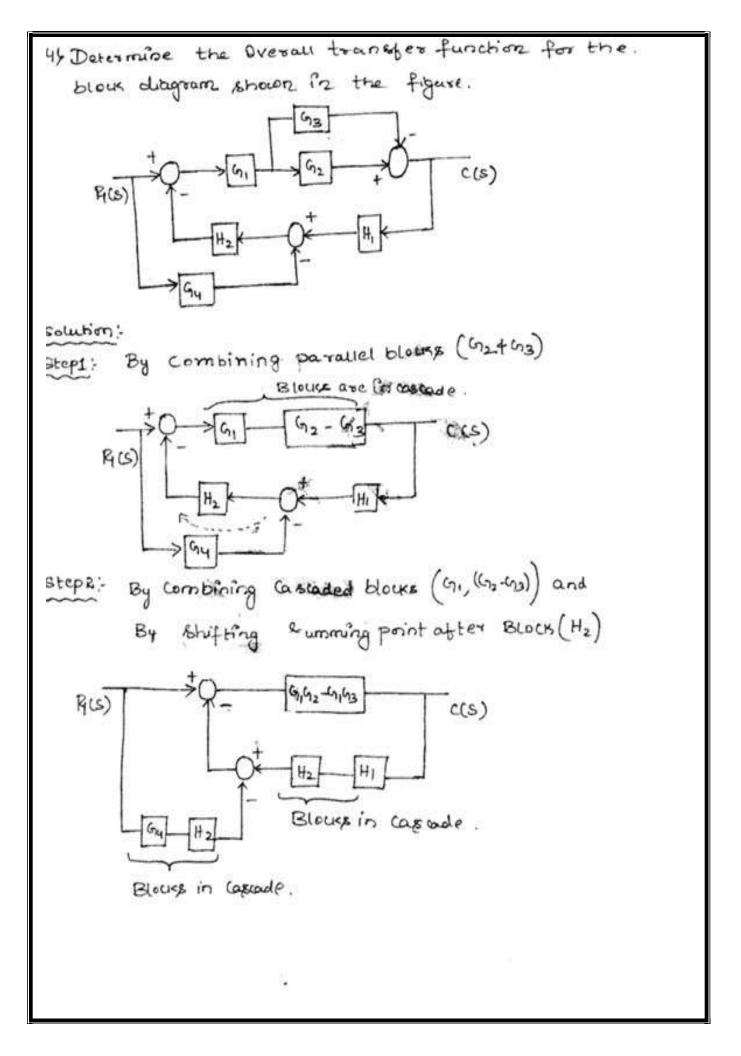


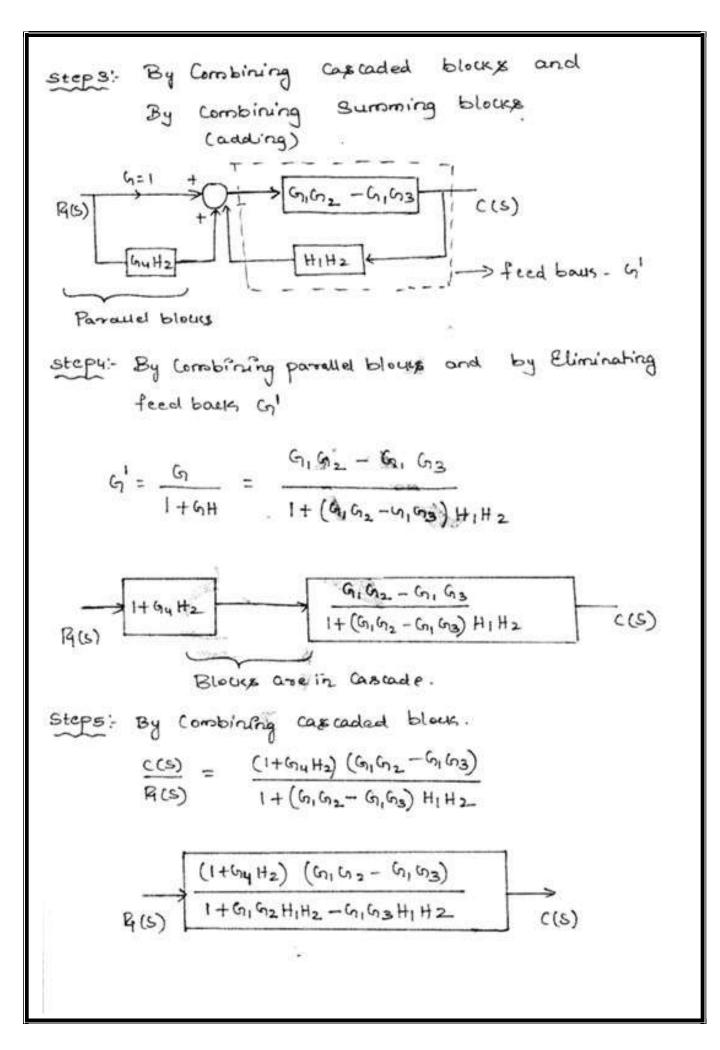


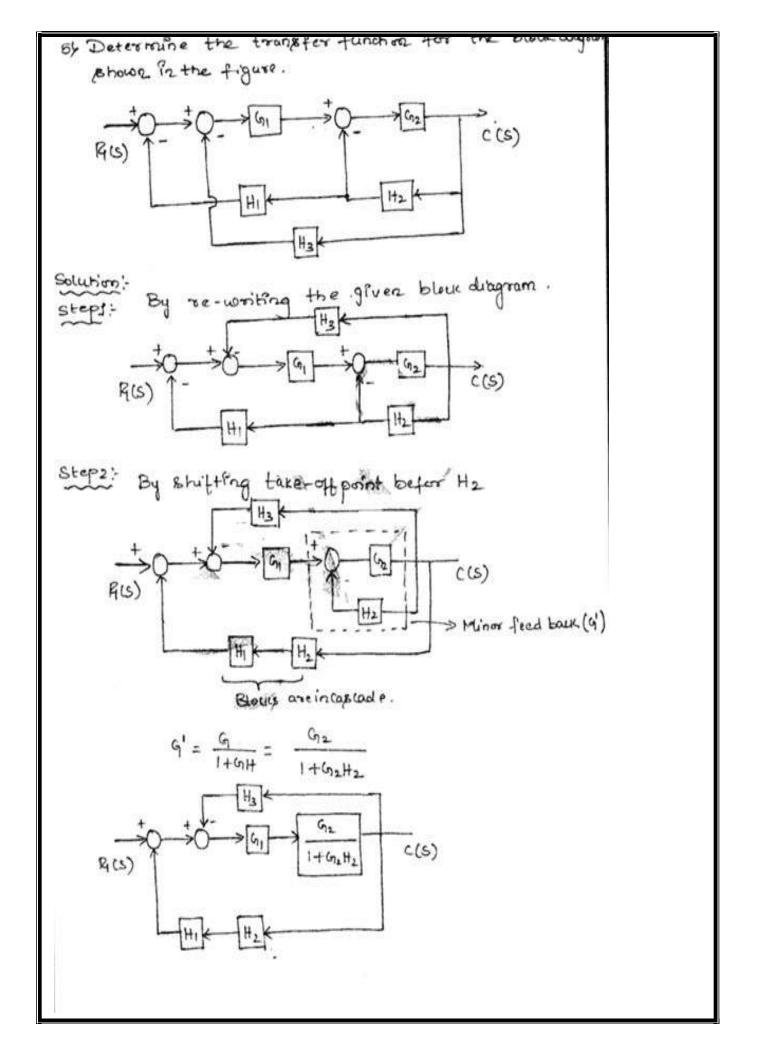


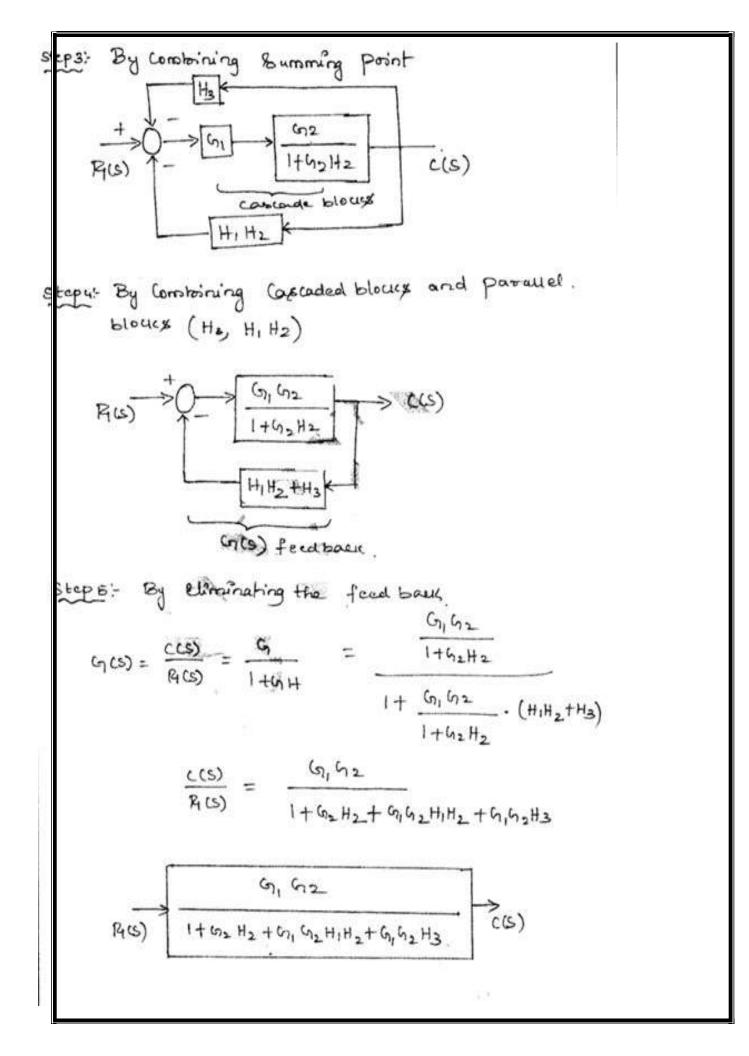


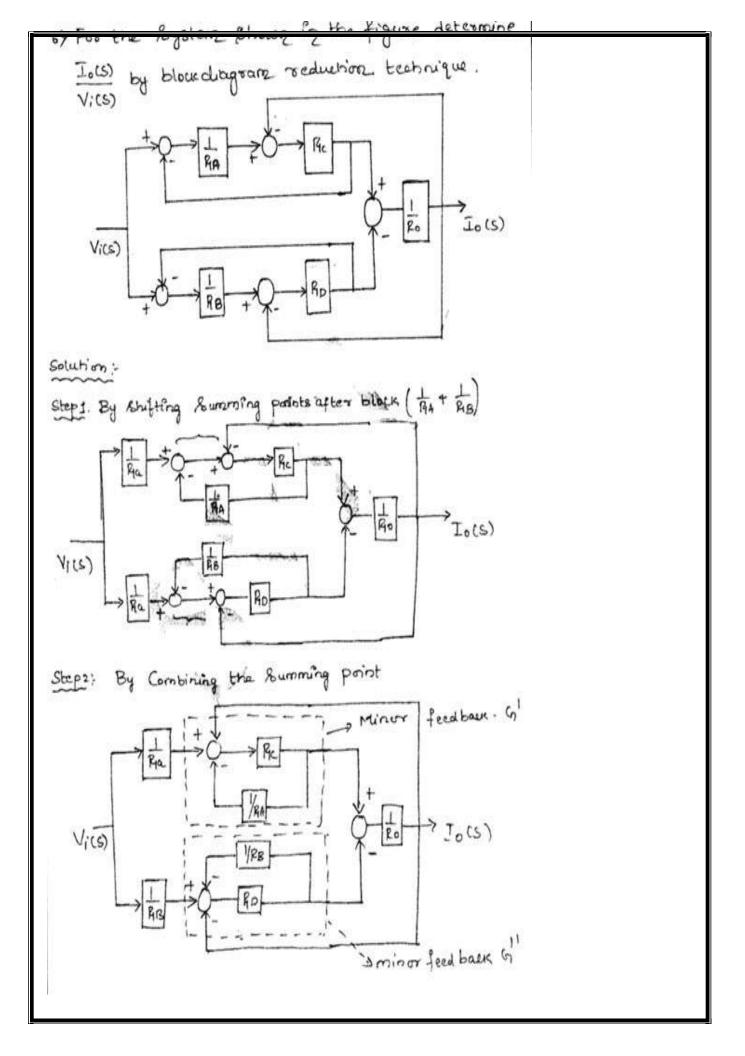


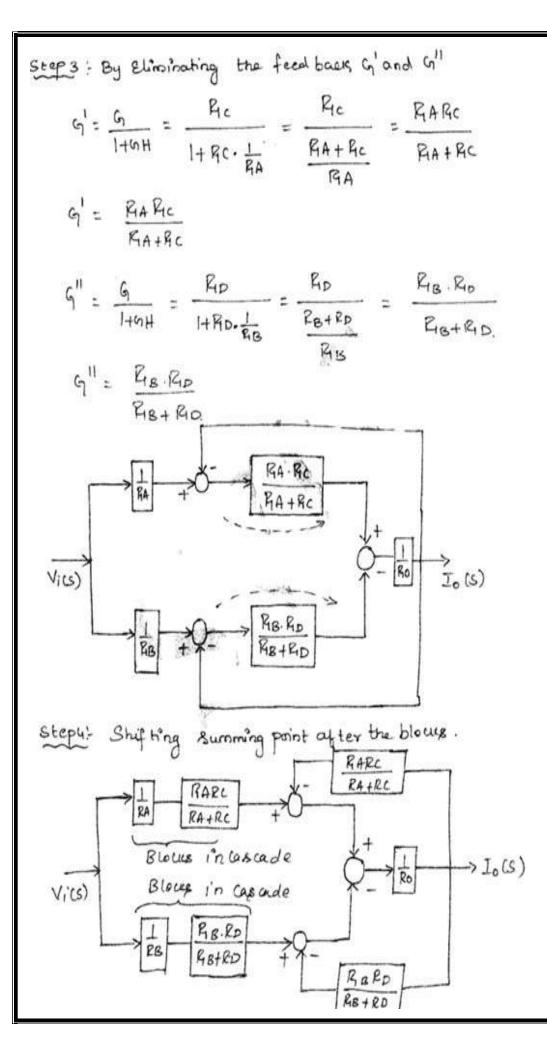


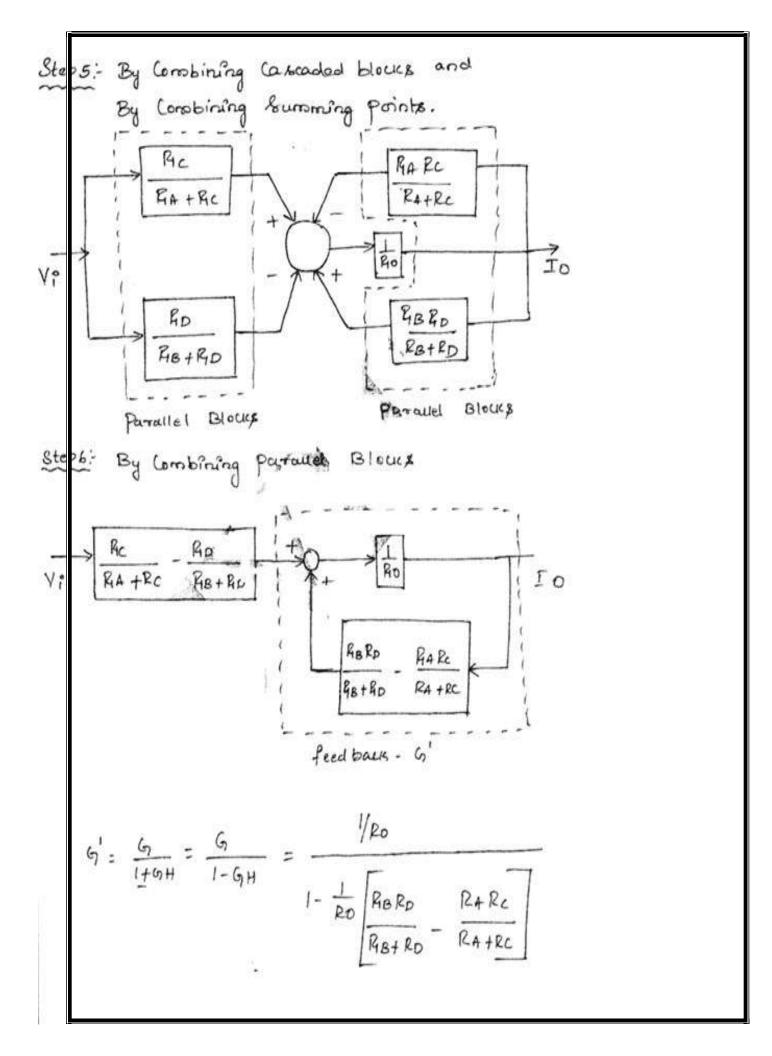




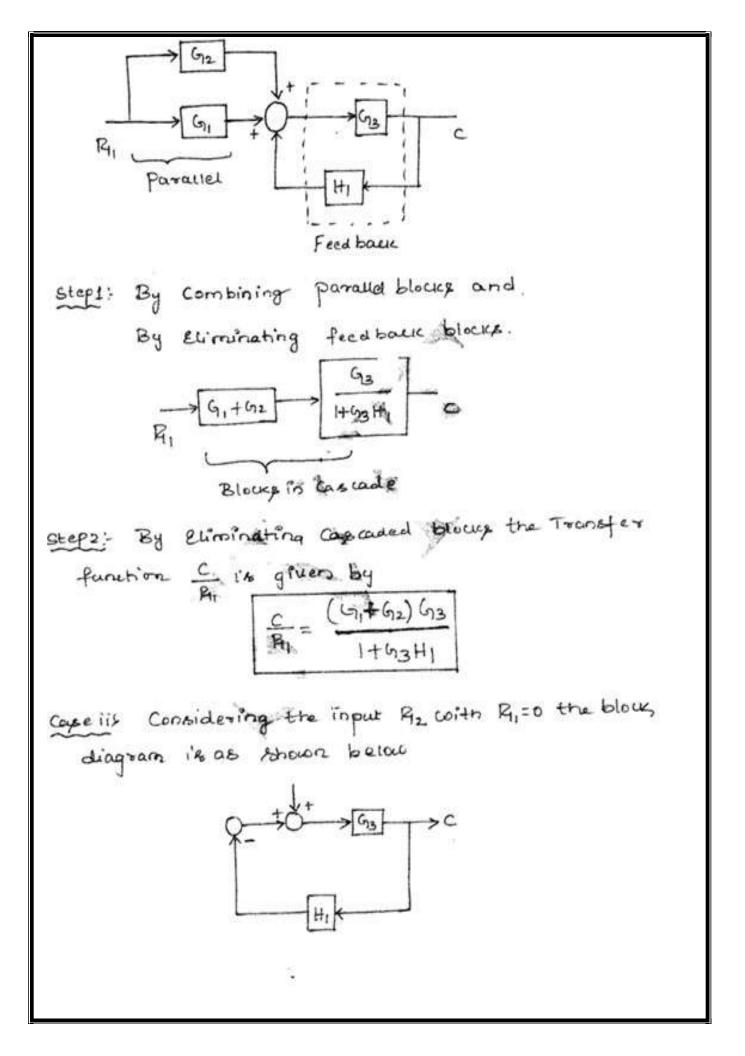


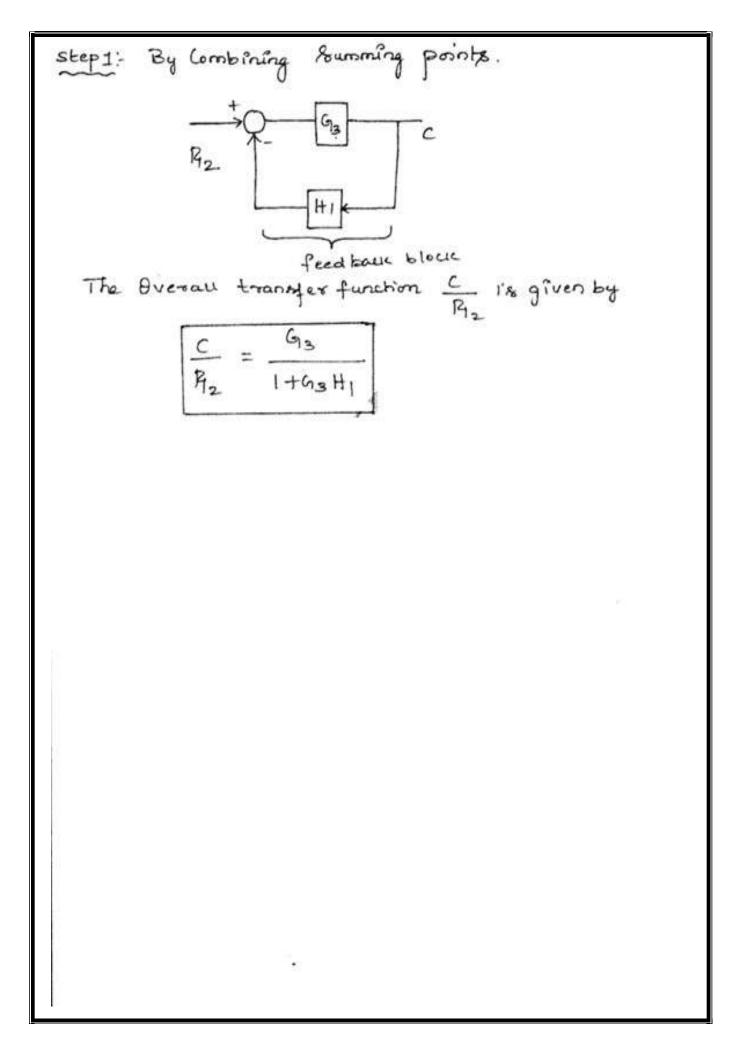


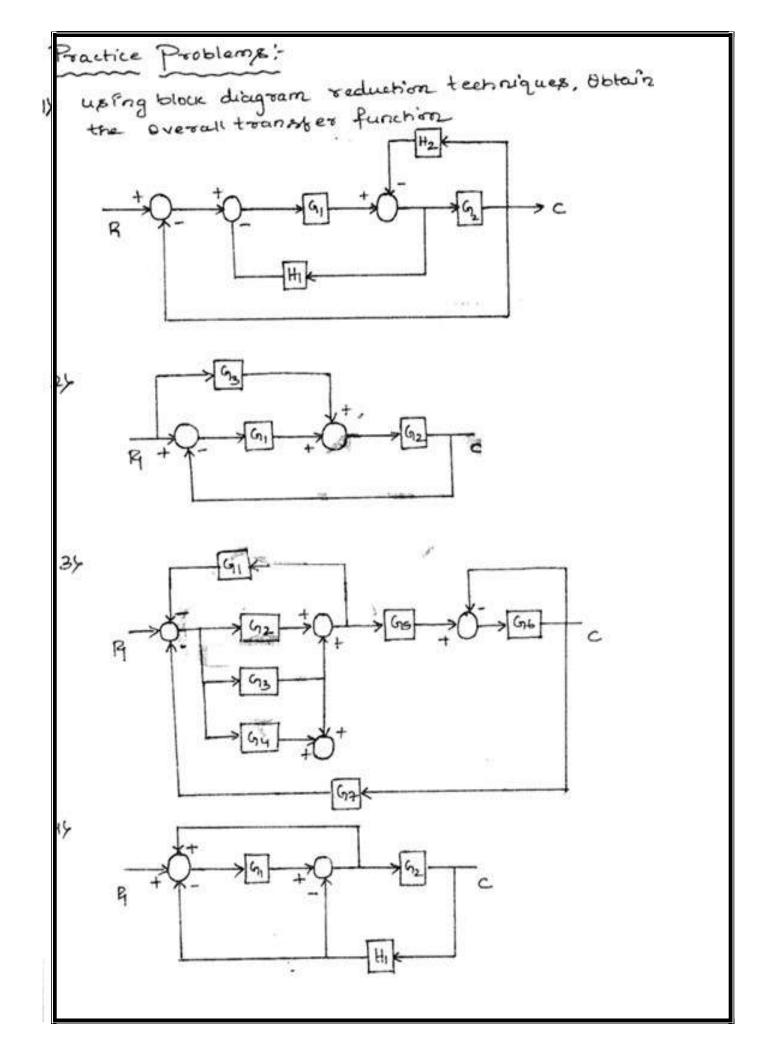


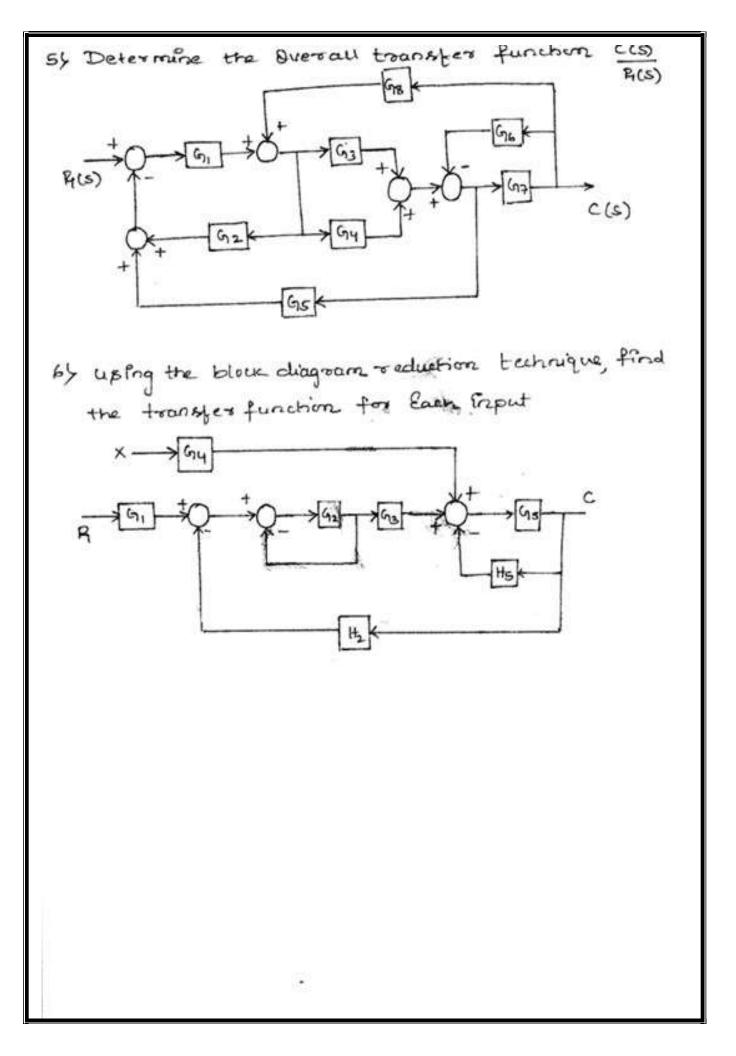


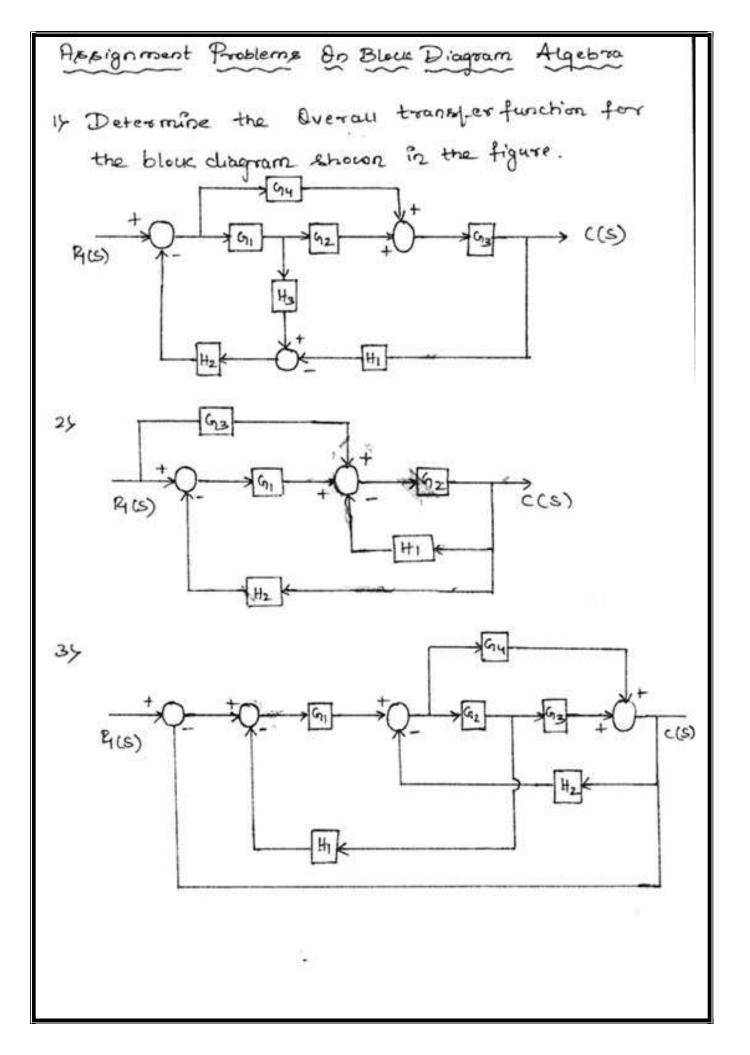
$$\frac{1}{\frac{1}{R_{1}}\left(\frac{R_{1}}{R_{1}}+R_{2}\right)\left(\frac{R_{1}}{R_{2}}+R_{2}\right)\left(\frac$$

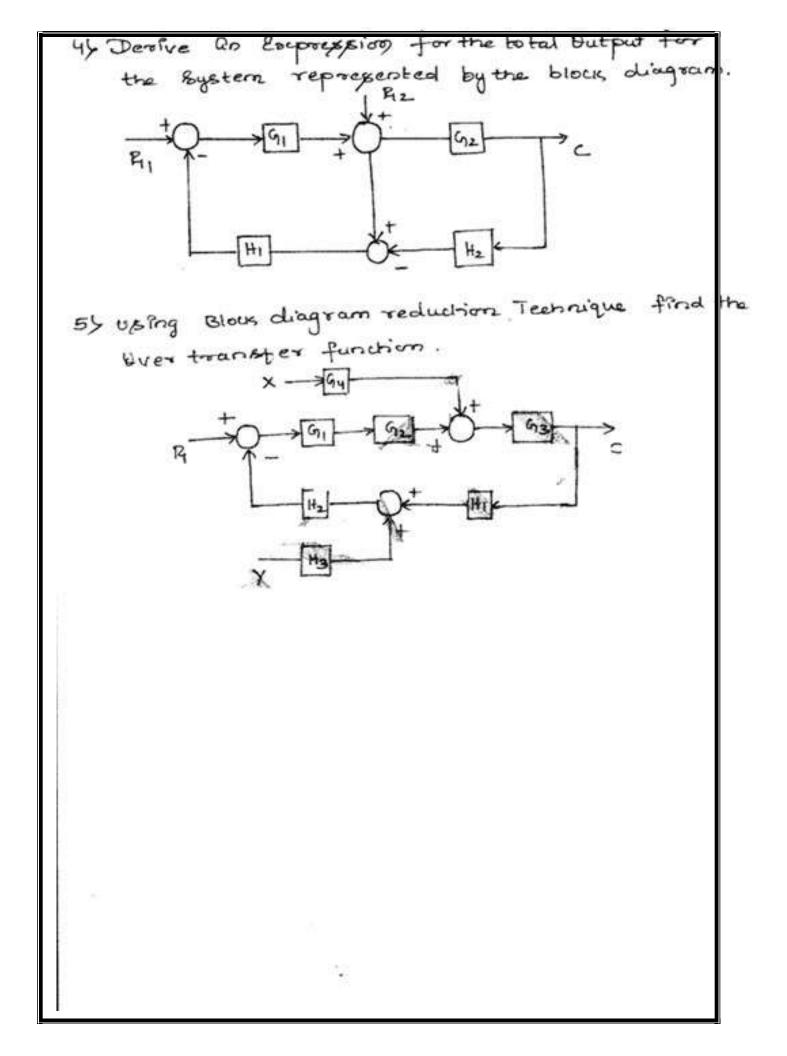








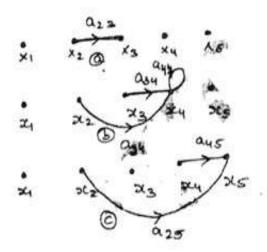




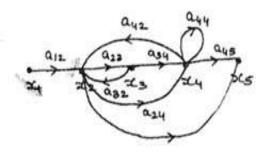
Signal Flow Graph It is defined as the graphical representation of a set of simultaneous Equations or transfer function. The Overall transfer functions for a Reignal flow graph can be determined by using Moson's gain foreroula. it is given by. M(S) = Z PKOK where . A=1- 2 Pmi+ Epmz- Z:Pmz+ PKibthe forward path gain of the Kth forward path. DB is the Co.factor. Forward path : It is the Towney from input to Sutput node to the direction of arrows without touching in between more than once. N -> 00 to for woord path present in brighal thou graph. Loop: A path in raid to be a loop if we start from a node and come back to the same node in the direction of arrows without touching any nodes in between more than once. Two-non-touching loops: Two loops are said to be non-touching, it they don't have a Common node of Variable between them. is the determinant of the soignal flow graph. Pm, is the sum of loop gains of all possible combinations

Zfraz is kum of product of loop gain of all possible
combinations of 2 non - touching loops.
Zfraz is the sum of product of the loop gain of
all combinations of 3 non - touching loops.
An is the co-battor of the graph. The Ecopression for An
is similar to
$$\Delta$$
, but it must be applied to that part
of the graph not touching the st forceard path.
If a forward path touches all single loop possent
in the graph then the corresponding $\Delta K = 1$
 \Rightarrow Procedure to Construct Signal flows graph from linear
Equations:
 $\chi_2 = a_{12}\chi_1 + a_{32}\chi_3 + a_{42}\chi_4$
 $\chi_3 = a_{23}\chi_2$.
 $\chi_4 = a_{24}\chi_2 + a_{34}\chi_3 + a_{44}\chi_4$
 $\chi_5 = a_{25}\chi_2 + a_{45}\chi_4$
* Where χ_1 is the input Variable and χ_5 is the Sutput
Variable.
* Under constructing flowsgraph, the nodes are used
to represent Variables.
* Therefore locate the nodes $\chi_1, \chi_2, \chi_3, \chi_4, \chi_5$ as
shown in the figure below.

* The first Equation states that 262 is Equal to the sum of three incoming soignals and it's signal flow graph is shown in the figure.
* Similarly the soignal flow graph for the remaining three Equations are shown in the figure.



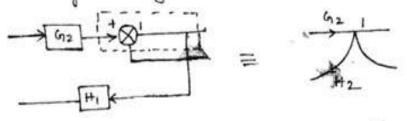
* The Complete signal flow graph is a combination. of all the four parts on shown in the figure.



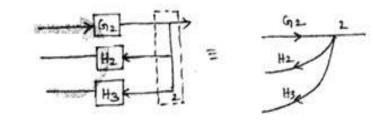
=> Procedure to draw Brignal flow graph from Block diagram.

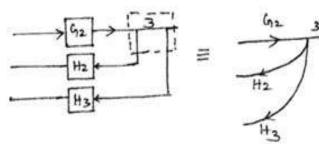
Step1: Replace the input inignal and Butput inignal by nodes. Step2: Replace all the summing points by nodes. Step3: Replace all the take off points by nodes. Step4: If the branch Connecting a summing point. and take off point has unity gain, then the Summing Point and take off point can be combined and.

represented by a fingle node.



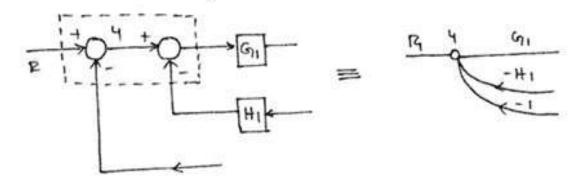
Same Arignal then all the take off points from the . Combined and represented by a single node.



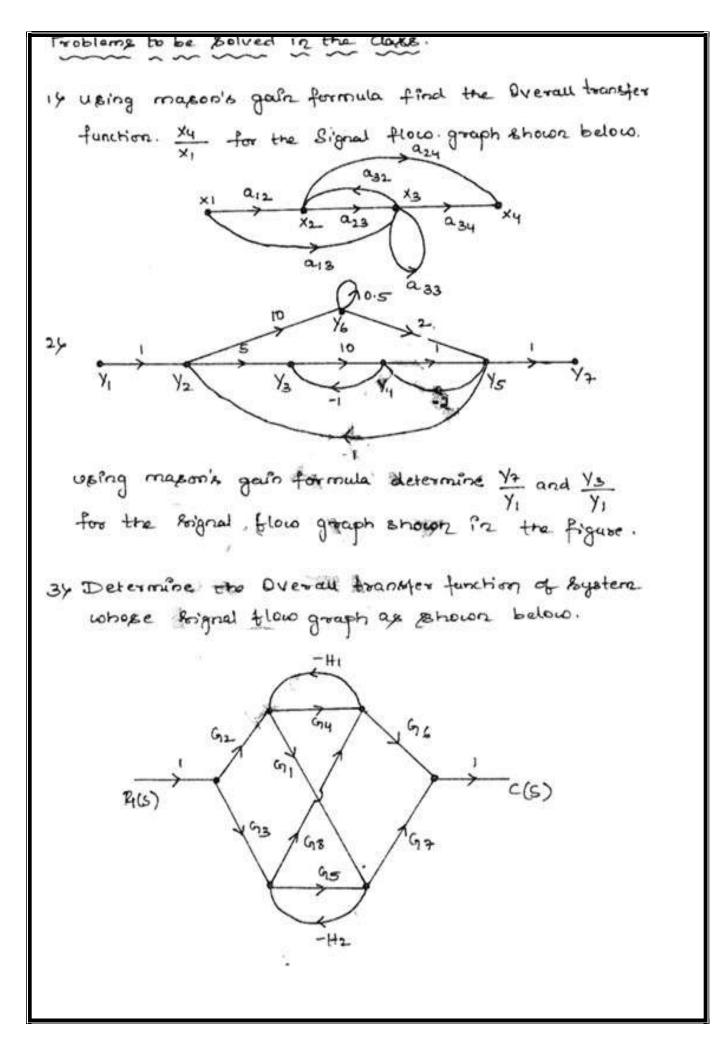


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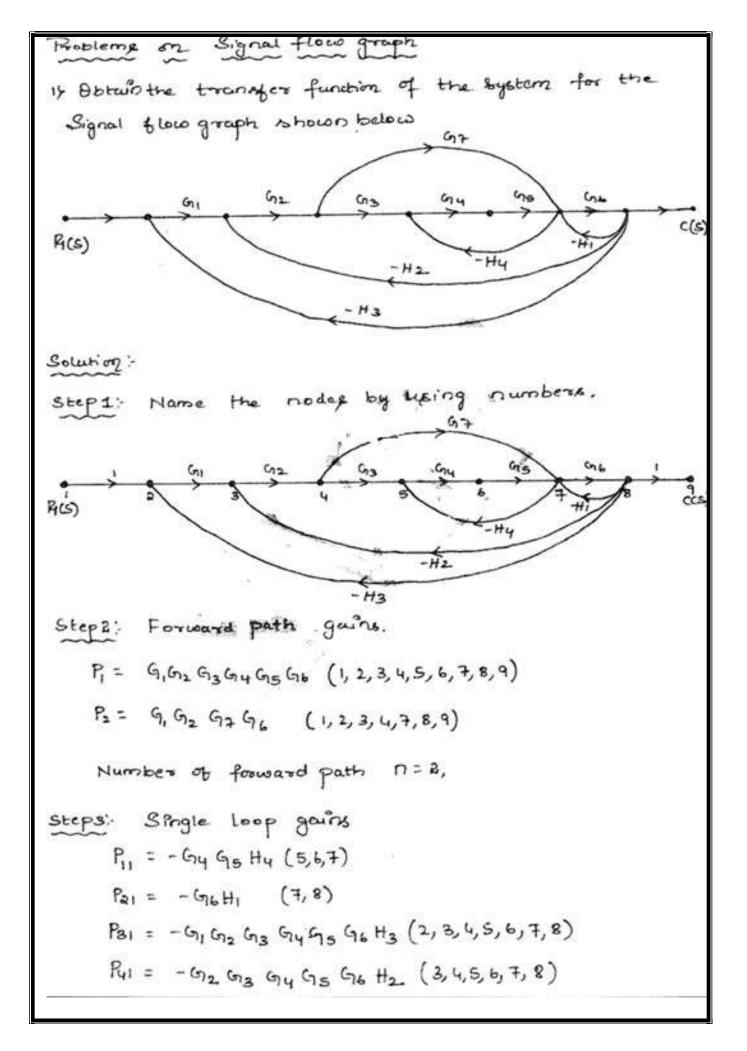
Steps: If the gala of the link Connecting two Summing Points is one, there the toop Sournming points (an be Combined and (an be replaced by a Single node as Schown in the figure.



Step 7: In sourning point, subbract a signal instead. of adding and then multiply the transmittance by -1 while representing for signal flow graph ax shown in the figure. above.



4) Drows the signal flow groups for the system of Equations given below and Obtain the transfer function uping reappoor's gains forrowly. Xo= G, X1 - H1 X2 - H2 X3 - H1 X1 $X_3 = G_1 X_1 + G_2 X_2 - H_3 X_3$ ×4 = G2×2+G3×3-H4×5 X5 = G5X4 - H5X6 X1 = GSX5 5) Dears the Signal flow graph for the block dragoan shown in the figure. Determine the overall transfer function using maxon is gaine formula. H2_ G14 63 C(S) R(s) 45 H by For the Electrical Circuit Shown in the figure, find the overall transfer function using Maxon's gein formula. mr 100KR + * R. R12 GT 10HF C2- INF Volt) Vict)



$$F_{31} = -G_{2}G_{3}+G_{6}H_{2} (3,4,7,8)$$

$$F_{61} = -G_{1}G_{2}G_{3}+G_{6}H_{3} (2,3,4,7,8)$$

$$Step 4: Two non-touching loop gains.
+ Two non touching loops and higher Order is absent
$$\Sigma Pm_{2} + 0 novards is 3 erro.$$

$$Step 5: To find \Delta K.$$
Note:::Number of forwards path is Baisel to Cumber of .
(co-tautor (\Delta K))

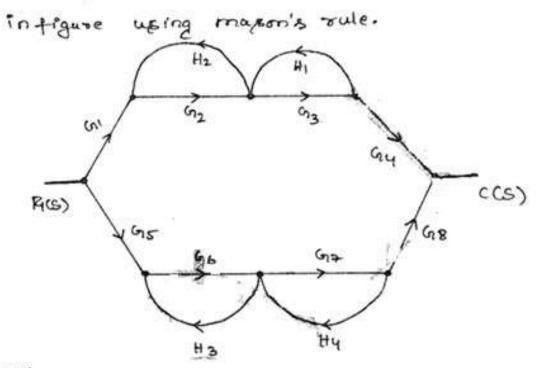
$$T By (omparing forward path is Baisel to Cumber of .
(co-tautor (\Delta K))
$$T By (omparing forward path gains with Single.
Usep gains.
$$fme_{K}: \Delta_{1} = 1 + Pr_{1} + P_{3} - P_{3}$$

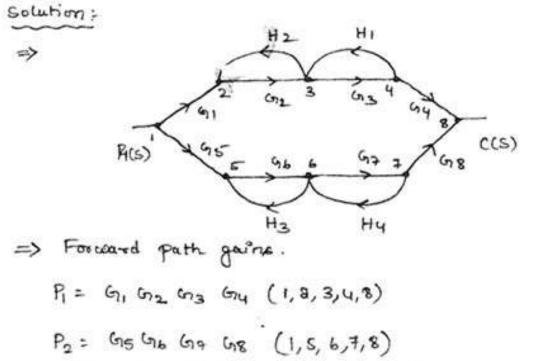
$$Step 6: Find the Overall transfer function using
mason's gain formula.
$$\frac{c(s)}{P_{1}(s)} = \frac{\sum_{k=1}^{n-2} P_{k} \Delta_{k}}{1 - \sum_{m=1}^{n} P_{m} + 0}$$

$$\frac{c(s)}{P_{1}(s)} = \frac{P_{1}\Delta_{1} + P_{2}\Delta_{2}}{1 - (P_{11} + P_{3} + P_{3} + P_{3} + P_{3})}$$$$$$$$$$

$$\frac{(c (s))}{P_{1}(s)} = \frac{(\Theta_{1} \Theta_{2} \Theta_{3} \Theta_{4} \Theta_{5} \Theta_{6})(1) + (\Theta_{1} \Theta_{2} \Theta_{7} \Theta_{6})(1)}{1 - ((-\Theta_{4} \Theta_{5} H_{4}) + (-\Theta_{4} H_{1}) + (-\Theta_{1} \Theta_{2} \Theta_{3} \Theta_{4} \Theta_{5} \Theta_{6} H_{3})} + ((-\Theta_{2} \Theta_{3} \Theta_{4} \Theta_{5} \Theta_{6} H_{2}) + ((-\Theta_{2} \Theta_{7} \Theta_{6} H_{2}) + (-\Theta_{2} \Theta_{7} \Theta_{6} H_{3}))$$

27 Obtain the Overall transfer for the system shown





Number of forward Dath 0=2.

$$\Rightarrow Stringle loop gentry':
P_{11} = G_{2} H_{2} (2, 3)
P_{21} = G_{13} H_{1} (3, 4)
P_{31} = G_{16} H_{3} (5, 6)
P_{41} = G_{1} H_{4} (6, 7)
$$\Rightarrow Two \quad Orn - touching loop gentry.
P_{12} = P_{11} P_{31} = G_{2} H_{2} G_{16} H_{3} (2, 3, 5, 3)
P_{32} = P_{31} P_{31} = G_{3} H_{2} G_{16} H_{4} (2, 3, 3, 5, 3)
P_{32} = P_{31} P_{31} = G_{3} H_{1} G_{6} H_{4} (3, 4, 5, 6)
P_{42} = B_{31} P_{31} = G_{3} H_{1} G_{6} H_{4} (3, 4, 5, 6)
P_{42} = B_{31} P_{41} = G_{3} H_{1} G_{6} H_{4} (3, 4, 5, 6)
P_{42} = B_{31} P_{41} = G_{3} H_{1} G_{6} H_{4} (3, 4, 5, 6)
P_{42} = B_{1} R_{41} = G_{13} H_{1} G_{9} H_{4} (3, 4, 5, 6)
P_{42} = B_{1} R_{41} = G_{13} H_{1} G_{9} H_{4} (3, 4, 5, 6)
TPm_{3} + Onwardsk (4 Sero.
$$\Rightarrow Co-foctor dt graph
\Delta_{1} = 1 - (P_{81} + P_{41}) + 0 = 1 - (G_{16} H_{3} + G_{15} H_{4})
\Delta_{2} = 1 - (P_{11} + P_{21}) + 0 = 1 - (G_{12} H_{2} + G_{13} H_{1})
\Rightarrow Overall transperi function
M(S) = \sum_{k=1}^{N} \frac{P_{k} \Delta_{1k}}{\Delta}$$

$$where \Delta is 1 - \SigmaPm_{1} + \SigmaPm_{2} - \Sigma Pm_{3} + - + \cdots$$

$$M(S) = \frac{\sum_{k=1}^{N} P_{k} \Delta_{k}}{P_{m_{2}1} + \frac{W}{m_{2}1}} Pm_{2} - 0} = \frac{P_{1}\Delta_{1} + P_{2}\Delta_{2}}{1 - (P_{11} + P_{21} + P_{31} + P_{42})}$$$$$$

$$\frac{((s)}{P_{1}(s)} = \frac{(\omega_{1}, \omega_{2}, \omega_{3}, \omega_{4})(1 - \omega_{6}H_{3} - \omega_{7}H_{4}) + (\omega_{5}, \omega_{6}, \omega_{7}, \omega_{8})(1 - \omega_{2}H_{2} + \omega_{3}H_{1})}{1 - [\omega_{2}H_{2} + \omega_{3}H_{1} + \omega_{6}H_{3} + \omega_{7}H_{4}] + [\omega_{2}H_{2}\omega_{6}H_{3} + \omega_{2}H_{2}\omega_{7}H_{4}]} + (\omega_{3}H_{1}\omega_{6}H_{3} + \omega_{3}H_{1}\omega_{7}H_{4}]$$

34 Draw the Signal flow graph for the System of Equations given below and Obtain the Overall transfer function using maxon's rule.

$$X_{2} = X_{1} - H_{3} \times 8$$

$$X_{3} = G_{1} \times 2 - H_{2} \times 8$$

$$X_{4} = G_{2} \times 3$$

$$X_{5} = G_{3} \times 4 - H_{4} \times 6$$

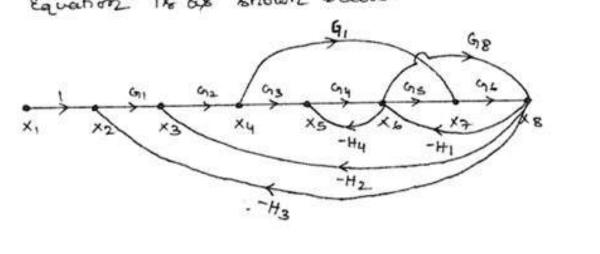
$$X_{6} = G_{4} \times 6 - H_{1} \times 8$$

$$X_{7} = G_{2} \times 4 + G_{5} \times 6$$

$$X_{8} = G_{8} \times 6 + G_{5} \times 7$$

Solution ?

The Signal flow graph satisfying the above Equation 1's as shown below.

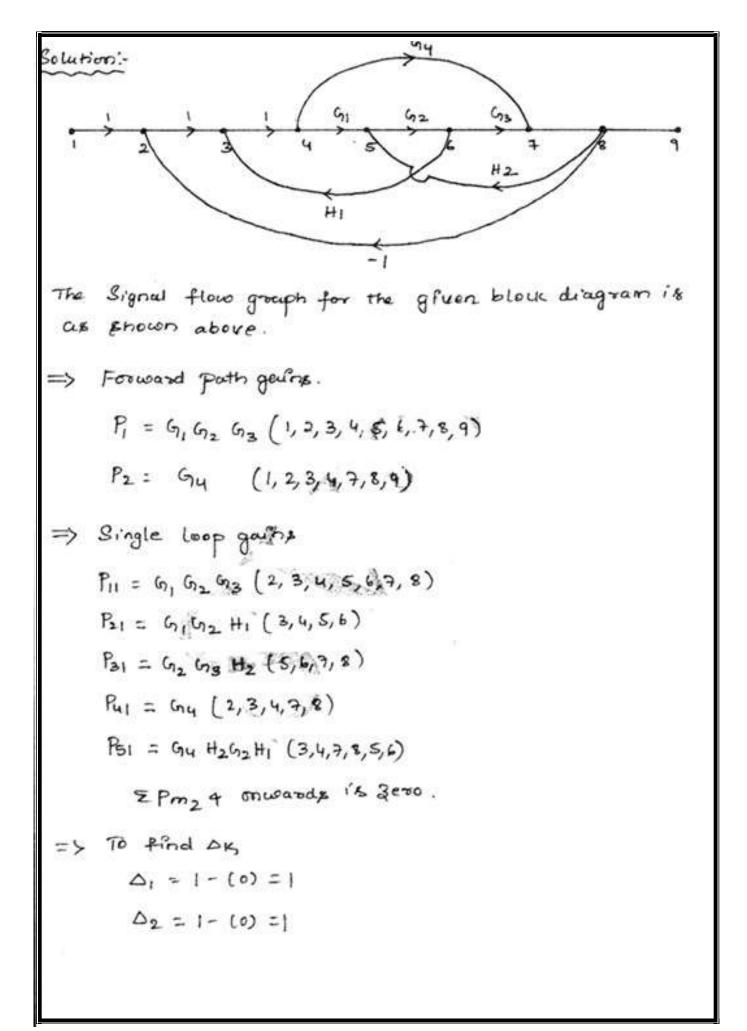


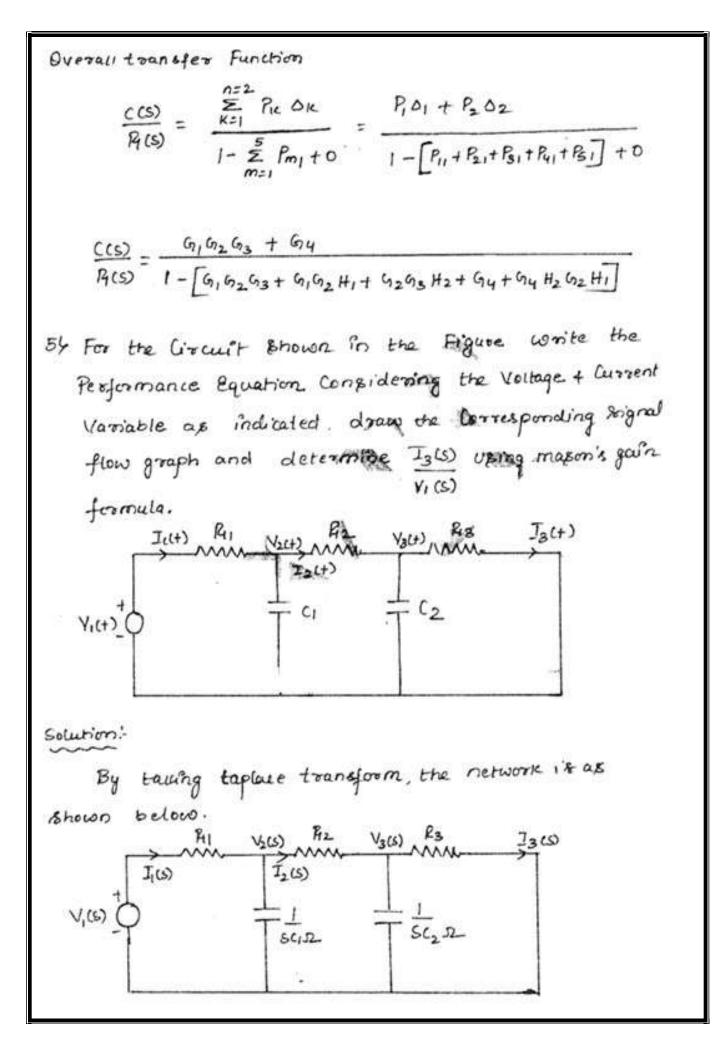
=> Forward parts garys.

$$P_1 = G_1 G_2 G_3 G_1 G_5 G_6 (1, 2, 3, 4, 5, 6, 7, 8)$$

 $P_2 = G_1 G_2 G_3 G_4 G_5 (1, 2, 3, 4, 5, 6, 7, 8)$
 $P_3 = G_1 G_2 G_3 G_4 G_8 (1, 2, 3, 4, 5, 6, 7, 8)$
Numbers of forward parts $0 = 3$.
=> Single (oop garss.
 $P_{11} = -G_1 G_2 G_3 G_4 G_5 G_6 H_3 (2, 3, 4, 5, 6, 7, 8)$
 $P_{21} = -G_1 G_2 G_3 G_4 G_5 H_3 (2, 3, 4, 5, 6, 8)$
 $P_{31} = -G_1 G_2 G_3 G_4 G_5 H_3 (2, 3, 4, 5, 6, 8)$
 $P_{41} = -G_2 G_3 G_4 G_5 H_3 (2, 3, 4, 5, 6, 8)$
 $P_{41} = -G_2 G_3 G_4 G_5 H_2 (3, 4, 5, 6, 7, 8)$
 $P_{51} = -G_2 G_5 G_4 G_5 H_2 (3, 4, 5, 6, 7, 8)$
 $P_{51} = -G_2 G_5 G_4 G_5 H_2 (3, 4, 5, 6, 8)$
 $P_{51} = -G_2 G_5 G_4 G_5 H_2 (3, 4, 5, 6, 8)$
 $P_{51} = -G_4 H_4 (5, 6)$
 $P_{51} = -G_8 H_1 (6, 7, 8)$
 $P_{51} = -G_8 H_1 (6, 7, 8)$
 $P_{51} = -G_8 H_1 (6, 7, 8)$
 $P_{52} = P_{51} P_{51} = G_1 G_2 G_5 G_6 H_3 G_4 H_4 (2, 3, 4, 5, 6, 7, 8)$
 $P_{52} = P_{51} P_{51} = G_2 H_2 G_5 H_2 G_4 H_4 (3, 4, 5, 6, 7, 8)$
 ΣP_{53} and $\Theta n wards (6 Ser0.$
 $=> (o-fauro of Graph.$
 $A_1 = 1-0 = 1$
 $A_2 = 1 - P_{51} + 0 = 1^2 + G_4 H_4$
 $A_3 = 1$

)





$$\begin{split} I_{1}(s) &= \frac{V_{1}(s) - V_{2}(s)}{R_{1}} \\ I_{1}(s) &= \frac{1}{R_{1}} V_{1}(s) - \frac{1}{R_{1}} V_{2}(s) \longrightarrow 0 \\ \\ T_{2}(s) &= \frac{V_{2}(s) - V_{3}(s)}{R_{2}} \\ &= \frac{1}{R_{12}} V_{2}(s) - \frac{1}{R_{12}} V_{3}(s) \longrightarrow 0 \\ \\ T_{3}(s) &= \frac{V_{3}(s)}{R_{3}} \longrightarrow 0 \\ \\ V_{2}(s) &= \frac{1}{sc_{1}} (I_{1}(s) - \tilde{I}_{2}(s)) \\ V_{2}(s) &= \frac{1}{sc_{1}} (I_{1}(s) - \tilde{I}_{2}(s)) \\ V_{2}(s) &= \frac{1}{sc_{1}} (I_{1}(s) - \tilde{I}_{3}(s)) \\ V_{3}(s) &= \frac{1}{sc_{2}} (\tilde{I}_{2}(s) - I_{3}(s)) \\ V_{3}(s) &= \frac{1}{sc_{2}} (I_{2}(s) - \frac{1}{sc_{2}} I_{3}(s) \longrightarrow 0 \\ \\ &= Signal \ \text{flew} \ \text{graph} \ \text{is drawn is kap & brown below} \\ \\ &= \frac{-V_{R_{1}}}{V_{1}(s)} \frac{-V_{R_{2}}}{I_{2}(s)} \frac{V_{2}(s)}{V_{R_{2}}} \frac{V_{2}(s)}{I_{3}(s)} \frac{V_{1}(s)}{V_{3}(s)} \frac{V_{1}(s)}{V_{1}} \frac{V_{2}(s)}{V_{3}(s)} = \frac{1}{I_{4}(s)} V_{2}(s) \frac{V_{2}(s)}{V_{1}} \frac{V_{2}(s)}{V_{2}(s)} \frac{V_{2}(s)}{V_{R_{2}}} \frac{V_{2}(s)}{I_{3}(s)} \frac{V_{2}(s)}{V_{3}(s)} \frac{V_{2}(s)}{V_{1}} \frac{V_{2}(s)}{V_{2}(s)} \frac{V_{2}(s)}{V_{1}} \frac{V_{2}(s)}{V_{3}(s)} \frac{V_{3}(s)}{V_{3}(s)} \frac{V_{3}(s)}{V_{3}($$

=> Forward path gains

$$P_{1} = \frac{1}{R_{1}} \cdot \frac{1}{sc_{1}} \cdot \frac{1}{R_{2}} \cdot \frac{1}{sc_{2}} \cdot \frac{1}{R_{3}} = \frac{1}{s^{2}c_{1}c_{2}R_{1}R_{2}R_{3}}$$
Number of forward path gain, $n = 1$
=> Single loop gains

$$P_{11} = -\frac{1}{sc_{1}R_{1}}$$

$$P_{21} = -\frac{1}{sc_{2}R_{2}}$$

$$P_{31} = -\frac{1}{sc_{2}R_{2}}$$

$$R_{41} = -\frac{1}{sc_{2}R_{3}}$$
=> Two - non-towning Loop gains

$$P_{12} = P_{11}R_{31} = \frac{1}{s^{2}c_{1}c_{2}R_{1}R_{2}}$$

$$P_{22} = P_{11}R_{41} = \frac{1}{s^{2}c_{1}c_{2}R_{1}R_{3}}$$

$$P_{32} = R_{2}P_{41}P_{41} = \frac{1}{s^{2}c_{1}c_{2}R_{2}R_{3}}$$

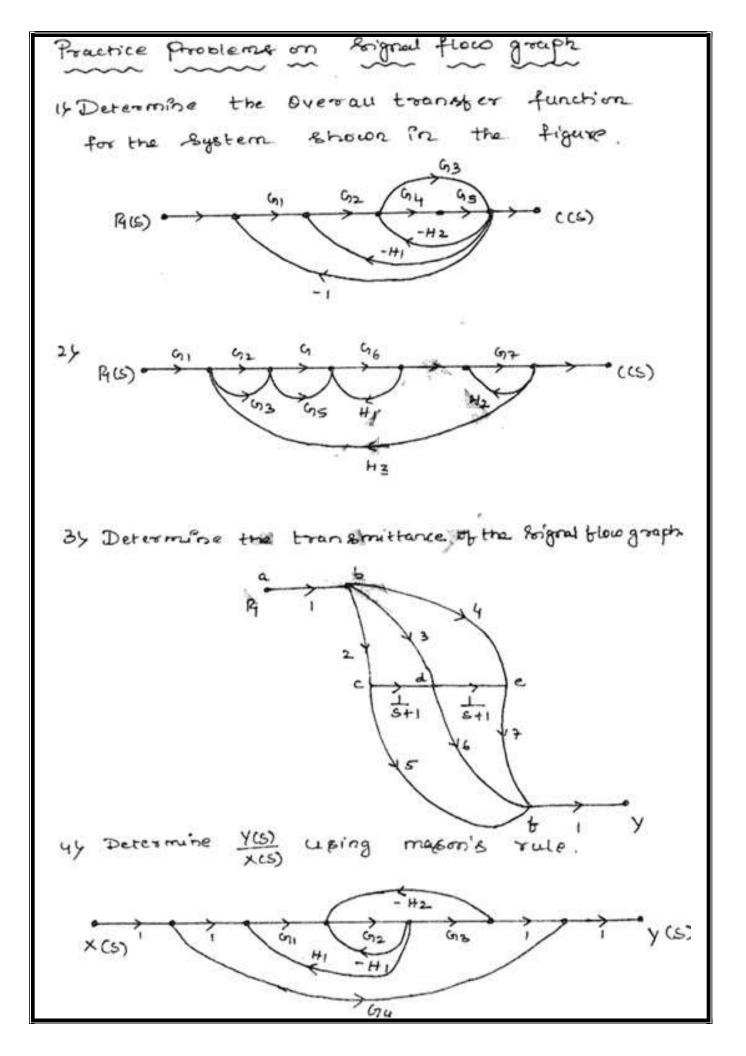
$$\sum P_{33} = R_{2}P_{41}P_{41} = \frac{1}{s^{2}c_{1}c_{2}R_{2}R_{3}}$$

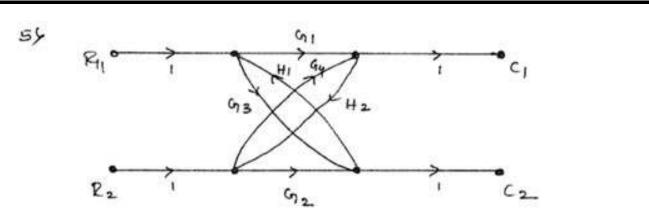
$$\sum P_{12} = P_{11}R_{41} = \frac{1}{s^{2}c_{1}c_{2}R_{2}R_{3}}$$

=> The Overall transfer function is given by

$$T.F = \frac{I_{3}(s)}{V_{1}(s)} = \frac{\sum_{k=1}^{n=1} P_{1k} \Delta_{k}}{1 - \sum_{m=1}^{4} P_{m1} + \sum_{m=1}^{3} P_{m2} - 0.}$$

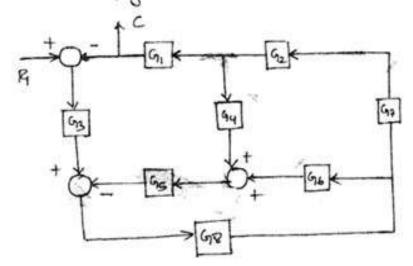
$$\frac{I_{3}(s)}{V_{1}(s)} = \frac{P_{1} \Delta_{1}}{1 - (P_{11} + P_{21} + P_{31} + P_{41}) + (P_{12} + P_{22} + P_{32})}$$



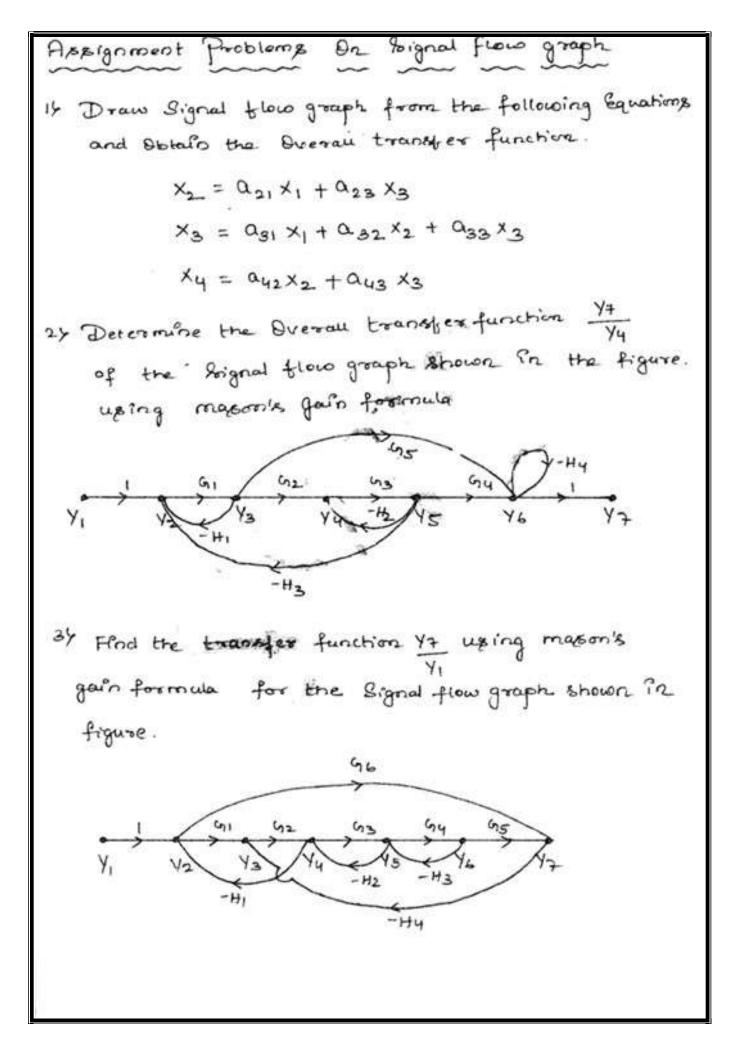


Mason's rule.

65 Draw a signal flow graph for the block diagram. shows in the figure.



۰.



44 Draw the Signal flow graph for the block diagram shown in the figure and Evaluate ((s) uping mason's formula. 62 t (2)) R(S) #) 614 54 Find the Overall transfer function ((5) using R(5) mapon's rule for the Aystern shoop in the figure. 446 A23 ASI AUS A34 C(S) R(S) A43 ASY A82 A21 A65

$$\begin{split} & \frac{Problem 3:}{Problem 3:} \\ & + \rightarrow F-V \text{ Analogy}: \\ & Equation 1: V(+) = L_1 \frac{d}{dt} i_1 + \frac{1}{c_1} \int (i_1 - i_2) dt + P_{I_1}(i_1 - i_2); \\ & Equation 2: P_{I_1}(i_1 - i_2) = L_2 \frac{d}{dt^2} + \frac{1}{c_2} \int (i_2 - i_3) dt; \\ & Equation 3: \frac{1}{c_1} \int (i_1 - i_1) dt = P_{I_2}(i_1 - i_3); \\ & Equation 4: P_{I_2}(i_1 - i_3) + \frac{1}{c_2} \int (i_2 - i_3) dt = L_3 \frac{d}{dt} i_3 + P_{I_3} i_3 + \frac{1}{c_3} \int i_3 dt; \\ & + \rightarrow F-I \text{ Analogy}: \\ & Equation 2: G_1(V_1 - V_2) = C_2 \frac{d}{dt} V_2 + \frac{1}{L_2} \int (V_2 - V_3) dt; \\ & Equation 3: \frac{1}{L_1} \int (V_1 - V_2) = C_2 \frac{d}{dt} V_2 + \frac{1}{L_2} \int (V_2 - V_3) dt; \\ & Equation 3: \frac{1}{L_1} \int (V_1 - V_2) + \frac{1}{L_2} \int (V_2 - V_3) dt = C_3 \frac{d}{dt} V_3 + \frac{1}{L_3} \int V_3 dt; \\ & \frac{1}{V_3} \int (i_1 - i_2) dt = L_1 \frac{d}{dt} + P_{I_1} i_1 + \frac{1}{C_3} \int (i_2 - i_3) dt + P_{I_2} (i_2 - i_3); \\ & Equation 2: \int (i_1 - i_2) dt = L_1 \frac{d}{dt} + \frac{1}{C_3} \int (i_2 - i_3) dt + P_{I_2} (i_2 - i_3); \\ & Equation 2: \int (i_1 - i_2) dt = L_1 \frac{d}{dt} + \frac{1}{C_3} \int (i_2 - i_3) dt + P_{I_2} (i_2 - i_3); \\ & Equation 2: \int \frac{1}{C_3} \int (i_2 - i_3) dt + P_{I_2} (i_2 - i_3) = L_2 \frac{d}{dt} i_3; \\ & \frac{1}{V_3} dt; \\ & \frac{1}{V_3} dt; \\ & \frac{1}{V_3} dt = \frac{1}{C_3} \int (i_2 - i_3) dt + P_{I_2} (i_2 - i_3) = L_2 \frac{d}{dt} i_3; \\ & \frac{1}{V_3} dt; \\ & \frac{1}{V_3} dt; \\ & \frac{1}{V_3} dt; \\ & \frac{1}{V_3} dt; \\ & \frac{1}{V_3} dt + \frac{1}{V_3} \int \frac{1}{V_3} dt + \frac{1}{V_3} \int \frac{1}{V_3} dt + \frac{1}{V_3} \int \frac{1}{V_3} dt \\ & \frac{1}{V_3} dt; \\ & \frac{1}{V_3} dt; \\ & \frac{1}{V_3} dt; \\ & \frac{1}{V_3} dt + \frac{1}{V_3} \int \frac{1}{V_3} dt + \frac{1}{V_3} \int \frac{1}{V_3} dt \\ & \frac{1}{V$$

Equation 1:
$$I(t) = C_1 \frac{d_1V_1}{dt} + G_1V_1 + \frac{1}{L_1}\int_{V_1} dt + \frac{1}{L_2}\int_{V_1} (V_1 - V_2)dt$$
:
Equation 2: $\frac{1}{L_3}\int_{V_2} (V_2 - V_3)dt = C_1 \frac{d_1V_2}{dt} + \frac{1}{L_3}\int_{V_2} (V_2 - V_3)dt + G_2(V_2 - V_3) = C_2 \frac{d_1V_3}{dt}$;
Equation 3: $\frac{1}{L_3}\int_{V_2} (V_2 - V_3)dt + G_2(V_2 - V_3) = C_2 \frac{d_1V_3}{dt}$;
Problem 5:
* $\rightarrow F - V$ Analog Y:
Equation 2: $\frac{1}{C_1}\int_{V_1} (i_2 - i_1)dt = L_1 \frac{d_1}{dt} + \frac{1}{C_1}\int_{V_2} (i_2 - i_1)dt + R_2(i_2 - i_3)$
Equation 3: R_2 $(i_2 - i_3) = L_3 \frac{d_1V_3}{dt} + \frac{1}{L_2}\int_{V_2} i_3dt$;
* $\rightarrow F - I$ Analogy:
Equation 1: $I(t) = C_2 \frac{d_1V_2}{dt} + \frac{1}{L_3}\int_{V_2} V_2 dt + \frac{1}{L_1}\int_{V_2} (V_2 - V_1)dt + G_2(V_2 - V_3)$;
Equation 2: $\frac{1}{L_1}\int_{V_2} (V_2 - V_3) = C_3 \frac{d_1V_3}{dt} + \frac{1}{L_2}\int_{V_3} dt$;
Froblem 6:
* $\rightarrow F - V$ Analogy:
Equation 1: $V(t) = L_2 \frac{d_1V_2}{dt} + R_1 i_2 + \frac{1}{L_2}\int_{V_3} dt$;
Equation 2: $\frac{1}{L_1}\int_{V_2} (V_2 - V_3) = C_3 \frac{d_1V_3}{dt} + \frac{1}{L_2}\int_{V_3} V_3 dt$;
Equation 1: $V(t) = L_2 \frac{d_1V_2}{dt} + R_1 i_2 + \frac{1}{L_2}\int_{V_3} i_2 t + \frac{1}{R_1}(i_2 - i_1) dt$
Equation 2: $\frac{1}{L_1}\int_{V_2} (V_2 - V_3) = C_3 \frac{d_1V_3}{dt} + \frac{1}{L_2}\int_{V_3} i_2 t + \frac{1}{R_1}(i_2 - i_1)$;
Equation 2: $\frac{1}{L_1}\int_{V_2} ((i_2 - i_1)dt + R_1(i_2 - i_1)) = L_1 \frac{d_1}{dt}$;

Problem 6: $\frac{\Theta_1(s)}{T(s)} = \frac{(J_2 s^2 + K_2) K_1}{J_1 J_2 J_m s^6 + (K_2 J_m J_1 + J_2 J_1 K_2 + J_2 J_1 K_1 + J_2 J_m K_1) s^4}$ $+(J_2 H_2 H_1 + J_1 H_1 H_2 + J_m H_1 H_2)S^2$ Problem 7: $\frac{x(s)}{=} = \frac{B_{12}s + K_{11}}{B_{12}s + K_{11}}$ $S(M,M,S^3 + (M_2,B_{12} + M_1B + M_1B_{12})S^2 + (M_1H_1 + M_1B_{12})S^2$ F(s) M214, + BB12)S+4,B) BLOCK Diagram Algebra Problem 1: $\frac{C(S)}{R(S)} = \frac{G_1 G_2}{1 + G_2 H_2} + G_1 G_2 + G_1 H_1$ Problem 2: $\frac{C(S)}{B(S)} = \frac{G_2(G_1 + G_3)}{1 + G_1 G_2}$ BIS) Problem 3: $\frac{C(S)}{S} = \frac{G_3G_6 + G_4G_6 + G_2G_5G_6 + G_3G_5G_6}{G_3G_5G_6}$ R(S) 1+9, 62+6, 63+66+6, 6266+6, 6366+63664 G4 G6 G7 + G2 G5 G6 G7 + G3 G5 G6 Roblem 4 :- $= \frac{G_1 G_2}{1 + G_1 G_2 H_1 + G_2 H_1 - G_1}$ C(S) R(S)

$$\frac{Froblem 5:}{R(s)} = \frac{G_{1} G_{2} (G_{3} + G_{4})}{1 + G_{1} G_{1} G_{2} - G_{3} G_{3} G_{3} + G_{1} G_{3} G_{3} + G_{1} G_{3} G_{3} + G_{1} G_{3} G_{3} + G_{1} G_{2} + G_{2} + G_{2} + G_{1} G_{2} + G_{$$

$$\frac{P_{\text{roblem 5}}}{M(s)} = \frac{C_{2}(s)}{R_{1}(s)} = \frac{C_{3}(1 - G_{4}H_{2}) + G_{1}G_{2}H_{2}}{(1 - G_{4}H_{2}) + H_{1}(G_{3}(G_{4}H_{2} - 1) - G_{1}G_{2}H_{2})}$$

$$\frac{P_{\text{roblem 6}}}{P_{1}(s)} = \frac{C(s)}{R(s)} = \frac{(1 + G_{5}G_{6}G_{8} + G_{2}G_{4}G_{5}G_{7}G_{8})}{1 + G_{5}G_{6}G_{8} + G_{2}G_{4}G_{5}G_{7}G_{8} + G_{1}G_{2}G_{3}G_{7}G_{8}}$$

Module no 2 Questions

Q2-1: Define transfer function and what are its properties. Jun. 2013, 5 Marks

Q2-2: Illustrate how to perform the following, in connection with block diagram reduction rules:

(i) Shifting a take-off point after a summing point.

(ii) Shifting a take-off point before a summing point.

Q2-3: The performance equations of a controlled system are given by the following set of linear algebraic equations: Dec. 2012, 8 Marks

> $E_1(s) = R(s) - H_3(s)C(s);$ $E_2(s) = E_1(s) - H_1(s)E_4(s);$ $E_3(s) = G_1(s)E_2(s) - H_2(s)C(s);$ $E_4(s) = G_2(s)E_3(s);$ $C(s) = G_3(s)E_4(s);$

(i) Draw the block diagram.

(ii) Find the overall transfer function $\frac{C(s)}{R(s)}$ using block diagram reduction technique

Q2-4: For the system represented by the following equations, find the transfer function

X(s)by signal flow graph, technique U(s)

 $\mathbf{x} = \mathbf{x}_1 + \boldsymbol{\alpha}_3 \mathbf{u}$ $\dot{\mathbf{x}}_1 = -\beta_1 \mathbf{x}_1 + \mathbf{x}_2 + \alpha_2 \mathbf{u} \, \dot{\mathbf{x}}_2 = -\beta_2 \mathbf{x}_1 + \alpha_1 \mathbf{u}.$

- Q2-5: Obtain the transfer function of field controlled servo motors. Jun. 2013, 8 Marks Q2-6: Obtain the transfer function of an armature controlled dc servo motor. Jun. 2009, 6 Marks
- **Q2-7:** Define the transfer function. Explain Mason's gain formula for determining the transfer function from signal flow graphs. Dec. 2010, 6 Marks

Q2-8: Explain briefly the following terms: Jul. 2009, 8 Marks (ii) Path gain. (iv) Canonical form. (iii) Loop gain. (i) Forward path.

Q2-9: Draw the signal flow graph for the system of equation given below and obtain the overall transfer function $\frac{X_6(s)}{X_1(s)}$ using MGF Jul. 2007, 12 Marks

> $X_2 = G_1 X_1 - H_1 X_2 - H_2 X_3 - H_6 X_6$ $X_3 = G_1 X_1 + G_2 X_2 - H_3 X_3$ $X_4 = G_2 X_2 + G_3 X_3 - H_4 X_5$ $X_5 = G_4 X_4 - H_5 X_6$ $X_6 = G_5 X_5$

Dec. 2014, 10 Marks

Dec. 2012, 4 Marks

Q2-10: For a negative feedback control system, starting from fundamentals, show that the closed loop transfer function M(s) is given by Jan. 2007, 8 Marks

$$\begin{split} M(s) &= \frac{N_g D_h}{D_g D_h + N_g N} \qquad, \\ \text{where } G(s) &= N/Dg \qquad \qquad \text{forward path gain} \\ H(s) &= \frac{N_h}{D_h} \qquad \qquad \text{feedback gain} \end{split}$$

Q2-11: Define the term "transfer function". The unit step response of single loop, unity feedback control system is given by,

$$c(t) = 1 - 1.25e^{-2t} + 0.25e^{-10t}$$

Determine its closed loop and open loop transfer functions. Jul. 2006, 08 Marks

Q2-12: Define the term "transfer function". The unit step response of single loop, unity feedback control system is given by,

$$c(t) = 1 - 3e^{-2t} + 2e^{-3t}$$

Jan. 2006, 06 Marks

Determine its closed loop and open loop transfer functions.

- Q2-13: Obtain a block diagram representation and evaluate the transfer function of an armature controlled dc motor. Jan. 2006, 08 Marks
- Q2-14: Illustrate how to perform the following in connection with block diagram reduction techniques: Jul. 2005, 10 Marks
 - (i) moving a summing point ahead of a block and behind a block.
 - (ii) moving a take off point ahead of a block and behind a block.
 - (iii) Transforming a non unity feedback to a unity feedback.
- **Q2-15:** Consider the system shown in 1. An armature-controlled dc servomotor drives a load consisting of the moment of inertia J_L and load torque T_L . Field current i_f is constant. The torque developed by the motor is T_m . The moment of inertia of the motor rotor is J_m . The viscous friction coefficient of motor is B_m . The angular displacements of the motor rotor and the load element are θ_m and θ_L , respectively. The gear ratio is $n = \frac{N_1}{N_2} = \frac{\theta_L}{\theta_m}$. Draw a block diagram and obtain the transfer functions $\frac{\Theta_L(s)}{E_a(s)}$ and $\frac{\Theta_L(s)}{T_L(s)}$.

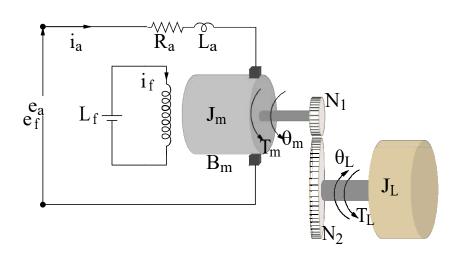


Figure 1: Armature-controlled dc servomotor system.

Q2-16: Consider the system shown in 2. An field-controlled dc servomotor drives a load consisting of the moment of inertia J_L and load torque T_L . Armature current i_a is constant. The torque developed by the motor is T_m . The moment of inertia of the motor rotor is J_m . The viscous friction coefficient of motor is B_m . The angular displacements of the motor rotor and the load element are θ_m and θ_L , respectively. The gear ratio is $n = \frac{N_1}{N_2} = \frac{\theta_L}{\theta_m}$. Draw a block diagram and obtain the transfer functions $\frac{\Theta_L(s)}{E_f(s)}$ and $\frac{\Theta_L(s)}{T_L(s)}$.

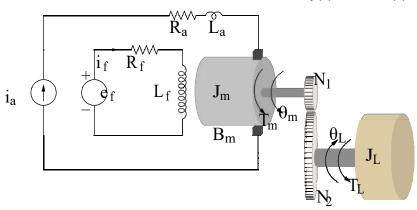


Figure 2: Field-controlled dc servomotor system.

Q2-17: Consider the system shown in 3. An armature-controlled dc servomotor drives a load with torque T_L . The torque developed by the motor is T_m . The moment of inertia of the motor rotor is J_m . The viscous friction coefficient of motor is B_m . The angular speed of the motor rotor is ω . Obtain the transfer functions $\frac{\omega(s)}{T_T(s)}$ and $\frac{\omega(s)}{T_T(s)}$.

Q2-18: Consider the system shown in 4. An armature-controlled dc servomotor drives a load consisting of the moment of inertia J_L and load torque T_L . Field current i_f is constant. The torque developed by the motor is T_m . The moment of inertia of the motor rotor is J_m . The viscous friction coefficient of motor is B_m . The angular displacements of the motor

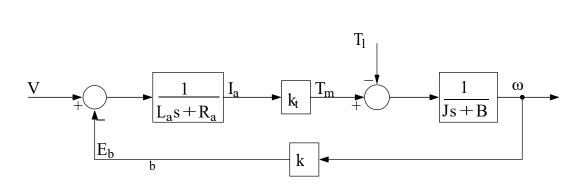


Figure 3: Block diagram of an armature-controlled dc servomotor system.

rotor and the load element are θ_m and θ_L , respectively. Draw a block diagram and obtain the transfer functions $\frac{\Theta_L(s)}{E_a(s)}$ and $\frac{\Theta_L(s)}{T_L(s)}$.

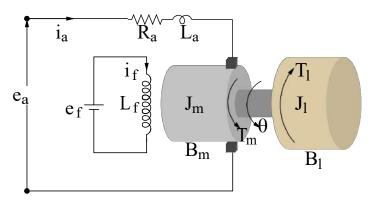


Figure 4: Armature-controlled dc servomotor system.

Q2-19: Consider the system shown in 5. An field-controlled dc servomotor drives a load consisting of the moment of inertia J_L and load torque T_L . Armature current i_a is constant. The torque developed by the motor is T_m . The moment of inertia of the motor rotor is J_m . The viscous friction coefficient of motor is B_m . The angular displacements of the motor rotor and the load element are θ_m and θ_L , respectively. The gear ratio is $n = \frac{N_1}{N_2} = \frac{\theta_L}{\theta_m}$. Draw a block diagram and obtain the transfer functions $\frac{\Theta_L(s)}{E_f(s)}$ and $\frac{\Theta_L(s)}{T_L(s)}$.

Ľa

J_m

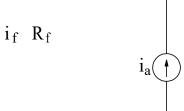
 B_m

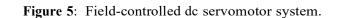
 J_1

 B_1

 \hat{R}_{a}

ef





Q2-20: Find the transfer function $\frac{X(s)}{E(s)}$ for the electro-mechanical system shown in Figure.6.

For a simplified analysis, assume that the coil has a back emf $e_b = K_b \frac{dx}{dt}$ and the coil current i produces a force $F_c = K_f i$ on the mass M.

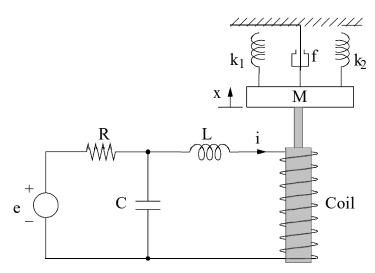


Figure 6: Electro-mechanical system.

Q2-21: Write the dynamic equations and draw a block diagram for the circuit shown in Figure. 7, also determine $\frac{V_1(s)}{E_1(s)}, \frac{V_2(s)}{E_1(s)}, \frac{V_1(s)}{E_2(s)}$ and $\frac{V_2(s)}{E_2(s)}$ by block diagram reduction technique.

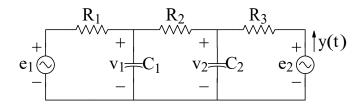


Figure 7:

Q2-22: Write the dynamic equations and draw a block diagram for the circuit shown in Figure. 8, also determine $\frac{V_1(s)}{V_s(s)}, \frac{V_2(s)}{V_s(s)}, \frac{V_1(s)}{I_s(s)}$ and $\frac{V_2(s)}{I_s(s)}$ by block diagram reduction technique.

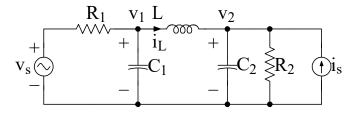


Figure 8:

Q2-23: Write the dynamic equations for the notch circuit shown in Figure. 9 and determine $\frac{E_o(s)}{E_i(s)}$. Also draw a block diagram and determine $\frac{E_o(s)}{E_i(s)}$ by block diagram reduction tech-

nique.

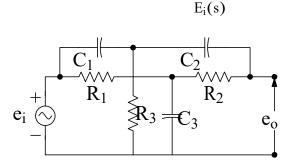


Figure 9:

Q2-24: An equivalent circuit of an electronic amplifier is shown in Figure.10. Determine its transfer function $\frac{V_{out}(s)}{V_{in}(s)}$. Take $v_{Gk} = v_G - v_K$

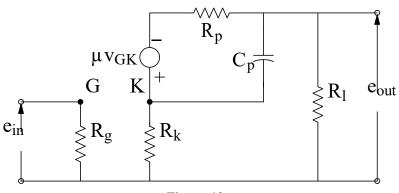
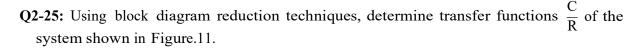


Figure 10:



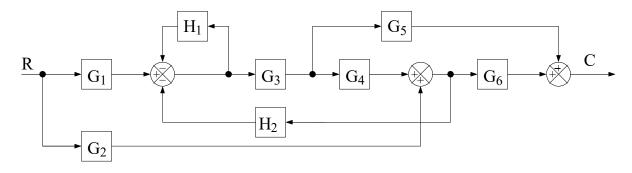


Figure 11:

Q2-26: Using block diagram reduction techniques, determine transfer functions $\frac{C}{R}$ of the system shown in Figure.12.

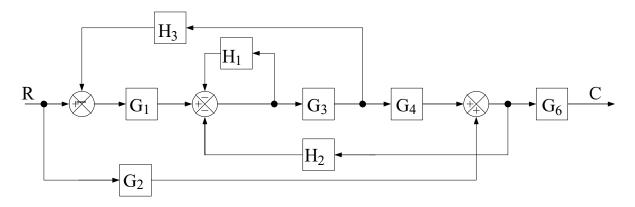
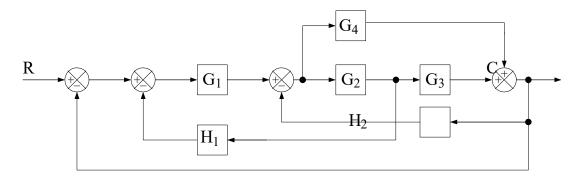


Figure 12:

Q2-27: Using block diagram reduction techniques, determine transfer functions $\frac{C}{R}$ of the system shown in Figure.13.

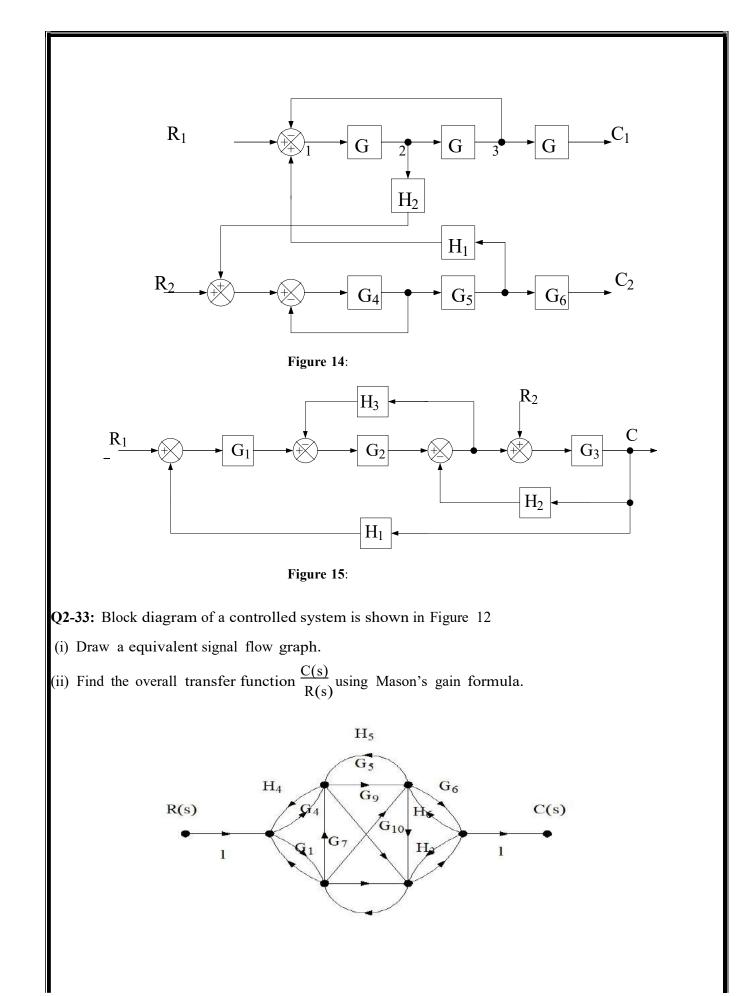




- **Q2-28:** Using block diagram reduction techniques, determine transfer functions $\frac{C_1}{R_1}, \frac{C_2}{R_1}$ $\frac{C_1}{R_2}$, and $\frac{C_2}{R_2}$ of the system shown in Figure.14.
- Q2-29: Using block diagram reduction techniques, determine transfer functions $\frac{1}{R_1}$, and $\frac{C}{R_2}$ of the system shown in Figure.15.

Q2-30: For the signal flow graph shown in Figure 16 obtain transfer function $\frac{C(s)}{R(s)}$. $\frac{C(s)}{R(s)}$ Q2-31: For the signal flow graph shown in Figure 17 obtain transfer function **Q2-32:** For the signal flow graph shown in Figure 18 obtain transfer functions $\frac{C_1(s)}{R_1(s)}, \frac{C_2(s)}{R_1(s)}$ $\frac{C_1(s)}{R_2(s)}$ and $\frac{C_2(s)}{R_2}$.

(s)



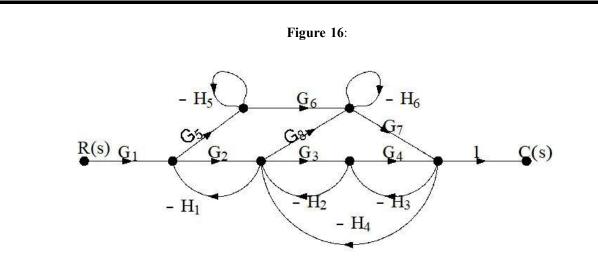


Figure 17:

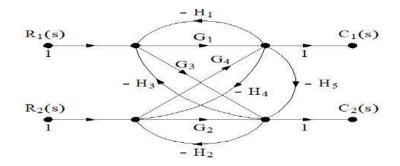


Figure 18:

MODULE-3

Syllabus: Time Response of feedback control systems: Standard test signals, Unit step response of First and Second order Systems. Time response specifications, Time response specifications of second order systems, steady state errors and error constants. Introduction to PI, PDand PID Controllers (excluding design).

Study Material Referred:

- Modern Control engineering-K Ogata.
- > Control Systems Engineering- J.Nagarath and M.Gopal.
- > Automatic Control Systems-Benjamin C. Kuo.
- Linear Control Systems –B S Manke.
- Control Systems- Anand Kumar.
- > VTU Previous year Question papers (2010-2016).
- > Problems and solution of controls systems -AK Jairath.

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2	Standard test signals	
3	Unit step response of First order Systems.	
4	Unit step response of Second order Systems.	
5	Time response specifications of second order systems,	
6	Steady state errors and error constants.	
7	Problems to be solves in class on Time Response.	
8	Solved Problems on Time Response.	
9	Practice Problems on Time Response.	
10	Assignment Problems on Time Response.	
	To be submitted before 2 ND Module Test	
11	Introduction to PI, PD and PID Controller.	
12	Theory Questions.	

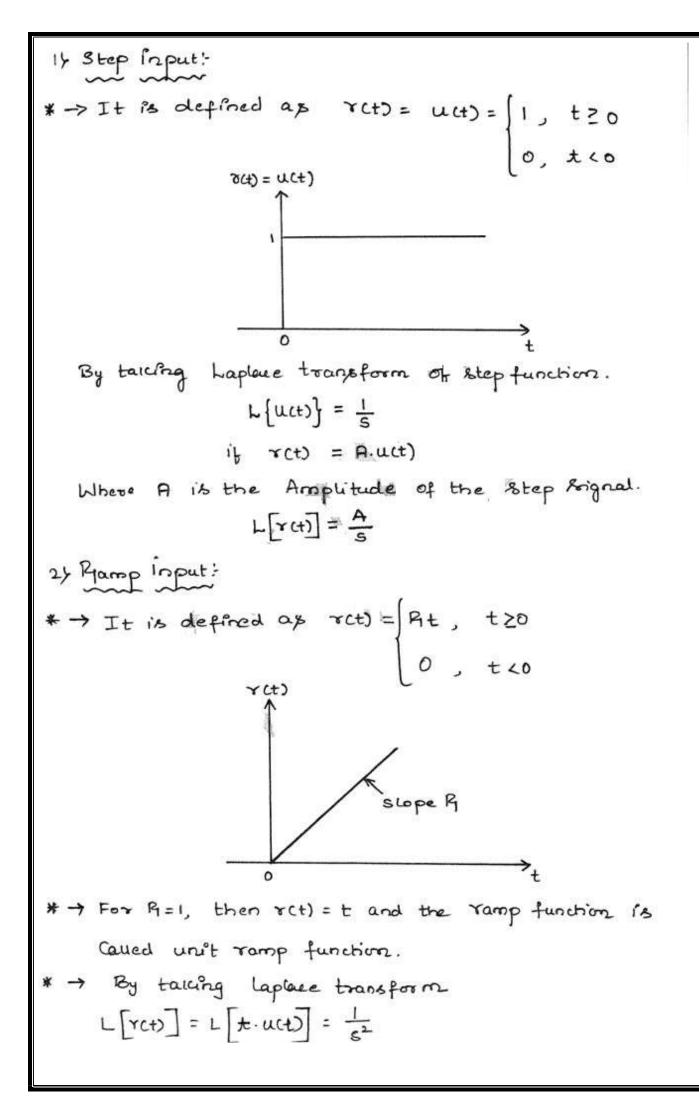
Note: 1) Assignment Carries 05 Marks (To be submitted before 2^{ND} test).

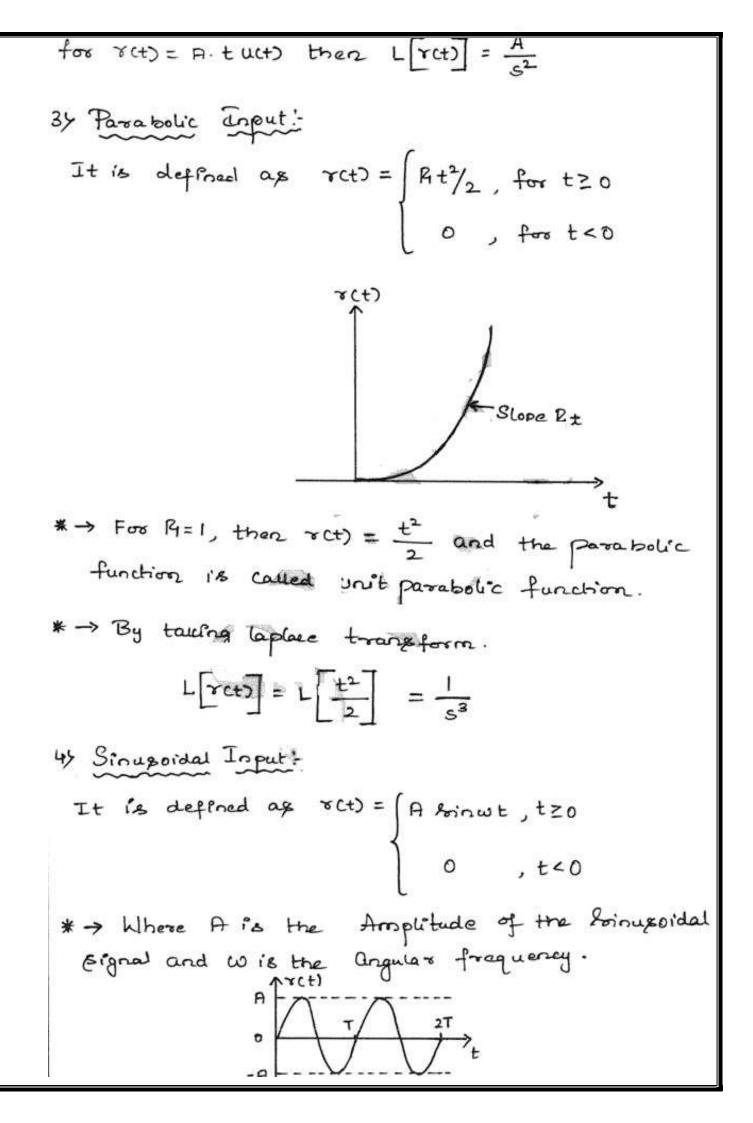
Module 2: Time Response Of Feedbaus Control Systems * > Time response is the Variation of the Controlled Dutput of the system with respect to time, when the system is subjected to test signal. * > Time response of Control system in defined as, To howa system behaves in accordance With time When a specified input test signal in applied. * > Thus the time response of a control system in divided in to two parts they are as Transient Response b) Steady State Ruponse response of a Costrolled System * -> The typi for a specified input test signal is as shown below. Steady State Y(t) Transient - K Steady (a) Input test Signal. State (b) Output Vesponse. * > Transient response of the system is defined as the past of time response. that goes to zero as time becomes very large. Thus Ct(t) has property that

- i.e. $t \to \infty C_t(t) = 0$
- F→ The transient part of the time response reveals the nature of the response and gives an indication of speed
- → The steady state response is the part of total response that remains after the transient has died but.
- (-> The Steady State part reveals the accuracy of Control System.
- t -> Steady state forces is Observed if the actual o/p docsnot match with in the Dutput.
- * > From the fig, the o/p corp is given by.

 $C(t) = C_{tr}(t) + C_{ss}(t)$

- t → The different input Signals that are used to test a Control System.
 - 14 Step Papet.
 - 2> Ramp input.
 - 37 Parabolic input.
 - 44 Impulse izput.
 - sy Sinysoidal Proput.





$$L \left[\beta \sin (\omega t) \right] = \frac{\beta \cdot \omega}{s^2 + \omega^2}$$
5) Impute Imput:
* > It is defined as $r(t) = \left\{ \delta(t) = 1, t = 0; 0, t \neq 0; \right\}$

* > By totang laplate transforms

$$L \left[r(t) \right] = L \left[\delta(t) \right] = 1$$
Note:
1) $L \left[\beta \cdot u(t) \right] = \frac{A}{s}$

2) $L \left[t^0 \cdot u(t) \right] = \frac{M_1}{s^{-1}}$

3) $L \left[e^{t \cdot \alpha t} \right] = \frac{M_1}{s^{-1}}$

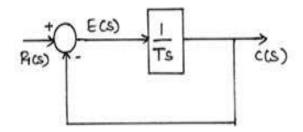
4) $L \left[sin \omega t \right] = \frac{\omega}{s^2 + \omega^2}$

5) $L \left[cos \omega t \right] = \frac{s}{s^2 + \omega^2}$

5) $L \left[e^{t \cdot \alpha t} \right] = \frac{s}{(s \mp \alpha)^2 + \omega^2}$

Time Respose Of First Order Control System.

* > Consider the block diagram of a general first Order Control System i'r as shown below.



* > The Overall transfer function is given by.

$$\frac{C(S)}{R(S)} = \frac{1}{1 + \frac{1}{TS}} = \frac{1}{TS + 1}$$

$$\frac{C(s)}{R(s)} = \frac{1}{sT+1}$$

* > 'T' is the time constant of the System.

- * -> The Highest power of 's' in the denominator of the Overall transfer function represents the Order of the control System.
- * > The Sutput of the System is Expressed as.

W.K.T that $P_1(s) = \frac{1}{s}$. $P_1(t) = 1 L[R(t)] = \frac{1}{s}$

$$\therefore C(S) = \frac{1}{S} \cdot \frac{1}{ST+1}$$

Breaking R.H.S in to Partial fractions

$$c(s) = \frac{1}{s} \cdot \frac{1}{sT+1} = \frac{R}{s} + \frac{R}{sT+1} \longrightarrow 0$$

$$R = \lim_{s \to 0} \left[s \cdot c(s) \right] = \lim_{s \to 0} \left\{ \frac{1}{sT+1} \right\} = 1$$

$$B = \lim_{s \to 0} \left[(sT+1) c(s) \right] = \lim_{s \to -\frac{1}{T}} \left\{ \frac{1}{s} \right\} = -T$$

$$By \text{ substituting the Value A and B in Equation 0}$$

$$c(s) = \frac{1}{s} - \frac{T}{sT+1}$$

$$C(s) = \frac{1}{s} - \frac{1}{s+\frac{T}{T}}$$

$$By \text{ taking inverse leplace tractions.}$$

$$c(t) = 1 - e^{-t/T} \quad \text{for } t \ge 0 \quad \longrightarrow \infty$$

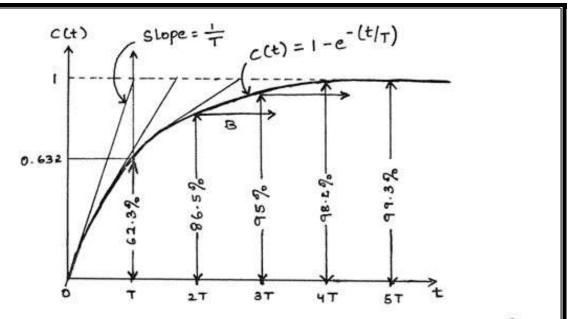
$$* \rightarrow \text{ Equation R instead that initially the output (lt)}$$
is given and finally it becomes unity.

$$* \rightarrow \text{ Right reached 63.29, of its total change.}$$

$$i.e. C(T) = 1 - e^{-T/T} = 1 - e^{-1} = 0.632$$

$$C(T) = 0.632.$$

the faster the system response. The response is as shown below.



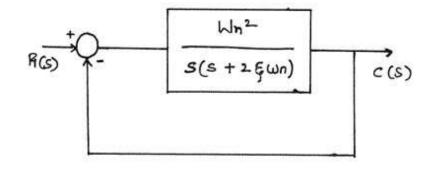
* The slope of the tangent line at t=0 is 1/7 Since

$$\frac{d}{dt} \Big|_{t=0} = \frac{1}{T} e^{t/T} \Big|_{t=0} = \frac{1}{T} - 3$$

- *→ The Jutput Would reach the final Value at t=T, is it main tained its initial speed of response. The sope of the response Curve c(t) monotonically from 1/2 at t=0 to Zero at t=∞
- * -> In one time constant, the Exponential response curve has gone form 0 to 63.2% of the final value. (
- * -> In two time (onstant, the response reaches 20%) of the final value. (At t = 2T)
- * -> At t=37, 47 and 57, the response reaches 95%, 8.2% and 99.3% respectively of the final Value.
- * -> Thus For t Z4T, the response remains with in 2% of the final value
- * → As Seen from Equation @ the steady state is reached mathematically only after an infinite time.

Time Gesponse Of Second Order Control System

* > The block diagram of a general second Order Control System is as shown below.



* > The Overall transfer function ingiven by.

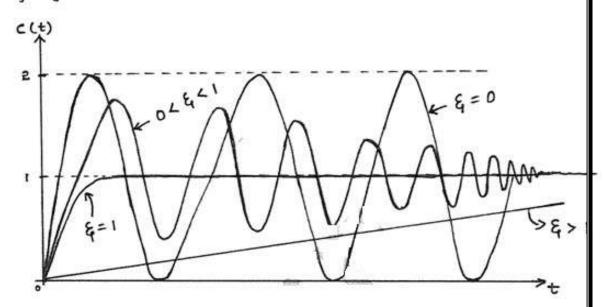
$$\frac{C(s)}{R(s)} = \frac{G(s)}{1+G(s)H(s)} = \frac{s(s+z_{\text{E}}\omega_n)}{1+\frac{\omega_n^2}{s(s+z_{\text{E}}\omega_n)}} \cdot 1$$

$$\frac{C(S)}{R(S)} = \frac{Wn^2}{S^2 + 2\xi w_0 S + w_0^2}$$

*→ Wn 1's Known as the natural frequency.
*→ § 1's the damping ratio of the System.
*→ If §=0, the System is undamped.
*→ If 0< §<1, the System is under damped.
*→ If §=1, the System is Under damped.
*→ If §=1, the System is Over damped.
*→ If §>1, the System is Over damped.
*→ If §>1, the System is Over damped.
*→ Denominator roots of C(S) are Known as the poles of the closed loop transfer function and.
*→ Numerator roots of C(S) are Known as the Sero's of the closed loop transfer function.

V

* -> The time response of a second Order control System subjected to a unit step input for different values of damping ratio is as shown below.



- * -> The time response of any system is characterized by the roots of the denominator polynomial, which in fact are the poles of the transfer function.
- * -> The denominator polynomial is therefore called the characteristic polynomial and it is called the characteristic Equation.
- * -> i.e 1+G(s)H(s) is Gnown as characteristic Equali

$$\begin{aligned} 1+G(s)H(s) &= 0 \\ S^{2}+2\xi\omega_{n}S+\omega_{n}^{2}=0 \\ S &= \frac{-a\xi\omega_{n}\pm\sqrt{4\xi^{2}\omega_{n}^{2}-4\omega_{n}^{2}}}{a} \quad \therefore \ c=-\frac{b\pm\sqrt{b^{2}-4ac}}{aa} \\ &= -\xi\omega_{n}\pm\sqrt{\xi^{2}\omega_{n}^{2}-\omega_{n}^{2}} \\ S &= -\xi\omega_{n}\pm\omega_{n}\sqrt{\xi^{2}-1} \\ \end{aligned}$$
When $\xi=0$ poles are present ϑ_{2} imaginary axis b

plane.

maginary

* → When &= 1 Poles of the System are fiel and Equal
* → When &> 1 Poles of the System are field and Unlaw.
* → When & Lies between 0+1 Poles of the System are Complex.
* → for o< &< 1 S = -& Wn ± j Wn √1-&²

Where EWD is Known as the damping factor of the System. This terms descides the rate of delay of the transcrent response.

ut is known as damped frequency of Oscillation. It is the angular frequency of the transcient response.

* -> Repponse of a second Order Control System Subjected to a unit step input. The Overall transfer function of a Second Order Control System is given by.

$$\frac{c(s)}{R(s)} = \frac{wn^2}{s^2 + 2\xi wn s + wn^2}$$

$$C(s) = \frac{\omega_n^2}{s^2 + 2\xi \omega_n s + \omega_n^2} \cdot R(s) \rightarrow 0$$

(1)

N.K.T
$$P(S) = Unit step (nput)$$

i.e $P_1(S) = \frac{1}{S}$
By substituting the value of $P_1(S)$ in

$$C(s) = \frac{Wn^2}{s^2 + 2\xi Wns + Wn^2} \cdot \frac{1}{s}$$

$$C(S) = \frac{Wn^{2-1}}{S(S^{2} + 2\xi WnS + Wn^{2})}$$

Breaking R.H.S in to Parthial Frenchions.

$$C(S) = \frac{Wn^{2-1}}{S(S^{2} + 2\xi WnS + Wn^{2})} = \frac{W_{1}}{S} + \frac{K_{2}S + K_{3}}{S^{2} + 2\xi WnS + Wn^{2}} \Rightarrow @$$

$$Wn^{2} = K_{1} (S^{2} + 2\xi WnS + Wn^{2}) + (H_{2}S + K_{3})S$$
Fut S=0; $Wn^{2} = K_{1} Wn^{2}$; $K_{1} = \frac{Wn^{2}}{Wp^{2}}$ $\frac{[K_{1} = 1]}{Wp^{2}}$
By Comparing the Constrainty of S²; S^{2} ; $O = K_{1} + K_{2}$ or $K_{12,2} = K_{1}$;

$$\frac{[K_{2} = -1]}{S^{2}; O = K_{1} + K_{2}}$$
 or $K_{12,2} = K_{1}$;

$$\frac{[K_{2} = -1]}{S}$$
By Comparing the Constrainty of S³
 $S: O = 2\xi WnK_{1} + K_{3}$
 $K_{3} = -2\xi WnK_{1}$; $\frac{[K_{1} = 1]}{[K_{3} = -2\xi Wn]}$
By Substituting the Value of K_{1} , H_{2} and K_{3} in (2)
 $C(S) = \frac{1}{S} + \frac{-S - 2\xi Wn}{S^{2} + 2\xi WnS + Wn^{2}}$
 $= \frac{1}{S} - \frac{S + 2\xi Wn}{S^{2} + 2\xi WnS + (\xi Wn)^{2} - (\xi Wn)^{2} + Wn^{2}}{(\xi + \xi_{1})^{2}}$

$$C(s) = \frac{1}{s} - \frac{(s+s\omega_{1})+s\omega_{1}}{(s+s\omega_{1})^{2}+\omega_{1}^{2}(1-s^{2})}$$

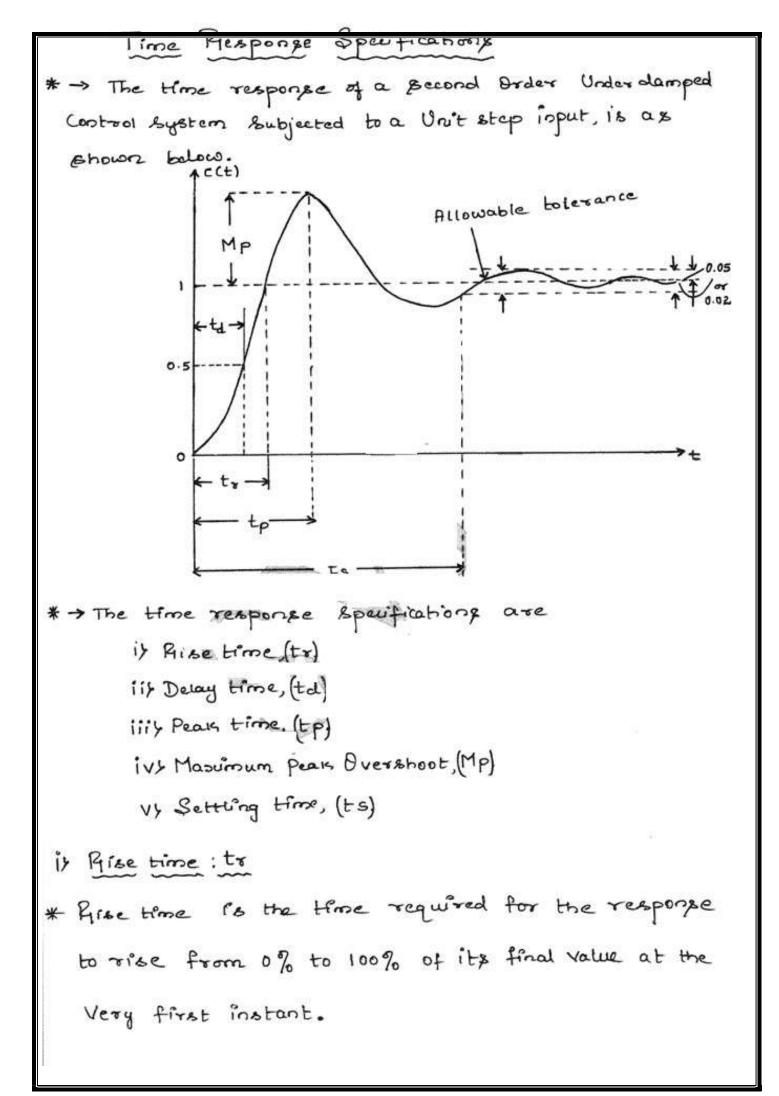
$$= \frac{1}{s} - \frac{(s+s\omega_{1})+s\omega_{1}}{(s+s\omega_{1})^{2}+\omega_{1}^{2}}$$
We know that $W_{d} = w_{n}\sqrt{1-s^{2}}$

$$C(s) = \frac{1}{s} - \left[\frac{(s+s\omega_{1})}{(s+s\omega_{1})^{2}+\omega_{1}^{2}} + \frac{s\omega_{n}}{(s+s\omega_{1})^{2}+\omega_{1}^{2}}\right]$$
By multiplying W for both numerates and denominator of 3^{rd} is real denominator of $(s+s\omega_{1})^{2}+\omega_{1}^{2}$

$$C(s) = \frac{1}{s} - \frac{(s+s\omega_{1})}{(s+s\omega_{1})^{2}+\omega_{1}^{2}} - \frac{s\omega_{n}\cdot w_{1}}{(w_{1}(s+s\omega_{1})^{2}+\omega_{1}^{2})}$$
By taking forverse laplace transforms.

$$C(t) = 1 - e^{\frac{1}{s}\omega_{n}t} - \frac{s\omega_{n}t}{(s+s\omega_{1})^{2}+\omega_{1}^{2}} e^{-\frac{1}{s}\omega_{1}t}$$
By taking forverse laplace transforms.

$$C(t) = 1 - e^{\frac{1}{s}\omega_{n}t} - \frac{s\omega_{n}t}{(s+s\omega_{n})(1-s^{2})} e^{-\frac{1}{s}\omega_{1}t}$$
By taking $e^{-\frac{1}{s}\omega_{1}t} - \frac{s\omega_{1}}{(s+s\omega_{1})(1-s^{2})} e^{-\frac{1}{s}\omega_{1}t}$
By taking $e^{-\frac{1}{s}\omega_{n}t} - \frac{s\omega_{n}t}{(s+s\omega_{n})(1-s^{2})} e^{-\frac{1}{s}\omega_{1}t}$
By taking $e^{-\frac{1}{s}\omega_{n}t} - \frac{s\omega_{n}t}{(s+s\omega_{n})(1-s^{2})} e^{-\frac{1}{s}\omega_{1}t}$
By taking $e^{-\frac{1}{s}\omega_{n}t} - \frac{1}{\sqrt{1-s^{2}}} e^{-\frac{1}{s}\omega_{1}t}$
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By taking $e^{-\frac{1}{s}\omega_{1}t} - \frac{1}{\sqrt{1-s^{2}}} e^{-\frac{1}{s}\omega_{1}t}$
By taking $e^{-\frac{1}{s}\omega_{n}t} - \frac{1}{\sqrt{1-s^{2}}} e^{-\frac{1}{s}\omega_{1}t}$
By taking $e^{-\frac{1}{s}\omega_{1}t} - \frac{1}{\sqrt{1-s^{2}}} e^{-\frac{1}{s}\omega_{1}t} - \frac{1}$



2) Delay time ; td

* The delay time is the time required for the response to reach 50% of the final value at the Very first time.

34 Peaks time : tp

* The peak time is the time required for the response to reach the first peak of the overshoot.

44 Masumum Overshoot : Mp

* The maximum peaks Overshoot is the maximum peaks Value of the response curve measured from unity. * If the final steady-state value of the response differs from unity, then it is common to use to. maximum percent overshoot. It is defined by

Peak over shoot: Mp = c(tp) - c(0)

Mascimum percent Overphoot:
$$\%Mp = \frac{C(tp) - C(\infty)}{C(\infty)} \times 100$$

* The amount of the maximum (Percent) Overshoot directly indicates the relative stability of the System.

54. Settling time: ts

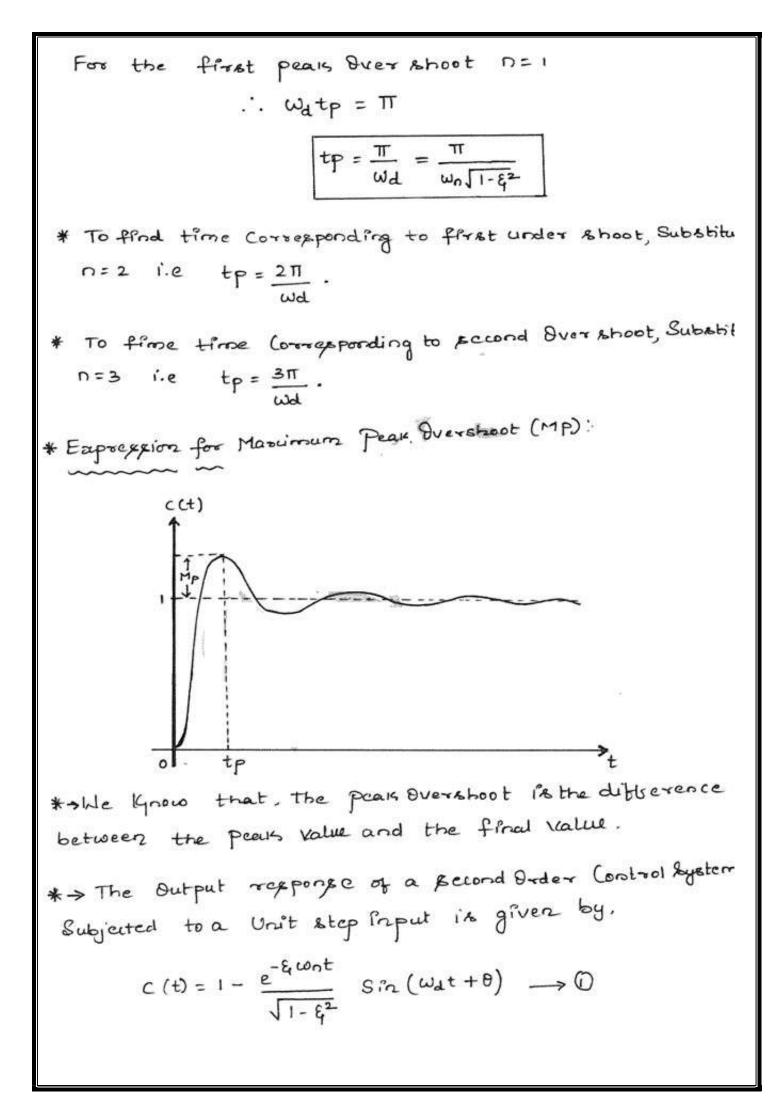
* The Setting time is the time required for the repponse Curve to reach and stay Within specified range of its final value (Within tolerance band, usually 2% or 5%).

*
$$\rightarrow$$
 Sin ($\omega_{d}t_{7} + \theta$) must be level to zero.
... Sin ($\omega_{d}t_{7} + \theta$) = 0
 $\omega_{d}t_{7} + \theta = \Pi T$; $n = 1, 2, 3$
 $\omega_{d}t_{7} = \Pi T - \theta$
For the Ist time $n = 1$
 $\omega_{d}t_{7} = \Pi - \theta$
Therefore, the rise time t_{7}^{*} is given by
 $t_{7} = \frac{\Pi - \theta}{\omega_{d}} \rightarrow \Theta$
 $\omega_{d} = \tan^{-1} \sqrt{1 - \xi^{2}}$
 $\omega_{d} = \omega_{0} \sqrt{1 - \xi^{2}}$
* \rightarrow By Substituting Pr Equations Θ we get.
 $\Pi = tan^{-1} \sqrt{1 - \xi^{2}}$

$$T = \frac{1-\xi^2}{4}$$

$$T_{8} = \frac{\xi}{1-\xi^2}$$

$$\begin{split} & \text{ILIAT} \qquad \text{ILI$$



**By kubshituking t=tp in Equation (1), we get the peak
reprove or massimum response.
.: Maximum response
$$C(tp) = 1 - \frac{e}{e^{-\frac{1}{2}th}hp}$$
 $Sin (Wd tp + 0)$
 $C(tp) = 1 - \frac{e^{-\frac{1}{2}Wh} \frac{11}{\sqrt{1-\frac{1}{2}^{2}}}}{\sqrt{1-\frac{1}{2}^{2}}} Sin (Wd + 0)$ $\therefore tp = \frac{11}{Wd}$
 $\psi_{d} = Wh (1-\frac{1}{2})$
 $\therefore C(tp) = 1 - \frac{e^{-\frac{1}{2}Wh} \sqrt{1-\frac{1}{2}^{2}}}{\sqrt{1-\frac{1}{2}^{2}}} Sin (Wd + 0)$ $\therefore tp = \frac{11}{Wd}$
 $\psi_{d} = Wh (1-\frac{1}{2})$
 $\therefore C(tp) = 1 - \frac{e^{-\frac{1}{2}Wh} \sqrt{1-\frac{1}{2}^{2}}}{\sqrt{1-\frac{1}{2}^{2}}} Sin (1+0)$
 $Sin (1+0) = -Sin 0$
 $\therefore C(tp) = 1 - \frac{e^{-\frac{1}{2}Wh} \sqrt{1-\frac{1}{2}^{2}}}{\sqrt{1-\frac{1}{2}^{2}}} \cdot (-Sin 0)$ $\therefore Sin 0 = \sqrt{1-\frac{1}{2}}$
 $i.e \frac{C(tp) = 1 + e^{-\frac{1}{2}W/\sqrt{1-\frac{1}{2}^{2}}}}{\sqrt{1-\frac{1}{2}^{2}}} \sqrt{1-\frac{1}{2}^{2}}}$
 $\psi \rightarrow we Know that$
 $Peak Over shoot is given by$
 $Mp = C(tp) - C(\infty)$
 $Mp = \frac{1}{2} + \frac{e^{\frac{1}{2}W/\sqrt{1-\frac{1}{2}^{2}}}}{-\frac{1}{2}} \cdots C(\infty) = 1 (final \frac{1}{2})$
 $Mp = e^{\frac{1}{2}W} \sqrt{1-\frac{1}{2}^{2}}}$

* Note: If the system is subjected to a step friput

$$q = C(tp) - C(\infty)$$

 $C(\infty)$
 $Rs, in this lare the $C(\infty) = 1$ % peak, Biver short
 ik given by
 $g Mp = C(tp) * 100$
 $\therefore g Mp = c^{\frac{1}{2}\pi}/\sqrt{1-q^{2}} * 100$
* The Variation of peak diver shoot work damping ratio
 ik as shown below.
 $\int_{0}^{Mp} \frac{e^{-\frac{1}{2}\pi}}{q} \frac{1}{q} \frac{e^{-\frac{1}{2}\pi}}{q} \frac{1}{q} \frac{1}{$$

% peaks Overschoot:
% Mp =
$$C(tp) - C(\infty)$$
 * 100
= $\frac{gre^{-\frac{1}{2}\pi}}{C(\infty)}$ * 100
 $\frac{gre^{-\frac{1}{2}\pi}}{\sqrt{1-\frac{1}{2}^{2}}}$ * 100
Eapression for Setting time:
* \rightarrow we know that, the sattling time is the time
required for the response Curve to reach and
Stay with is specified range 4 it's final value.
 $\frac{1+e^{-\frac{1}{2}}wnt}{1+\frac{e^{-\frac{1}{2}}wnt}{1+\frac{1}{2}^{2}}}$
* The Dutput response of a second Order System
Subjected to a unit step input in given by:
 $C(t) = 1 - \frac{C}{\sqrt{1-\frac{1}{2}^{2}}}$ sin $(w_{d}t+\theta) \rightarrow 0$

* The Envelop of time response its 0,8 khown above
and the Envelop time response its given by
Upper bound

$$1 + \frac{e}{\sqrt{1-\xi^2}}$$

tower bound
 $1 + \frac{e}{\sqrt{1-\xi^2}} = \frac{-\xi W_n t}{\sqrt{1-\xi^2}}$
* \Rightarrow From the response the Upper bound is given by
 $1 + \frac{e}{\sqrt{1-\xi^2}} = 1.05$
* \Rightarrow From the response the lower bound is given by
 $1 - \frac{e}{\sqrt{1-\xi^2}} = 0.95$
* \Rightarrow From the response the lower bound is given by
 $1 - \frac{e}{\sqrt{1-\xi^2}} = 0.95$
* \Rightarrow At the satting time ts for 5% tollerance (In Both Gses
We get,
 $\frac{-\xi W_n t_s}{\sqrt{1-\xi^2}} = 2.05$
 $\sqrt{1-\xi^2}$
* \Rightarrow By taking natural log θ_n both Anders.
 $\ln \left[e^{-\xi W_n t_s}\right] = \ln \left[0.05 \cdot \sqrt{1-\xi^2}\right]$
Note: $\ln \left[x^{4}\right] = y \cdot \ln (x)$
 $\ln \left[x \cdot y\right] = \ln (x) + \ln (y)$
 $-\xi W_n t_s = \left[\ln 0.05 + \ln \sqrt{1-\xi^2}\right]$
 $t_s = -\frac{1}{-2} \left\{-2.99 + \ln \sqrt{1-\xi^2}\right\}$

$$t_{s} = \frac{-1}{\xi w_{n}} \left\{ -2 \cdot 9 \right\} \quad \therefore \quad \ln \sqrt{\leq 0} = \infty$$

$$* \rightarrow \text{Approximate 5\% Setting time.}$$

$$t_{s} = \frac{3}{\xi w_{n}}$$

$$* \rightarrow \text{Similarly for 2\% Setting time.}$$

$$t_{s} = \frac{1}{\xi w_{n}} \left\{ \ln 0.02 + \ln \sqrt{1-\xi^{2}} \right\}$$

$$t_{s} = \frac{1}{\xi w_{n}} \left\{ \ln 0.02 \right\}$$

$$t_{s} = \frac{1}{\xi w_{n}} \left\{ \ln 0.02 \right\}$$

*
$$\Rightarrow$$
 From the definition of transfer function W.B.T

$$H(S) = \frac{B(S)}{C(S)} \quad \text{or } B(S) = C(S) H(S)$$
* \Rightarrow Substituting for $B(S)$ in (3) we get

$$E(S) = R(S) - C(S) \cdot H(S)$$
From Fig; $C(S) = E(S) G(S)$
 $\therefore E(S) = R(S) - E(S) G(S) H(S)$

$$E(S) + E(S) G(S) H(S) = R(S)$$

$$E(S) + E(S) G(S) H(S) = R(S)$$

$$E(S) = \frac{R(S)}{1 + G(S) H(S)} = R(S)$$

$$\therefore E(S) = \frac{R(S)}{1 + G(S) H(S)} \rightarrow (9)$$
If the Systems is unity feedbound,

$$E(S) = \frac{R(S)}{1 + G(S)} \rightarrow (9)$$

$$Error T.F (A given by
$$\frac{E(S)}{R(S)} = \frac{1}{1 + G(S)} + H(S)$$
Steady state Errors is given by $= \lim_{S \to 0} S \cdot E(S)$
* \Rightarrow By Substituting Equation (9) for $E(S)$, we get

$$SE = \lim_{S \to 0} S \cdot \frac{R(S)}{1 + G(S) H(S)}$$$$

Error Constants
* > There are two types of Error Constants they
are.
a) Static Error Constants
by Dynamic Error Constants.
a) Static Error Constants are classified in to 3 bypes
* position Error Constant. (Kp)
* Velocity Error Constant. (Kp)
* Velocity Error Constant. (Kn)
* Acceleration Error Constant. (Kn)
* Dosition Error Constant: (Tt is defined when the
System is Subjected to a Step Paput.
W.K.T. the Standy State Error is given by

$$P_{--} = \lim_{S \to 0} S \cdot E(S)$$

But $E(S) = R(S)$
 $I + G(S) H(S)$
But $R(S)$ is a unit step input $R(S) = \frac{1}{S}$
 $\therefore e_{SS} = \lim_{S \to 0} \frac{S \cdot \frac{1}{S}}{1 + G(S) H(S)}$
 $e_{SS} = \frac{1}{S + 0}$

By defining position Error Constant as $H_{p} = \lim_{s \to 0} G(s) H(s)$ $\therefore SSE \boxed{C_{ss} = \frac{1'}{1 + H_{p}}}$ $\therefore SSE \boxed{C_{ss} = \frac{1'}{1 + H_{p}}}$

* \rightarrow If the system is subjected to a step i/p of Strength 'A' units. i.e r(t) = A u(t) then $R(s) = \frac{A}{s}$ \therefore SSE, $e_{AS} = \frac{A}{1+K_P}$

* Velocity Error Constant in It is defined when the system is subjected to samp input of Velocity input. The Steady State Error is given by. $SSE, C_{SS} = \lim_{S \to 0} S E(S)$ $C_{SS} = \lim_{s \to 0} S \cdot \frac{R(S)}{1 + G(S) H(S)}$ But R(s) is a unit samp input, i.e., $R(s) = \frac{1}{s^2}$ $\therefore e_{gs} = \lim_{s \to 0} \frac{\sqrt{s} \cdot \frac{1}{s^2}}{1 + G(s) + (s)}$ $e_{ss} = \lim_{s \to 0} \frac{1}{s + s G(s) H(s)}$ ess = -1 0 + 1 m S.G(S)H(S) S→0 $e_{ss} = \frac{1}{\underset{s \to 0}{\overset{lm}{1}} s.g(s) H(s)}$

* > By defining velocity Error constant as

$$H_V = \lim_{S \to 0} S.G(S) H(S)$$

 $\therefore SSE: \boxed{P_{SS} = \frac{1}{H_V}}$
* > If the System is subjected to ramp input of strong
'A' units.
i.e., $v(t) = At u(t)$ then $R(S) = \frac{A}{S^2}$
 $\therefore SSE; \boxed{P_{SS} = \frac{A}{H_V}}$
* > Acceleration Error Constant: It is defined when
the System i's subjected to a parabolic input.
The Steady state Errors is given by.
SSE, $C_{SS} = \lim_{S \to 0} S.E(S)$
 $e_{SS} = \lim_{S \to 0} S.E(S)$
 $e_{SS} = \lim_{S \to 0} \frac{S \cdot \frac{R(S)}{1+G(S)H(S)}}{1+G(S)H(S)}$
But $P_1(S)$ is a parabolic input, i.e. $\frac{R(S) = \frac{1}{S^2}}{1+G(S)H(S)}$
 $e_{SS} = \lim_{S \to 0} \frac{S}{S^2 + S^2G(S)H(S)}$
 $e_{SS} = \frac{1}{O + \lim_{S \to 0} S^2G(S)H(S)}$
 $e_{SS} = \frac{1}{O + \lim_{S \to 0} S^2G(S)H(S)}$

By defining acceleration Error Constant as

$$I_{a} = \lim_{s \to 0} s^{2} G(s) H(s)$$

$$\therefore sse, \quad ess = \frac{1}{H_{a}}$$
* If the System is subjected to a parabolic input
of Strength 'A' units.

$$i.e r(t) = A \frac{t^{2}}{2} \quad then R(s) = \frac{A}{s^{3}}$$

$$\therefore sse, \quad exs = \frac{A}{H_{a}}$$

Note:

$$I_{P}^{i} = \lim_{S \to 0} G(S) H(S) ; SSE = \frac{1}{1+K_{P}}$$

$$I_{V} = \lim_{S \to 0} S_{q}(S) H(S) ; SSE = \frac{1}{H_{Q}}$$

$$K_{a} = \lim_{S \to 0} S_{q}(S) H(S) ; SSE = \frac{1}{H_{Q}}$$
Effects of change in G(S) H(S) on Steady state Error
* \rightarrow The Number of poles of G(S)H(S) at the Origin of
the s-plane gives the type of system. It is given by
G(S) H(S) = $\frac{H(1+ST_{Z1})(1+ST_{Z2})-\cdots}{S^{n}(1+ST_{P1})(1+ST_{P2})}$
The above Equation is also called as Time lenstant
form.
* \rightarrow Centrol System are there fore classified in
accordance with the Number of Integration, i.e., no pole of G(S) H(S) at the Origin of S-plane)
Type - 0 System (n=1, on integration, i.e., one pole of G(S) H(S) at the Origin of S-plane)
Type - 2 System (n=2, two integrations, i.e. two poles of
G(S) H(S) at the Origin of S-plane) and some

teady sture croor igpe o system For a type O' system Conside- $G(S) H(S) = H(1 + ST_{z_1})(1 + ST_{z_2})$ Note: n=0 5 = 1 $(1+STP_1)(1+STP_2)$ * -> We know that, the popition for constant is given by, $14p = \lim_{s \to 0} G(s) H(s)$ $\therefore 14p = \lim_{s \to 0} \frac{4(1 + ST_{z_1})(1 + ST_{z_2})}{(1 + ST_{P_1})(1 + ST_{P_2})} = 14$ steady state Error (popition) is given by $e_{ss}(position) = \frac{1}{1+14p} = \frac{1}{1+14} = finite value$ * -> 111", W. 4.T the velocity Error Constant i's given by 19v = lim S. G(S) H(S) $\frac{14 \, \text{m}}{\text{s} \to 0} \, \text{s} \cdot \frac{14 \, (1 + \text{s} T_{z_1}) \, (1 + \text{s} T_{z_2})}{(1 + \text{s} T_{P_1}) \, (1 + \text{s} T_{P_2})} = 0$ Steady state Error (velocity) is given by $e_{ss}(velocity) = \frac{1}{v} = \frac{1}{0} = \infty$

* 111¹⁴ W.K.T Acceleration Error Constant is given
by

$$I_{a} = \lim_{S \to 0} S^{2}G(S) H(S)$$

$$\therefore I_{a} = \lim_{S \to 0} S^{2} \cdot \frac{H(1+ST_{z_{1}})(1+ST_{z_{2}})}{(1+STP_{p})(1+STP_{p})} = 0$$
Steady State Error (Acceleration) is given by

$$C_{AB} (Acceleration) = \frac{1}{14a} = \frac{1}{0} = \infty$$
Steady State Error : Type 1 System
For a Type 1 System

$$G(S) H(S) = \frac{H(1+ST_{z_{1}})(1+ST_{z_{2}})}{S^{4}(1+STP_{1})(1+STP_{2})}$$
Note:

$$G(S) H(S) = \frac{H(1+ST_{z_{1}})(1+STP_{2})}{S^{4}(1+STP_{1})(1+STP_{2})}$$
Note:

$$given by,$$

$$I_{a} = \lim_{S \to 0} G(S) H(S)$$

$$\therefore H_{p} = \lim_{S \to 0} \frac{H(1+ST_{z_{1}})(1+STP_{2})}{S(1+STP_{1})(1+STP_{2})} = \frac{I_{4}}{0} = \infty$$
Steady state Error (position) is given by

$$e_{BB}(position) = \frac{1}{1+I4p} = \frac{1}{1+\infty} = 0$$

*
$$\rightarrow 111^{19}$$
 IN.16.T the velocity Error (onstant is given
by.
If $y = \int_{300}^{10} SG(S) H(S)$
 $\therefore 14y = \int_{300}^{10} \frac{S \cdot 14(1+STz_1)(1+STz_2)}{S(1+STP_1)(1+STP_2)} = 14$
Steady state Error (velocity) is given by
 C_{SS} (velocity) = $\frac{1}{K_V} = \frac{1}{K} = Finite$ Value.
* $\rightarrow 111^{19}$ W.K.T, the Acceleration Error constant is
given by,
 $K_a = \int_{300}^{10} S^- G(S) H(S)$
 $\therefore 14a = Ure S^2 K(1+STz_1)(1+STz_2) = 0$
Steady state Error (Acceleration) is given by
 C_{SS} (Acceleration) = $\frac{1}{K_1a} = \frac{1}{0} = 10$

Steady state
$$larrow$$
: Type 2 System.
For a Type 2 System:
 $G(S) H(S) = \frac{K_1(1+ST_{21})(1+ST_{22})}{S^2(1+ST_{P1})(1+ST_{P2})}$ Note: $n=2$
 $g(S) H(S) = \frac{K_1(1+ST_{21})(1+ST_{P2})}{S^2(1+ST_{P1})(1+ST_{P2})}$ Note: $n=2$
 $s \rightarrow 0$
 $K_1 p = \lim_{S \to 0} G(S) H(S)$
 $\therefore K_1 p = \lim_{S \to 0} G(S) H(S)$
 $\therefore K_1 p = \lim_{S \to 0} \frac{K_1(1+ST_{21})(1+ST_{22})}{S^2(1+ST_{P1})(1+ST_{P2})} = \frac{K_1}{0} = \infty$
Steady state large position (is given by
 $las (position) = \frac{1}{1+K_1p} = \frac{1}{1+\infty} = 0$
 $K \rightarrow 111^{19}$ we Know that, Veloutly larger (onstant is given by
 $K_1 p = \lim_{S \to 0} S \cdot G(S) H(S)$
 $\therefore K_1 v = \lim_{S \to 0} S \cdot G(S) H(S)$
 $\therefore K_1 v = \lim_{S \to 0} S \cdot G(S) H(S)$
 $\therefore K_1 v = \lim_{S \to 0} \frac{S \cdot K_1(1+ST_2)(1+ST_{22})}{S^2(1+ST_{P1})(1+ST_{P2})} = \infty$
Steady stateleres (veloutly) is given by
 $\ell_{AS} (veloutly) = \frac{1}{K_1v} = \frac{1}{\infty} = 0$

given by.	hat, the Ace	eleration fro	or Constant ins
اد	a = lim s² G	(S) H (S)	
: 14a = (lm <u>s² 4 (1+</u> s→0 <u>s² (1+</u>	STZ,) (1+STZ2 STP1) (1+STP	$\frac{1}{2} = 14$
Steady state	Error (Accele	ration) l're gf	even by
CBB (AC	iceleration2) = .	$\frac{1}{14a} = \frac{1}{14a} = \frac{1}{14a}$	finite Value.
k→ Steady - sto are Summari:	sed in table	as shown t	selows.
are Summari:	sed is table	as shown tog as shown to dy = State Cr	Sero co.
K→ Steady - Sta are Summari: Type Of Input	sed in table Stea	dy = State Cr	Serow.
are Summari:	sed in table Stea	dy = State Cr	3 PELO 60.
Type Of Input	sed in table Stea Type-0 System	dy = State Cr	3 PELO 60.
Type Of Input Unit - Step	Stea Type-0 System <u>1</u> 1+4P	ax shown to dy = State Cr Type-1 System 0	Type-2 System 0

*→Generalised Error (a-Efficient Method (or Dynamic
error (a-Efficients) (Generalize Error Series)
→ The Error transfer function i's given by

$$\frac{E(s)}{R(s)} = \frac{1}{1+f_{1}(s)+f(s)} \longrightarrow 0$$
→ By Expanding Eqn (b) by Taylon's Series

$$\frac{1}{1+f_{1}(s)+f(s)} = C_{0} + C_{1}S + C_{2}S^{2} + C_{3}S^{3} + - + \cdots \rightarrow \odot$$
hlbere Co, Ci, C₂ - ... ke on are the generalised or
dynamic Error Constant.
→ By taking (imits on both sides as S=0

$$\frac{1}{s \rightarrow 0} \frac{1}{1+f_{1}(s)+f(s)} = C_{0}$$

$$\frac{1}{s \rightarrow 0} \frac{1}{1+f_{1}(s)+f(s)} = C_{1}$$

$$\frac{1}{s \rightarrow 0} \frac{1}{s + c_{2}S + 3C_{3}S^{2} + 4C_{4}S^{3} + - + \cdots \rightarrow \odot}$$

$$\frac{1}{s \rightarrow 0} \frac{1}{s + f_{1}(s)+f(s)} = C_{1}$$

$$\frac{1}{s \rightarrow 0} \frac{1}{s + f_{2}(s)+f(s)} = C_{1}$$

$$\frac{1}{s \rightarrow 0} \frac{1}{s + f_{2}(s)+f(s)} = C_{1}$$

$$\frac{1}{s \rightarrow 0} \frac{1}{s + f_{2}(s)+f(s)} = 2 \cdot 1 \cdot C_{2} + 3 \cdot 2 \cdot (s)^{2} + 4 \cdot 3 \cdot C_{4} \cdot s^{2} + - + \cdots \oplus$$

$$\frac{1}{s} \frac{1}{s \rightarrow 0} \frac{1}{s + 1} \frac{1}{(1 + f_{1}(s)+f(s))} = 2 \cdot 1 \cdot C_{2} + 3 \cdot 2 \cdot (s)^{2} + 4 \cdot 3 \cdot C_{4} \cdot s^{2} + - + \cdots \oplus$$

$$C_{2} = \frac{1}{2!} \lim_{S \to 0} \frac{d^{2}}{ds^{2}} \left(\frac{1}{1+\zeta(S)H(S)} \right)$$

In general
$$C_{n} = \frac{1}{n!} \lim_{S \to 0} \frac{d^{n}}{ds^{n}} \left(\frac{1}{1+\zeta(S)H(S)} \right)$$

By substituting Eqn (2) in (1)
$$\frac{E(S)}{R(S)} = C_{0} + C_{1}S + C_{2}S^{2} + C_{3}S^{3} + - + \cdots$$

E(S) = $C_{0}R(S) + C_{1}SR(S) + C_{2}S^{2}R(S) + - + \cdots$

By taking Inverse (aplace transform, we get the generalized Error periods

(c(t) = $C_{0}r(t) + C_{1}r(t) + C_{2}\frac{d^{2}}{dt^{2}}r(t) + - + \cdots$

SSE, $C_{AB} = \frac{lim}{t \to \infty} e(t)$

Things to remember: K→ Closed loop transfer function of a second Order Control System is given by, $\frac{C(s)}{R(s)} = \frac{\omega_n^2}{s^2 + 2\xi \omega_n s + \omega_n^2}$ * Where Wn = natural frequency of Oscillations E = Damping ratio Ewn = Damping factor Wd = WnJI-42 = Damping frequency of escillations. * > The repponse of a second Order control system Subjected to a unit step loput is given by, $c(t) = 1 - \frac{e}{1 - \epsilon^2} \sin(\omega_{at} + \theta)$ * -> Rise time , Tr is given by $T_{r} = \frac{\pi - \vartheta}{\omega_{d}} \qquad \text{But } \vartheta = \tan^{-1} \sqrt{1 - \xi^{2}}$ W2 = WD - VI- 52 $T_{g} = T - \tan^{-1} \sqrt{1 - \xi^{2}}$ ω_{d} * -> Pears time (Tp is given by $T_{p} = \frac{\pi}{\omega_{d}} = \frac{\pi}{\omega_{n} \sqrt{1-\xi^{2}}}; \quad for n = 2, t_{p} = \frac{2\pi}{\omega_{d}}$ $for n = 2, t_{p} = \frac{2\pi}{\omega_{d}}$

* > Peak Over shoot
$$(M_p)$$
 is given by

$$M_p = A \cdot e^{\frac{1}{5} T / \sqrt{1-\frac{5}{4} 2^{-5}}}$$
Where (A) is the strength of the input.
For Ex: If the System is subjuited to a step input
 g_{s} strength '2' unit then the Expression for $(M_p)^{-1}$
is given by

$$M_p = 2 \cdot e^{-\frac{5}{5} T / \sqrt{1-\frac{5}{4} 2^{-5}}}$$
* > % peak Overshoot i's given by

$$\int M_p = \frac{-\frac{5}{5} T / \sqrt{1-\frac{5}{4} 2^{-5}}}{\frac{1}{5} (M_p)^{-5} (M_p)^{-5} (M_p)^{-5}}$$
* > Settling time $(T_s)^{-5} (M_p)^{-5} (M_p)^{-5}$
Steady State Environ is given by

$$\frac{E(S)}{R_1(S)} = \frac{1}{1+G(S)+F(S)}$$
Steady State Environ (SSE) is given by

$$SSE = (M_p - (M_p) = \frac{M_p}{S \to 0} S \cdot E(S)$$

$$\therefore SSE = \lim_{S \to 0} = S \cdot \frac{R(S)}{1 + Q(S)H(S)}$$

$$\Rightarrow Error Constants$$

$$I \in Static Error Constants (Kp) is given by$$

$$I \in Cim Q(S)H(S); SSE = CAS = \frac{1}{1 + Kp}$$

$$I \notin R(S) = \frac{A}{S}$$

$$then, SSE = CAS = \frac{A}{1 + Kp}$$

$$I \notin R(S) = \frac{A}{S^2}$$

$$K_V = \lim_{S \to 0} S \cdot Q(S)H(S); SSE = CAS = \frac{1}{K_V}$$

$$I \notin R(S) = \frac{A}{S^2}$$

$$then, SSE = CAS = \frac{A}{K_V}$$

$$I \notin R(S) = \frac{A}{S^2}$$

$$then, SSE = CAS = \frac{A}{K_V}$$

$$(f Atceleration Error Constant (Ka) is given by$$

$$I \notin R(S) = \frac{A}{S^2}$$

$$then, SSE = CAS = \frac{A}{K_V}$$

$$I \notin R(S) = \frac{A}{S^2}$$

$$then, SSE = CAS = \frac{A}{K_V}$$

$$I \notin R(S) = \frac{A}{S^2}$$

$$then, SSE = CAS = \frac{A}{K_V}$$

$$I \notin R(S) = \frac{A}{S^2}$$

$$then, SSE = CAS = \frac{A}{K_V}$$

* -> Effects	of change in G	(s) H(s) on sta	ady state Error
$G(S)H(S) = \frac{H((1+ST_{z_1})(1+ST_{z_2})}{S'(1+ST_{P_1})(1+ST_{P_2})} \cdots$			
'n' is the type of the System.			
→ Type 'O' System; n=0; no integration; no poles of G(S)H(S) at the Origin of S-plane			
-> Type 'i' System; n=1; one integration; One poles of G(S)H(S) at the Origin of S-plane			
-> Type'z' System; n=2; two integrations, two police of G(S) H(S) at the Origin of S-plane			
* Table for Steady State Error for Vanious Popula and Systems			
Type 8¢ Input Steady State-			σr
gre of input	Typ-0 System	Type - 1 System	Type & System
Unit-step	1+ 4P	o	0
Unit-ramp	8	- <u> </u> Kiv	o
Unit - Parabolic	8	ø	- I Kia
	14p = lim G(s) H(s) S→0	14v=lim 5.G(s) H(s) s->0	4a=lims=g(s)H(s) s→0

* Generalized Proof Series:

$$C_{n} = \frac{1}{n!} \lim_{s \to 0} \frac{d^{n}}{ds^{n}} \left[\frac{1}{1+G(s)H(s)} \right]$$

$$e(t) = (_{0} \forall (t) + (_{1} \forall'(t) + (_{2} \forall''(t) + (_{3} \forall''(t) + -+ - - -)))$$

$$SSE = C_{bs} = \lim_{t \to \infty} e(t)$$

$$SSE = C_{bs} = \lim_{t \to \infty} e(t)$$

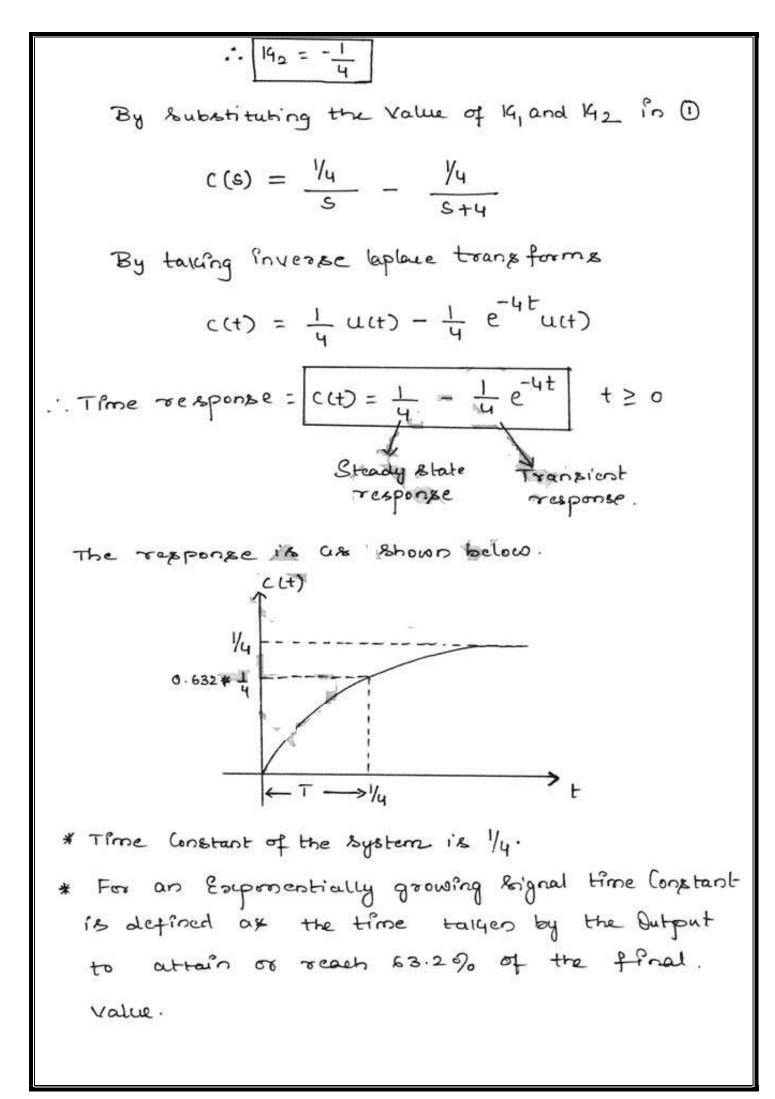
$$K_{ax} \text{ Maximum peak Rissponse}$$

$$Max \text{ Rissponse} = \mathcal{L}(t_{p}) = 1 + e^{-\frac{1}{2}}$$

$$i_{b} \quad c(t) = \text{unit step with strength A}$$

$$c \quad ctp = A \left(1 + e^{-\frac{1}{2}\pi/\sqrt{1-\frac{1}{2}}}\right)$$

* Problems in The Presence of Centrol System:
1) A first Order Control System is represented by
a transfer function
$$\frac{C(S)}{P_1(S)} = \frac{1}{S+4}$$
. Determine the
time Constant and response of the System for
a unit step Proput.
Solution: Given $\frac{C(S)}{P_1(S)} = \frac{1}{S+4}$
 $C(S) = \frac{1}{S+4}$. R(S)
But the given Proput PS unit step
i.e $C(S) = \frac{1}{S+4}$: $\frac{1}{S}$
 $C(S) = \frac{1}{S+4}$: $\frac{1}{S}$



Given C(t) for a unit ktep input
i.e r(t) = u(t) = 1 or
$$R(S) = \frac{1}{S}$$

 $\therefore C(S) = \frac{600}{(S+10)(S+60)} \cdot R(S)$
Cloped loop transfer functions is given by
 $\frac{C(S)}{R(S)} = \frac{600}{(S+10)(S+60)}$
or
 $\frac{C(S)}{R(S)} = \frac{600}{S^2 + 70S + 600}$
The above Equations is for the form of
 $\frac{C(S)}{R(S)} = \frac{100^2}{S^2 + 2500 + 00^2}$
By Comparing $S^2 + 70S + 600$ with $S^2 + 2500 + 00^2$
 $2500 = 70 + 00^2 = 600$
 $wn = \sqrt{600}$
 $wn = \sqrt{600}$
 $wn = \sqrt{600}$
 $8 = \frac{70}{2 \times 24 \cdot 49}$
 $8 = \frac{70}{2 \times 24 \cdot 49}$
 $8 = 1.429$

* For an lapponenhially decaying hyrat time constant
is defined as the time taken to attain 36.8% of
initial value.
(+)

$$A = (+)$$

 $C(+) = A e^{-4t}$
 $C(+) = A e^$

3) A negative feed books system has the following
transfer function:
$$q(s) = \frac{q}{s(s+2)}$$
, $H(s) = 1$.
Determine its natural frequency, damping outro,
damped frequency q Oscillation, damping feator, rike
time, $\frac{q}{2}$ peak over shoet, peak time and approximate
 5% Setting time. Resume A unit step foput.
Settion: $q(ver, q(s)) = \frac{q}{s(s+2)}$; $H(s) = 1$;
The Overall transfer function: i's given by
 $\frac{c(s)}{R(s)} = \frac{q}{s(s+2)}$: $\frac{q}{s^2+2s}$
 $\frac{(cs)}{R(s)} = \frac{q}{s^2+2s+q}$
Comparing $s^2 + 2s+q$ with $s^2 + 2gWs + Wn^2$
 $2gWs = 2 + Wn^2 = q$
Natural frequency; $Wn = \sqrt{q}$
 $\frac{g(wn = 1)}{Wn = 3}$ read/see
Damping factor; $gWn = \frac{1}{3}$
 $\frac{q}{q = \frac{1}{3}}$

Theat, three =
$$tp = \frac{11}{W_{d}}$$

 $tp = \frac{TT}{2.828}$
 $tp = 1.11$ See
Setting time for 5% tollerance.
 $ts = \frac{3}{9W_0}$
 $ts = \frac{3}{1}$ See:
 $ts = \frac{3}{1}$ See:
 $ts = \frac{3}{1}$ See:
 $ts = \frac{3}{1}$ See:
 $ts = 3$
44 A second Order Chetrol System (& represented
by $\theta_0 + 4\theta_0 + 25\theta_0 = 258e_0$ determine damping reatio,
natural frequency, damping factor, damping frequency
of Oscillations, peak three, over sheet for a unit step
input.
colution:
Gruen the differential Equation of the System
 $\frac{d^2\theta_0}{dt^2} + 4\frac{d\theta_0}{dt} + 25\theta_0 = 25\theta_c$
By tauling (aplace transforms, assuming sero
initial (anditions, we get
 $s^2\theta_0(s) + 4s\theta_0(s) + 25\theta_0(s) = 25\theta_c$ (s)
 $\theta_0(s) \{s^2 + 4s + 25\} = 25\theta_c$ (s)

Transfer function is given by

$$\frac{\partial_{0}(s)}{\partial_{c}(s)} = \frac{25}{s^{2}+4s+25}$$
Comparing $s^{2} + 4s+25$ with $s^{2}+2\xi W_{n}s + W_{n}^{2}$
 \Rightarrow Natural frequency (Wn)
 $W_{n}^{2} = 25$
 $W_{n} = \sqrt{25}$
 $W_{n} = \sqrt{25}$
 $W_{n} = 5$ rad/sec
 \Rightarrow Damping factors (ξWn)
 $2\xi W_{n} = 4$
 $\xi Wn = \frac{14}{2}$
 $\xi Wn = \frac{14}{2}$
 $\xi Wn = \frac{14}{2}$
 $\xi Wn = \frac{12}{2}$
 \Rightarrow Damping ratio (ξ_{1})
 $\xi Wn = 2$
 $\xi = \frac{2}{W_{n}} = -\frac{2}{5}$
 $\overline{\xi_{1}} = 0.4$
 \Rightarrow Damped frequency of Oscillations (Wd)
 $W_{d} = W_{n}\sqrt{1-\xi_{1}^{2}}$
 $W_{d} = 5\sqrt{1-(0.4)^{2}}$
 $W_{d} = 4.582$ rad/sec

→ Fease time (tp)

$$tp = \frac{11}{W_{d}}$$

$$= \frac{11}{W_{d}}$$

$$= \frac{11}{W_{d}}$$

$$= \frac{11}{W_{d}}$$

$$tp = 0.685$$
 Secs
> Max peaks Overschoot (Mp)
Mp = A e^{-Q} T/J1-Q²
Given A = 1 [unit step forpat]
Mp = e^{-0.4} T/J1-Q²
[Mp = 0.253]
55 A unity feedback system is characterised by
Opeer loop transfer function $Q(S) = \frac{14}{S(S+10)}$
determine the gain K so that the System
cofil have a damping ratio of 0.5 for this value QK.
Determine the settling time, Peaks Overschoot,
and rise time for a unit step input.
Solution:
 $G(S) = \frac{14}{S(S+10)} = \frac{14}{S^2+10.5}$
 $H(S) = 1$

The queral everytes function is given by

$$\frac{c(s)}{R(s)} = \frac{G(s)}{1+G(s)H(s)}$$

$$= \frac{\frac{K}{s^2+10s}}{1+\frac{K}{s^2+10s}} = \frac{\frac{14}{s^2+10s}}{\frac{s^2+10s+14}{s^2+10s}}$$

$$\frac{c(s)}{R(s)} = \frac{\frac{14}{16}}{\frac{s^2+10s+K}{s^2+10s+K}}$$
By Comparing $s^2 + 10s + K$ with $\delta^2 + 2\xi W_0 s + W_0^2$.
 $W_0^2 = K + 2\xi W_0 = 10$
Given $\xi = 0.5$
 $\therefore 2x0.5 W_0 = 10$
 $W_0 = 10$ rod/sec
 $\therefore [K_1 = W_0^2 = 100]$
 $\# \rightarrow Settleng time (tg)$
for 5% tollevance.
 $t_s = \frac{3}{\xi} W_0$
 $t_s = \frac{3}{6.5 \times 10} = \frac{3}{5}$
 $\frac{1}{\xi = 0.5}$ Seeps

$$t_{x} = \frac{\pi - \pi/3}{10\sqrt{1 - (0.9)^{-1}}}$$

$$t_{x} = \frac{\pi - 1.0471}{\pi \cdot 1.0471}$$

$$g. 56$$

$$t_{x} = 0.2418$$
Seep

65 For a Serve mechanism system $G(S) = \frac{161}{6^{2}}$

$$H(S) = 1.145_{2}S. Determine the Value of $K_{11} \cdot K_{12}$, so that the peak Over shoot to Unit step input is

0.25 and the peak time is a see.

Solution: Given $G(S) = \frac{K_{11}}{S^{2}}$

$$H(S) = 1.145_{10}S, Mp = 0.25, tp = 2.5865_{10}S$$

$$\frac{C(S)}{R(S)} = \frac{C(S)}{1 + G(S) + 1}$$

$$\frac{C(S)}{R(S)} = \frac{K_{11}}{1 + (\frac{K_{11}}{S^{2}})(1 + K_{2}S)}$$

$$\frac{C(S)}{R(S)} = \frac{K_{11}}{1 + (\frac{K_{11}}{S^{2}})(1 + K_{2}S)}$$

$$\frac{C(S)}{R(S)} = \frac{K_{11}}{S^{2} + K_{11}K_{12}S + K_{11}}$$
By Comparing $S^{2} + K_{11}K_{12}S + K_{11}$ with $S^{2} + 25$ Wors + Wn²

$$\frac{[K_{11} = Wn^{2}]}{G} + \frac{[K_{11}K_{12} = 25Wn]}{G}$$$$

$$Mp = A \cdot e^{-\frac{q_{1}}{1}/1 - \frac{q_{2}}{2}} \text{ for a unit step input } A = 1$$

$$Mp = 1 \cdot e^{-\frac{q_{1}}{1}/1 - \frac{q_{2}}{2}}$$
By taking natural log on Both knide.
$$\ln \{Mp\} = \ln \{e^{-\frac{q_{1}}{1}/1 - \frac{q_{2}}{2}}\}$$

$$\ln \{0.25\} = -\frac{q_{11}}{\sqrt{1 - q_{2}^{2}}} \ln \{e\} \qquad \because \ln[x^{V}] = y \ln[x]$$

$$-1.3862 = -\frac{q_{11}}{\sqrt{1 - q_{2}^{2}}}$$

$$Taking square On both snides$$

$$(-1.3862)^{2} = (-\frac{q_{11}}{2})^{2}$$

$$(-1.3862)^{2} (1 - \frac{q_{1}}{2}) = \frac{q^{2}}{2} q \cdot 8696$$

$$1.9215 = 1.9215 q^{2} = q \cdot 8696 q^{2}$$

$$1.9215 = 11.7411 q^{2}$$

$$q^{2} = 0.1629$$

$$q = \sqrt{0.1629}$$

$$q = \sqrt{0.1629}$$

Peak time ;
$$tp = \frac{\pi}{Wd}$$
.
 $Wd = Wn \sqrt{1-g^2}$.
 $tp = \frac{\pi}{Wn \sqrt{1-g^2}}$.
 $Wn = \frac{\pi}{tp \sqrt{1-g^2}}$.
 $W_n = \frac{\pi}{2 \cdot \sqrt{1-(0.4036)^2}}$.
 $W_n = 1 \cdot 7168 \operatorname{rod} / 8ex$.
 $U_1 = Wn^2 = 1 \cdot 7168^{2-1}$.
 $U_{1} = 2.9474$.
 $U_{1} = 2.9474$.
 $U_{1} = 2 \cdot 9474$.
 $U_{1} = 2 \cdot 9474$.
 $U_{2} = 2 \cdot (0.4036) \cdot (1.7168)$.
 $U_{1} = 2 \cdot (0.4036) \cdot (1.7168)$.
 $U_{2} = 2 \cdot (0.4036) \cdot (1.7168)$.
 $U_{3} = 2 \cdot (0.4036) \cdot (1.7168)$.
 $U_{4} = 0.4036$.
 U_{4}

#9 A step of two is applied to the Unity teer source
system shown Pr the figure. determine the
Value of A415 Such that damping rahio is 0.6
and damped free quency of Oscillation is stad/see
what Ps the peak value of response.
R(S)
$$\rightarrow A$$
 $\stackrel{1}{\xrightarrow{5}}$ $\stackrel{1}{\xrightarrow{5$

$$A = Wn^{2} ; \quad 14 = 2 \xi Wn$$

Grven $\xi = 0.6$

$$W_{d} = 8 \text{ rad/see}$$

$$W_{d} = Wn \sqrt{1 - \xi^{2}}$$

$$Wn = \frac{Wd}{\sqrt{1 - \xi^{2}}}$$

$$W_{n} = \frac{8}{\sqrt{1-0.6^{2}}}$$

$$W_{n} = 10$$

$$\therefore A = W_{n}^{2} = 10^{2}$$

$$\boxed{A = 100}$$

$$W_{4} = 2\xi W_{n} = 2 \neq 0.6 \neq 10$$

$$\boxed{K_{4} = 12}$$

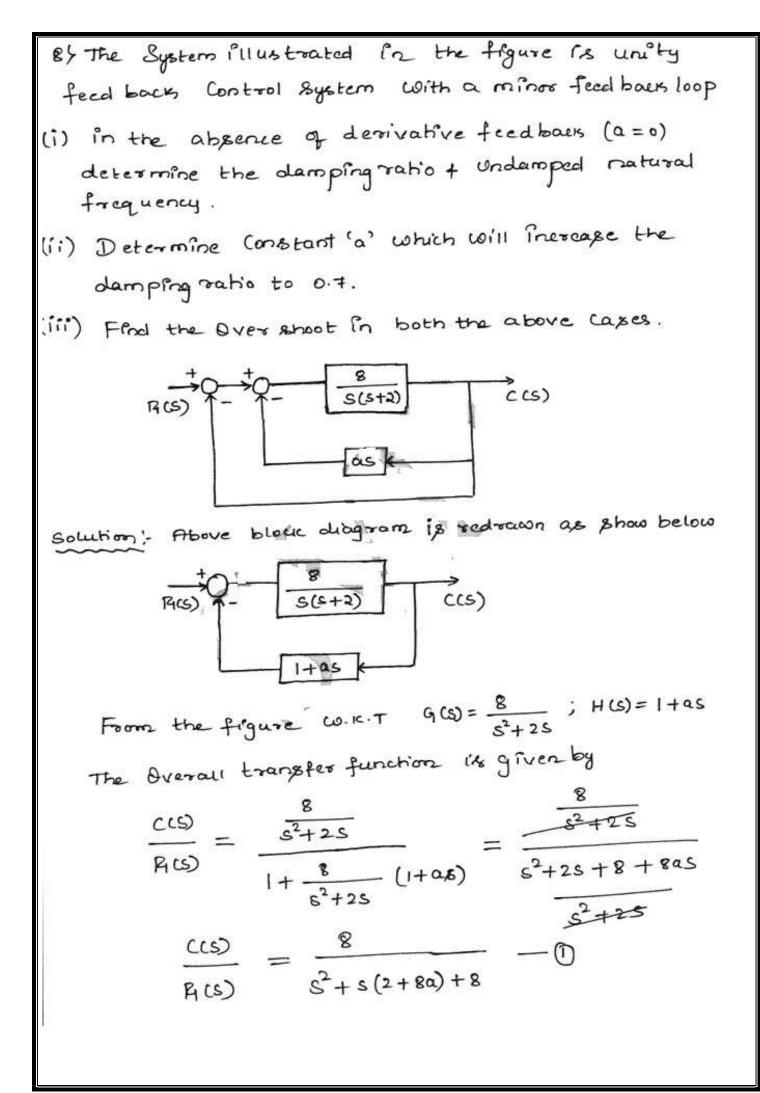
$$* \Rightarrow Peak \quad Value \ a_{1} \ the \ scappon 8e.$$

$$C_{max}(t) = c(tp) = A_{1}(1 + e^{\frac{1}{2}})$$

$$A_{1} \ Phi \ the \ strength \ a_{1} \ the \ Unit \ step \ input$$

$$C_{max}(t) = c(tp) = a \left[1 + e^{-0.6 \pi / \sqrt{1-0.6^{2}}}\right]$$

$$\boxed{C_{max}(t)} = c(tp) = a \left[1 + e^{-0.6 \pi / \sqrt{1-0.6^{2}}}\right]$$



(ase (i) when a=0 from Equation (1) $\frac{C(S)}{R(S)} = \frac{8}{S^2 + 2S + 8}$ By Comparing S2+25+8 With S2+24WnS+Wn2 $W_{0}^{2} = 8 + 2 \xi W_{0} = 2$ natural frequency:-Wn=J8 = 252 = 2.828 rad/see Damping factor (Ewn) 2 GWN = 2 & Wn = 2 & Wn=1 Damping ratio (4) EWN=1 $\xi = \frac{1}{blo}$ $\xi_1 = \frac{1}{2.828}$ E = 0.3538 case (ii) when a = 0 given & = 0.7 <u>(2)</u> = <u>8</u> from 1 s2+s(2+8a)+8 R(S) By comparing s2+s(2+8a)+8 with s2+26Wns+Wn2

$$\begin{split} & U_{0}^{2} = \$ \ ; \ \forall n = \sqrt{\$} = \sqrt{2} 2^{-1} \\ & \overline{|\psi_{n}|^{2} - 2 \cdot \$ 2\$} \ \text{voal}/\text{kec} \\ & 2 \xi \forall n = 2 + \$ 2 \\ & \$a = 2 \xi \forall n - \$ \\ & a = 2 \xi \forall n - \$ \\ & a = 2 \xi \forall n - \$ \\ & a = 2 \underline{\times} 0.7 \pm 2 \cdot \underline{\$ 2\$ 2\$ - 2} \\ & \underline{|v_{0}|^{2}} \\$$

94 The Open loop transfer function of a Unity feedbook
Control System is given by
$$G(S) = \frac{14}{S(1+ST)}$$
. Determine
(i) By what factor should the Unoplifies gain 'K' be
reduced in Order that the damping ratio is
increased from 0.2 to 0.8
(ii) By What factor should 'K' be routiplied, so
that the System Overshoot for unit step
Excitation is reduced from 50% to 20%
Solution: Given $G(S) = \frac{14}{S(1+ST)}$ $H(S) = 1$
The Overall transfer function (& given by
 $\frac{C(S)}{R(S)} = \frac{G(S)}{1+G(S)+H(S)}$
 $\frac{C(S)}{R(S)} = \frac{14}{(1+\frac{14}{S+S^2T}-1)} = \frac{\frac{14}{S+S^2T}}{\frac{S+S^2T+14}{S+S^2T}}$
 $\frac{C(S)}{R(S)} = \frac{14}{(1+\frac{14}{S+S^2T}+1)} = \frac{\frac{14}{S+S^2T}}{\frac{S+S^2T+14}{S+S^2T}}$
By dividing Both numerators and denominator by T
 $\frac{C(S)}{R(S)} = \frac{14/T}{\frac{5}{T}+\frac{5^2T}{T}+\frac{14}{T}} = \frac{14/T}{S^2+\frac{14}{T}S+\frac{14}{T}}$

Comparing
$$s^{2} + \frac{1}{T} s + \frac{K}{T}$$
 with $s^{2} + 2s W_{0} s + W_{0}^{2}$
 $W_{0}^{2} = \frac{K}{T}$ $\sigma_{x} = W_{0} = \sqrt{\frac{K}{T}}$
 $2 g W_{0} = \frac{1}{T}$
 $g_{1} = \frac{1}{2 W_{0}T} = \frac{1}{2T\sqrt{\frac{K}{T}}}$
Note $\sqrt{T} * \sqrt{T} = T$
 $g_{1} = \frac{1}{2 \sqrt{K}} \sqrt{T} \cdot \sqrt{\frac{K}{T}}$
Note $\sqrt{T} * \sqrt{T} = T$
 $g_{2} = \frac{1}{2 \sqrt{K}} \sqrt{T} \cdot \sqrt{\frac{K}{T}}$
Note $\sqrt{T} * \sqrt{T} = T$
 $g_{1} = \frac{1}{2 \sqrt{K}} \sqrt{T} \cdot \sqrt{\frac{K}{T}}$
Note $\sqrt{T} * \sqrt{T} = T$
 $g_{1} = \frac{1}{2 \sqrt{K}} \sqrt{T} \cdot \sqrt{\frac{K}{T}}$
Note $\sqrt{T} * \sqrt{T} = T$
 $f_{1} = \frac{1}{2 \sqrt{K}} \sqrt{T} \cdot \sqrt{\frac{K}{T}}$
Note $\sqrt{T} * \sqrt{T} = T$
 $f_{2} = \frac{1}{2 \sqrt{K}} \sqrt{T} \cdot \sqrt{\frac{K}{T}}$
Note $\sqrt{T} * \sqrt{T} = T$
 $f_{1} = \frac{1}{2 \sqrt{K}} \sqrt{T} \cdot \sqrt{\frac{K}{T}}$
Note $\sqrt{T} * \sqrt{T} = T$
 $f_{1} = \frac{1}{2 \sqrt{K}} \sqrt{T} \cdot \sqrt{\frac{K}{T}}$
 $\sqrt{T} * \sqrt{\frac{K}{T}} = \sqrt{\frac{K}{T}}$
 $\frac{K_{1}}{\frac{1}{2 \sqrt{K_{1}T}}}$
 $\frac{K_{1}}{\frac{1}{2 \sqrt{K_{1}}}} = \frac{1}{2 \sqrt{\frac{K}{K_{1}}}}$
 $\frac{K_{1}}{\frac{1}{2 \sqrt{K_{1}}}} = \frac{1}{2 \sqrt{\frac{K}{K_{1}}}}$
 $\frac{K_{1}}{\frac{1}{2 \sqrt{K_{1}}}} = \sqrt{\frac{K_{2}}{K_{1}}}$

$$\frac{|k_{12}|}{|k_{11}|} = \left(\frac{k_{11}}{k_{22}}\right)^{2} \longrightarrow 0$$

$$\frac{|k_{12}|}{|k_{11}|} = \left(\frac{0 \cdot 2}{0 \cdot 8}\right)^{2}$$

$$\frac{|k_{12}|}{|k_{11}|} = \frac{1}{16} = 0.0625$$

$$\frac{|k_{12}|}{|k_{11}|} = \frac{1}{16} = 0.0625$$

$$\frac{|k_{12}|}{|k_{12}|} = 0.0625 |k_{11}|$$
Conclusion: The gain has to be reduced by 1-0.0625

$$\frac{|k_{12}|}{(1-0.0625) * 100} = 93.75\% \qquad \boxed{Note H(S) = 1}$$
The gain has to be reduced by 93.75% to improve
the damping ratio 0.2 to 0.2
Case(ii) : Let K_{1} k_{11} and k_{1} = k_{1} when
Mp = Mp_{1} = 0.6 and

$$|k_{1} = k_{12} and k_{1} = \frac{k_{12}}{2} when
Mp = Mp_{2} = 0.2
\Rightarrow For a unit step input
Mp = e^{-kT/\sqrt{1-k_{1}^{2}}}$$
By taking natural leg on Both hidge

$$Ln \{Mp \} = Ln \{e^{-kT/\sqrt{1-k_{1}^{2}}} Ln \{e\}$$

$$\ln Mp = -\frac{6\pi}{\sqrt{1-6^2}}$$

$$(\ln Mp) \cdot (\sqrt{1-6^2}) = -6\pi$$

$$By taking Square's Da Beth Sides
$$(\ln Mp)^2 (1-6^2) = 6^2\pi^2$$

$$(\ln Mp)^2 - (\ln Mp)^2 \delta^2 = 6^2\pi^2 + (\ln Mp)^2 \delta^2$$

$$(\ln Mp)^2 = 6^2\pi^2 + (\ln Mp)^2 \delta^2$$

$$(\ln Mp)^2 = \delta^2 (\pi^2 + (\ln Mp)^2)$$

$$\delta^2 = ((\ln Mp)^2 - \pi^2 + (\ln Mp)^2)$$

$$\delta^2 = ((\ln Mp)^2 - \pi^2 + (\ln Mp)^2)$$

$$\delta_1 = \sqrt{\frac{(\ln Mp)^2}{\pi^2 + (\ln Mp)^2}} = \delta_1 = \sqrt{\frac{(\ln Mp)^2}{\pi^2 + (\ln Mp)^2}}$$

$$\delta_1 = \sqrt{\frac{(\ln Mp)^2}{\pi^2 + (-0.5108)^2}}$$

$$\delta_2 = \sqrt{\frac{((\ln Mp)^2}{\pi^2 + (\ln Mp)^2}}$$

$$\delta_2 = \sqrt{\frac{((\ln 0.2)^2}{\pi^2 + (\ln 0.2)^2}}$$$$

$$F_{2} = \sqrt{\frac{(-1.6094)^{2}}{\Pi^{2} + (-1.6094)^{2}}}$$

$$\overline{F}_{2} = 0.455$$
By Substituting the Value & 6, and 6, in Equation ()

$$\frac{|K_{2}|}{|K_{1}|} = \left(\frac{0.1604}{0.455}\right)^{2}$$

$$\frac{|K_{2}|}{|K_{1}|} = 0.1237$$

$$K_{2} = 0.1237$$

$$K_{2} = 0.1287 K_{1}$$
* \rightarrow Eastial gain has to multiplied by 0.1237 to reduced
the Over sheet from 60% to 20%
105 The step response of second Order Under
damped unity feed backs system shows in figure
for a input of 2011) determine the Open loop
and closed loop transfer functions of the
system.

$$\int_{0}^{C(+)} \frac{1}{2} \int_{0}^{1} \int_{0}^{2} \int_{0}^{1} \int_{0}^{2} \int_{0}^{1} \int_{0}^{2} \int_{0}^{1} \int_{0}^{2} \int_{0}^{1} \int_{0}^{2} \int_{0}^{1} \int_{$$

Solution:
From the fig :- We linew that

$$Mp = 0.5, tp = 2 \text{ Secs}, input = 2 \text{ urt})$$
To find:
$$\frac{CCS}{R(S)} = \frac{Wn^2}{S^2 + 2\xiWn5 + Wn^2} \rightarrow 0$$
A

$$Mp = 0.5, tp = 2 \text{ Secs}, input = 2 \text{ urt})$$
Given $A = 2$

$$0.5 = 2 \cdot e^{-\xi \Pi / \sqrt{1 - \xi^2}}$$
(Strength of
the input)
Given $A = 2$

$$0.5 = 2 \cdot e^{-\xi \Pi / \sqrt{1 - \xi^2}}$$
By talking natural log one both sides

$$ln\left(\frac{0.5}{2}\right) = ln\left(e^{-\xi \Pi / \sqrt{1 - \xi^2}}\right)$$

$$(-1.3862) = -\frac{\xi \Pi}{\sqrt{1 - \xi^2}} \quad (n(e))$$

$$\left(-1.3862\right) \sqrt{1 - \xi^2} = -\xi \Pi + 1$$
By talking Squares one both Endes

$$(-1.3862) \sqrt{1 - \xi^2} = -\xi \Pi + 1$$
By talking Squares one both Endes

$$(-1.3862)^2 (\sqrt{1 - \xi^2})^2 = (-\xi \Pi)^2$$

$$(-1.3862)^2 (-(-1.3862)^2\xi^2) = \xi^2 \Pi^2$$

$$(-1.3862)^2 - (-1.3862)^2\xi^2 = \xi^2 \Pi^2$$

$$(-1.3862)^2 = 10 \cdot 411 \xi^2$$

$$(-1.2862)^2 = 10 \cdot 411 \xi^2$$

$$(-1.2862)^2 = \xi^2$$

$$\begin{aligned} \xi^{2} &= \frac{1.4215}{11.7411} \\ \xi^{2} &= 0.16296 \\ \xi &= 0.16296 \\ \hline \xi &= 0.40368 \end{aligned}$$

$$\begin{aligned} \hline \text{Peake time, } tp &= \frac{11}{Wal} \\ tp &= \frac{11}{Wal} \\ tp &= \frac{11}{Wal} \\ \hline tp &= \frac{11}{Wal} \\ 2 &= \frac{11}{Wal} \\ \hline tp &= \frac{11}{Wal} \\ 2 &=$$

$$\frac{C(5)}{R(5)} = \frac{(1.717)^2}{S^2 + (2.80.4037 + 1.717) S + (1.717)^2}$$

$$\frac{C(5)}{R(5)} = \frac{2.948}{S^2 + 1.3863S + 2.948} \quad \text{cloaced loop transfer function}$$

$$\frac{C(5)}{R(5)} = \frac{2.948}{S^2 + 1.3863S} \begin{bmatrix} 1 + \frac{2.948}{S^2 + 1.3863S} \end{bmatrix}$$

$$\frac{C(5)}{R(5)} = \frac{2.948}{S(5+13863)} = \frac{4(5)}{1+6(5) + 1(5)}$$

$$\frac{C(5)}{R(5)} = \frac{\frac{2.948}{S(5+13863)}}{1+\frac{2.948}{S(5+13863)}} = \frac{6(5)}{1+6(5) + 1(5)}$$
Open loop transfer function = $\frac{6(5)}{S^2 + 1.38635}$

11) The closed loop poles of a System are at -2+js
and -2-j3, Compute the Value of demping ratio and
the damped frequency of Oscillation of System and
What is the % Overshoot of the System for a unit
Step input
Solution: The Overall transfer function of a Becond Order
Control System is given by
$$\frac{c(s)}{R(s)} = \frac{Wn^2}{s^2+26Wn5+Wn^2}$$
The poles of a System is given by
$$s = -26Wn \pm \sqrt{46^2Wn^2 - 44Mn^2}$$
Note: $ax^2 + bx - C = 0$
$$X = \frac{-b \pm \sqrt{b^2 - 44Mn^2}}{2a}$$
$$S = -\frac{26Wn \pm \sqrt{400^2(6^2 - 1)}}{2}$$
$$S = -26Wn \pm 2Wn \sqrt{6^2 - 1}$$
$$= -6Wn \pm Wn \sqrt{6^2 - 1}$$
$$= -6Wn \pm Wn \sqrt{6^2 - 1}$$
$$= -6Wn \pm Wn \sqrt{6^2 - 1}$$
$$S = -6Wn \pm Wn \sqrt{6^2 - 1}$$
$$S = -6Wn \pm Wn \sqrt{1 - 6^2}$$
$$S = -6Wn \pm Wn \sqrt{1 - 6^2}$$

$$gfuen : S = -2 \pm j3$$

$$\therefore \xi W_0 = 2 + W_d = 3$$
Damping foctors = $\xi W_0 = 2$
Damping frequency of Oscillations = $[U_d = 3] \text{ rod/sec}$

$$W_d = W_n \sqrt{1 - \xi^2}$$
By taking sequares on Both sodes
$$W_d^2 = (W_0 \sqrt{1 - \xi^2})^2$$

$$W_d^2 = W_0^2 (\sqrt{1 - \xi^2})^2$$

$$W_d^2 = W_0^2 (1 - \xi^2)$$

$$W_d^2 = W_0^2 - (\xi W_0)^2$$

$$W_0^2 = 4M^2 + (\xi W_0)^2$$

$$W_0^2 = 9 + 4 = 13$$

$$W_0 = \sqrt{13}$$

$$W_0 = 2$$

$$\xi = \frac{2}{W_0}$$

$$\begin{aligned} &\xi = \frac{2}{3.6055} \\ &\xi = 0.5547 \end{aligned}$$

Peak Over shoot for a Unit step input
$$A = 1$$

$$Mp = 1 * e^{-\xi T / \sqrt{1 - \xi^2}}$$

$$Mp = e^{-0.5547 T / \sqrt{1 - (0.5547)^2}}$$

$$Mp = 0.1236$$

$$Mp = 0.1236$$

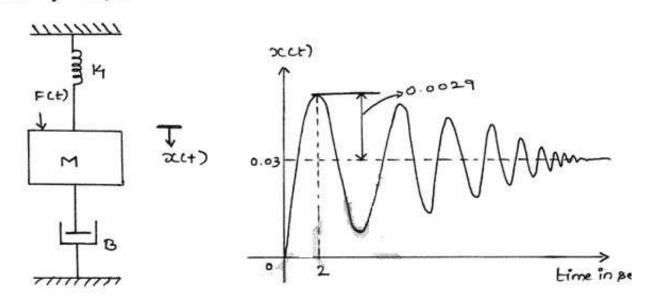
$$Np = e^{-\xi T / \sqrt{1 - \xi^2}} * 100\%$$

$$20 Mp = e^{-\xi T / \sqrt{1 - \xi^2}} * 100\%$$

$$20 Mp = 12.36\%$$

į.

force of 8.9 newtons is applied to the system the mass oscillates as shown in figures. Determine the Values of M, B and K



Solution:

The Equilibrium Equation the Mechanical System shows above, is given by

$$F(t) = M \frac{d^2 x^{(t)}}{dt^2} + K x^{(t)} + B \frac{d x^{(t)}}{dt}$$

By taking laplace transforms assuming zero initial Condition

$$F(s) = Ms^{2}x(s) + Hx(s) + BS x(s)$$

= (Ms^{2} + BS + H) x(s)
$$\frac{x(s)}{F(s)} = \frac{1}{Ms^{2} + BS + H} - 0$$

By dividing 'M' for both, Numerator and denominator

$$\frac{x cs}{Fcs} = \frac{\frac{1}{M}}{\frac{Ms^2}{M} + \frac{B}{M}s + \frac{H}{M}} = \frac{\frac{1}{M}}{\frac{s^2 + \frac{B}{M}s + \frac{H}{M}}{\frac{s^2 + \frac{H}{M}s + \frac{H}{M}}{\frac{s^2 + \frac{H}{M$$

By comparing
$$s^{2} + \frac{B}{M} s + \frac{H}{M}$$
 with $s^{2} + 2\xi H_{0}s + H_{0}^{2}$ we get

$$\frac{W_{0}^{2} = \frac{H}{M}}{W} - \textcircled{3}$$

$$\frac{2\xi H_{0} = \frac{B}{M}}{M} - \textcircled{3}$$

$$\frac{B}{M} = \frac{4}{M} e^{-\xi \Pi / \sqrt{1 - \xi^{2}}}$$

$$\frac{0.0029}{0.032} = 0.03 \cdot e^{\xi \Pi / \sqrt{1 - \xi^{2}}}$$

$$\frac{0.0029}{0.032} = e^{-\xi \Pi / \sqrt{1 - \xi^{2}}}$$

$$\frac{0.0029}{0.032} = e^{-\xi \Pi / \sqrt{1 - \xi^{2}}}$$

$$\frac{1}{M} \left[\frac{0.0029}{0.032} \right] = \ln \left[\frac{-\xi \Pi / \sqrt{1 - \xi^{2}}}{\sqrt{1 - \xi^{2}}} \right]$$

$$-2.3365 = \frac{-\xi \Pi}{\sqrt{1 - \xi^{2}}} \ln \left[\frac{C}{2} \right]$$

$$(-2.3365)(\sqrt{1 - \xi^{2}}) = -\xi \Pi - 1$$
By taking squares on both sides
$$(-2.3365)'(\sqrt{1 - \xi^{2}}) = (-\xi \Pi)^{2}$$

$$5.45923 - 5.45923 \xi^{2} = \xi^{2} - 9.8696$$

$$5.45923 = \xi^{2} - 883\xi^{2}$$

$$0d$$

$$\frac{5 \cdot 45923}{15.32883} = 8^{2}$$

$$\overline{S} = \sqrt{\frac{5 \cdot 45923}{15.32883}}$$

$$\overline{S} = \sqrt{\frac{5 \cdot 45923}{15.32883}}$$

$$\overline{S} = \sqrt{\frac{5 \cdot 45923}{15.32883}}$$

$$\overline{S} = \frac{\sqrt{5} \cdot 59677}$$

$$\overline{S} = \frac{1}{Nn\sqrt{1-\frac{6}{3}}}$$

$$H_{0} = \frac{TT}{Nn\sqrt{1-\frac{6}{3}}}$$

$$H_{0} = \frac{TT}{\frac{1}{\sqrt{1-\frac{6}{3}}}}$$

$$H_{0} = \frac{TT}{\frac{1}{\sqrt{1-\frac{6}{3}}}}$$

$$H_{0} = \frac{TT}{\frac{1}{\sqrt{1-(0.59(77))^{2}}}}$$

$$\overline{W_{0} = 1.4575} \text{ rad/sec}$$
From the Equation (1)

$$X(S) = \frac{1}{Ms^{2} + BS + 15} \cdot F(S)$$

$$G(uen F(S) = \frac{8 \cdot 9}{S} + \frac{Note:}{S} F(t) = 8 \cdot 9N \therefore F(S) = \frac{8 \cdot 9}{S}$$

$$and x(tr) a \neq t \rightarrow \infty = 0.0372 \text{ (from fry)}$$
From the final Value theorem $X(\infty) = \frac{1}{5 \rightarrow 0} \text{ S} X(S)$

$$0.03 = \lim_{s \to 0} \frac{5}{5} \cdot \frac{1}{Ms^{2} + Bs + 1K} \left(\frac{8 \cdot q}{s}\right)$$

$$0.03 = \frac{1}{K} (8 \cdot q)$$

$$0.03 = \frac{8 \cdot q}{K}$$

$$K = \frac{8 \cdot q}{6 \cdot 03}$$

$$K = \frac{8 \cdot q}{0 \cdot 03}$$

$$K = \frac{2 \cdot q}{6 \cdot 67} \frac{N/M}{M}$$
From (2)
$$M = \frac{14}{M0^{2}}$$

$$M = \frac{2 \cdot 67}{(1 \cdot q 5 + 5)^{2}}$$

$$M = \frac{74 \cdot 42}{M} N - Sec^{2}/M$$

$$2\xi w_n = \frac{B}{M}$$

 $B = 2\xi w_n M$
 $B = 2 * 0.5967 * 1.9575 * 77.42$

÷,

13) The block diagram of a unity feedback Control System
i's show [In figure. Determine the Characteristic
Equation of the System, Peak time, Peak Overshoot.
Time at which the first Undershoot occurs the time
Period of Oscillations and Number of Cycle Completed
before reacting the stoady state.

$$\frac{20}{(S+1)(S+5)}$$
CCS)
Solution:
Given System is unity feed back System

$$g(S) = \frac{20}{(S+1)(S+5)}$$
H(S)=1
The Overall transfer function is given by

$$\frac{20}{(S+1)(S+5)} = \frac{20}{S^2+68+55}$$

$$\frac{(CS)}{R(S)} = \frac{20}{(S+1)(S+5)^{-1}} = \frac{20}{S^2+68+55+20}$$

$$\frac{C(S)}{S^2+48+55} = \frac{20}{S^2+68+55}$$
By Comparing S²+68+25 with S²+28wnstwn²
wh²=25, 28wn=6
whn = 5

$$\Rightarrow 2\xi Wn = 6$$

$$\xi Wn = \frac{6}{2}$$

$$\frac{\xi}{4Wn = 3}$$

$$\xi = \frac{3}{4}$$

$$\frac{\xi}{5} = \frac{3}{5}$$

$$\frac{\xi = 0.6}{5}$$

$$\frac{\xi}{5} = 0$$

$$\frac{\xi}{5}$$

14% Conpider Unity feed back Cootrol Rystern whose Open loop
transfer function is given by
$$G(s) = \frac{\partial u_{s+1}}{\sigma(s+\delta_s)}$$
, Obtain the.
response if unit step roput for the same, alculate
rise time, maximum peaks Overshoot, peak time and
Setting time.
Solution: $G(ven G(s) = \frac{\partial u_{s+1}}{s(s+\delta_s)}$; $H(s) = 1$; \cdots unity feedback
System.
The Overall transfer function is given by
 $\frac{C(s)}{R(s)} = \frac{G(s)}{1+G(s)H(s)} = \frac{\frac{\partial u_{s+1}}{s^2+\delta(s)}}{1+\frac{\delta(s+1)}{s^2+\delta(s)} - 1} = \frac{\frac{\partial u_{s+1}}{s^2+\delta(s)}}{\frac{s^2+\delta(s+1)}{s^2+\delta(s)} - 1}$
The Output $C(s)$ is given by;
 $C(s) = \frac{\partial u_{s+1}}{s^2+s+1}$. $R(s)$
Given $R(s)$ is a unit step reput $\therefore R(s) = \frac{1}{s}$
 $C(s) = \frac{\partial u_{s+1}}{s^2+s+1} + \frac{1}{s}$
By Breakting RHS Re to partial foaction.
 $C(s) = \frac{\partial u_{s+1}}{s^2+s+1} + \frac{K_{s}S + K_{s}}{s^2+s+1} \rightarrow 0$
 $0 \cdot u_{s+1} = K_{1}(s^2+s+1) + (K_{s}S + K_{s}) S \rightarrow 3$

To find
$$K_1$$

Put s=0 in Equation -3
 $0+1 = K_1(0+0+1) + (K_2 \cdot 0 + K_3)0$
 $1 = K_1$
 $[K_1 = 1]$
Toffind K_2 and K_3
By comparing the Co-Stocients of S² in Equation 3
 S^2 ; $0 = K_1 + K_2$.
 $[K_2 = -K_1]$
 $[K_2 = -K_1]$
 $[K_2 = -K_1]$
By comparing the Selfbruchts of S Pr Equation 3
S; $0.4 = K_1 + K_3$
 $[K_3 = 0.4 = 161]$
By bubshituting the Value of K_1, K_2 and K_3 in 3
 $\therefore C(S) = \frac{1}{S} + \frac{-S - 0.L}{S^2 + S + 1}$
The above Equations can be reconsisten as
 $((S) = \frac{1}{S} - \frac{S + 0.6}{S^2 + 2 \cdot \frac{1}{2}S + (\frac{1}{2})^2 - (\frac{1}{2})^2 + 1}$
 $= \frac{1}{S} - \frac{S + 0.6}{(6+05)^2 + \frac{3}{4}}$

$$= \frac{1}{5} - \frac{(5+0.5)^{2} + (\frac{35}{2})^{2}}{(5+0.5)^{2} + (0.866)^{2}} - \frac{0.1}{0.866} + \frac{0.866}{(5+0.5)^{2} + (0.866)^{2}}$$
By taking Inverse laplace transforms

Time response:
$$\frac{(1+2)^{-0.5t}}{(1+2)^{-0.5t}} = \frac{0.1}{0.866} + \frac{0.866}{(5+0.5)^{2} + (0.866)^{2}}$$
By taking Inverse laplace transforms

Time response:
$$\frac{(1+2)^{-0.5t}}{(1+2)^{-0.5t}} = \frac{0.115}{0.115} = \frac{0.5t}{0.115} = \frac{0.5t}{0.115}$$
By Multiplying and dividing Equation (i) by $\sqrt{1^{2}+0.15^{2}}$

 $\therefore cct) = 1 - e^{-0.5t} \sqrt{11+0.115^{2}} = \frac{1}{\sqrt{11+0.115^{2}}} (0.8(0.866t) + \frac{0.116}{\sqrt{11+0.115^{2}}} = \frac{1}{\sqrt{11+0.115^{2}}} = \frac{1}{\sqrt{11+0.15^{2}}} = \frac{1}{\sqrt{11+0.15^{2}}} = \frac{$

1.006 e sin (0.866 to+ 8) = 0 $\sin\left(0.866 t_{r} + \theta\right) = 0$ 0.866t + + = OTT For the Ist time, n=1 $t_{\tau} = \frac{\overline{11} - \theta}{0.866}$ to = II- 83.43 * IL 0.866 : to = 1.9462 Sec When the repponse is masuroun at t=tp, $\frac{d c(t)}{dt} = 0.$ By Ditterestiating Equation () W.r.t 't' $\frac{dc(t)}{dt} = 0 - 1.006 \left[-0.5 e^{0.5t} \sin(0.866t + 8) + c^{0.5t} \cos(0.866t + 8) + c^{0.5t} + c^{0.5t} \cos(0.866t + 8) + c^{0.5t} \sin(0.866t + 8) + c^{0.5t} \cos(0.866t + 8) + c^{0.5t} \sin(0.866t + 8) +$ Cos (0.866+ + 0) $1.006 \ e^{-0.5t} \left[0.5 \sin \left(0.866t + \theta \right) - 0.866 \cos \left(0.866t + \theta \right) \right] = 0$ $0.55in(0.866t+0) = 0.866 \cos(0.866t+0) = 0$ $tan(0.866t+0) = \frac{0.866}{0.5} = 1.732$ $\tan(0.865t+8) = 1.732$ $0.866t + \theta = tan^{-1}(1.732)$ 0.866t + 0 = 60 + 180 : to avoid Sign add 180

0.866 t = 180+60 - 83.43°
0.866 t = 156.57°
t =
$$\frac{156.57 + \frac{11}{180}}{0.866}$$

Plans time; t = 3.155 seex
By Substituting t = tp in Equation @ we get the
maximum response.
Teaus response, Cmax(t) = C(tp)
 $C(tp) = 1 - 1.006 e^{-0.5 \times 3.155}$
 $C(tp) = 1 - 1.006 e^{-0.5 \times 3.155}$
 $C(tp) = 1 - 1.006 e^{-0.5 \times 3.155}$
 $C(tp) = 1.1749$
The Envelope of the time response is given by
 $= 1 \pm 1.005 e^{-0.51}$ at the settling time ts,
for 5% to levance, Envelope of the Wave form is
 $for 5\%$ to levance, Envelope of the Wave form is
 $for 5\%$ to levance, Envelope of the Wave form is
 $for 5\%$ to levance, Envelope of the Wave form is
 $for 5\%$ to levance, Envelope of the Wave form is
 $for 5\%$ to levance, Envelope of the Wave form is
 $for 5\%$ to levance, Envelope of the Wave form is
 $for 5\%$ to levance, Envelope of the Wave form is
 $for 5\%$ to levance is $e^{-0.51\times 3}$
 $for 5\%$ to levance is $e^{-0.51\times 3}$

$$\frac{0.05}{1.006} = e^{-0.5} t_{\mu}$$
By tailing notional log to both Anides
$$\left(n\left(\frac{0.05}{1.006}\right) = \ln\left(e^{-0.5} t_{\mu}\right)\right)$$

$$-3.0017 = -0.5 t_{\mu}$$

$$t_{5} = \frac{-3.0017}{-0.5}$$

$$\therefore [t_{5} = 6.003] \text{ Reve}$$
155 Calculate the Static Enver Constants for the System, if transfer function $G(S) = \frac{10(S+2)}{S(S+3)(S+4)}$
Solution:
$$Given: G(S) = \frac{10(S+2)}{S(S+3)(S+4)}$$
U.14.T the position enverse constant $(K_{10}) = \frac{6}{S \to 0} G(S)H(S)$

$$K_{10} = \frac{10}{S \to 0} f_{10} S \to 0$$
W.K.T the Velouity Envers Constant $(K_{10}) = \frac{6}{S \to 0} S.G(S).H(S)$

$$I_{4,10} = \frac{10x2}{3x4}$$

$$I_{4,10} = \frac{10x2}{3x4}$$

W.14.T the Acceleration Error constant (4) = lim S² G(S)H(S)
If
$$a = \lim_{S \to 0} S^2 \cdot \frac{10 (S+2)}{s'(S+3)(S+4)}$$

If $a = 0$ for $s \to 0$
Iby Find IGP, IGV and IGa for a System having
G(S) = $\frac{S+10}{s(s^3+7s^2+12s)}$ also. Evaluate the steady state
Error, when the input sct is given by
if sct) = 5u(t)
if sct) = at utt)
if sct) = at utt)
if sct) = 4t²u(t)
Solution:
Gfven G(S)H(S) = $\frac{S+10}{s^2(s^2+7s+12)}$ with H(S) = 1
The given system is Type-2 and Order - 4 System
* \rightarrow The position 2 2000 (ons tant is given by

$$\begin{aligned} & \text{lim} \quad G(S)H(S) \\ & \text{lim} \quad \frac{S+10}{S^2(S^2+7S+12)} \\ & \text{lim} \quad \frac{S+0}{S^2(S^2+7S+12)} \\ & \text{lip} = \infty \quad \text{for} \quad S \to 0 \quad \longrightarrow 0 \end{aligned}$$

* The Velocity Error Constant is given by

$$|K_{1y} = \lim_{S \to 0} S \cdot G(S) H(S)$$

$$= \lim_{S \to 0} \frac{S}{S'(S^2 + TS + 12)}$$

$$|K_{1y} = \infty \quad \text{for } S \to 0 \quad \rightarrow (2)$$
* The Acceleration Error Constant is given by

$$K_{1a} = (\lim_{S \to 0} S^2 \cdot G(S) H(S),$$

$$|K_{1a} = (\lim_{S \to 0} \frac{S^2}{S'(S^2 + TS + 12)})$$

$$|K_{1a} = \frac{10}{12} = \frac{5}{6} \quad \text{for } S \to 0 \Rightarrow (3)$$
* The Steady state Error for an input

$$r(t) = Suitt$$
Given input is unit step

$$\therefore \text{ Steady state Error is given by}$$

$$C_{cs} = \frac{A}{1+Kp} \quad \text{for } strength of A$$

$$A = 5$$

$$\therefore C_{AA} = \frac{5}{1+Kp}$$

$$Note : From Eqn (1) Kp = \infty$$

* > Steady state Error for an input

$$r(t) = 2 \pm u(t)$$

Given input is ramp input $\{r(t) = A \pm u(t)\}$
 \therefore Steady state Error is given by $\frac{A}{160}$
But $A = 2$
 $C_{BS} = \frac{2}{K_{V}}$
 $C_{BS} = \frac{2}{K_{V}} = 0$ from Eqn (3) $K_{10} = \infty$
* -> Steady state Error for an input
 $r(t) = U \pm 2U(t)$
Given input is para bolic input $\{r(t) = A \pm 2 \}$
 \therefore Steady state Error is given by
 $C_{BS} = \frac{A}{16c}$ for strength of A
 $C_{BS} = \frac{4 + 2}{5}$
 $C_{BS} = \frac{4 + 12}{5}$
 $C_{BS} = \frac{4 + 12}{5}$
 $C_{BS} = 9.6$ from Eqn (3) $K_{10} = \frac{5}{6}$

17) Find the static Error Constants for the System
represented by leop transfer function
$$G(S)H(S) = \frac{10}{S(S+1)}$$

also determine the steady state Error of the System
when the Popul $T(C) = 10 + at$.
Colution:
 $G(Vere G(S)H(S) = \frac{10}{S(S+1)}$ for $H(S) = 1$
 $* \rightarrow The position Error Constant in given by
 $K_p = \binom{cm}{m} G(S) H(S)$
 $K_p = \binom{m}{m} S G(S) H(S)$
 $= \binom{m}{m} S' \frac{10}{g(S+1)}$
 $K_{10} = 10$ for $S \rightarrow 0 \rightarrow C$
 $* \rightarrow The Acceleration Error Constant in given by
 $K_a = \binom{m}{m} S^2 G(S) H(S)$
 $= (\frac{m}{m} S' - \frac{10}{g(S+1)})$
 $K_a = \binom{m}{m} S^2 G(S) H(S)$
 $= (\frac{m}{m} S^2 - \frac{10}{g(S+1)})$
 $K_a = \binom{m}{m} S^2 - \frac{10}{g(S+1)}$
 $K_b = 10$ for $S \rightarrow 0 \rightarrow C$$$

*
$$\rightarrow$$
 Steady state Error for an lique $r(t) = 10 + 2t$ is

$$C_{SS} = \frac{10}{1+16p} + \frac{2}{K_{15}}$$

$$C_{SS} = \frac{10}{1+\alpha} + \frac{2}{10} \quad \text{from (1) and (2)}$$

$$E_{SS} = \frac{10}{1+\alpha} + \frac{2}{10}$$

$$C_{SS} = 0 + 0.2$$

$$C_{SS} = 0 + 0.$$

Solution: The above block diagram is redracon as show
below.
if The Overall transfer (CS) is given by. E(S) 10
$\frac{C(S)}{E(S)} = \frac{G(S)}{1+G(S)H(S)} = \frac{\overline{s^2(s^2+S+10)}}{1+\frac{10}{s^2(s^2+S+10)}}$
$\frac{10}{E(S)} = \frac{\frac{5^4 + S^3 + 10S^2}{5^4 + S^3 + 10S^2 + 10S}}{\frac{5^4 + S^3 + 10S^2 + 10S}{-S^4 + S^3 + 10S^2}}$
$\frac{C(s)}{E(s)} = \frac{10}{s^4 + s^3 + 10s^2 + 10s}$ (5)
$\frac{+}{P_{1}(s)} = 1$ $\frac{+}{P_{1}(s)} = 1$ $\frac{+}{P_{1}(s)} = 1$ $\frac{+}{P_{1}(s)} = 1$
$\frac{C(s)}{R(s)} = \frac{10}{s(s^3 + s^2 + 10s + 10)}$
<u>C(S)</u> Represent Type-1 System R(S)
$G(s) H(s) = \frac{10}{s(s^3 + s^2 + 10s + 10)} \text{for } H(s) = 1$

is he know that "14p" position toros constant in given by. Kp = lim G(S) H(S) $hp = \lim_{s \to 0} \frac{10}{s(s^3 + s^2 + 10s + 10)}$ Kp= ∞ for s > 0 → 0 *-> we know that "Kin" yelowity Error constant Pre given by 190 = lim 5.900 H(S) $K_{10} = \lim_{s \to 0} \frac{s}{s} \cdot \frac{10}{s(s^3 + s^2 + 10s + 10)}$ * -> we know that "Ma" Acceleration Error Constant is given by. $K_a = \lim_{n \to \infty} s^2 \cdot G(s) H(s)$ = $\lim_{s \to 0} s^2 \cdot \frac{10}{\pounds (s^3 + s^2 + 10s + 10)}$ [4a=0] for s->0 -> (3) iii) Steady state 2000 for an input ret) = 10000) is. $C_{BB} = \frac{10}{1 + 14p} = \frac{10}{1 + \infty} \quad \text{from } E_{qr} (1).$ Kip = 00 CAR = 10 = 0

$$l_{B,E} = \frac{5}{\infty} + \frac{5}{14}$$

$$l_{A,B} = 0 + \frac{5}{16}$$

$$l_{A,B} = \frac{5}{14}$$

$$l_{B,E} = \frac{5}{14}$$

$$l_{B,E} = \frac{5}{14}$$

$$l_{B,E} = \frac{5}{14}$$

$$l_{B,E} = 0.1$$

$$l_{B,E} = 50$$

$$Mer insum Value 2, K, 1650$$

$$loy For a Signal 4 low graphs khows for the figure
Mention type number, and Order of the bystem
and determine the stocky stake Error for step
and ramp (aput. Cet) = sciD-y(4)
$$\frac{-2}{16}$$

$$l_{B,E} = \frac{-2}{(S+4)} (1, 2, 3, 4, 5)$$

$$R_{E} = \frac{-3}{(S+4)} (1, 3, 4, 5)$$$$

* Single loop gains.

$$P_{11} = \frac{-24}{(s+3)(s+4)(s+5)}$$

$$EP_{m_{2}} \text{ and } \theta_{2} \text{ Wards is } (0)$$
* (o-factors of graph?

$$\Delta_{1} = 1 - 0 = 1$$

$$\Delta_{2} = 1 - 0 = 1$$

$$\frac{Y(s)}{R(s)} = \frac{\prod_{k=1}^{n=2} P_{k_{1}} \Delta_{k_{1}}}{1 - \sum_{m=1}^{l} P_{m_{1}} + 0}$$

$$\frac{Y(s)}{R(s)} = \frac{P_{1}\Delta_{1} + P_{2} \Delta_{2}}{1 - P_{11}}$$

$$\frac{Y(s)}{R(s)} = \frac{\frac{12}{(s+3)(s+4)} - \frac{3}{(s+4)}}{1 + \frac{24}{(s+3)(s+4)(s+5)}}$$

$$\frac{Y(s)}{R(s)} = \frac{\frac{12 - 3(s+3)}{(s+3)(s+4)(s+5) + 24}}{(s+3)(s+4)(s+5) + 24}$$

$$\frac{Y(s)}{R(s)} = \frac{12 - 3s + 9(s+5)}{(s+3)(s+4)(s+5) + 24}$$

$$\frac{Y(s)}{R(s)} = \frac{12 - 3s + 9(s+5)}{(s+3)(s+4)(s+5) + 24}$$

$$\frac{Y(s)}{R(s)} = \frac{12 - 3s + 9(s+5)}{(s+3)(s+4)(s+5) + 24}$$

* The Proves Signal is given by

$$e(t) = v(t) - y(t)$$

 ore
 $E(s) = R(s) - y(s)$
 $E(s) = R(s) - \frac{(-3s+3)(s+5)}{(s+3)(s+4)(s+5)+24} - R(s)$
 $E(s) = R(s) \left[1 + \frac{(3s-3)(s+5)}{(s+3)(s+4)(s+5)+24}\right]$
 $E(s) = R(s) \left[\frac{(s+3)(s+4)(s+5)+24}{(s+3)(s+4)(s+5)+24}\right]$

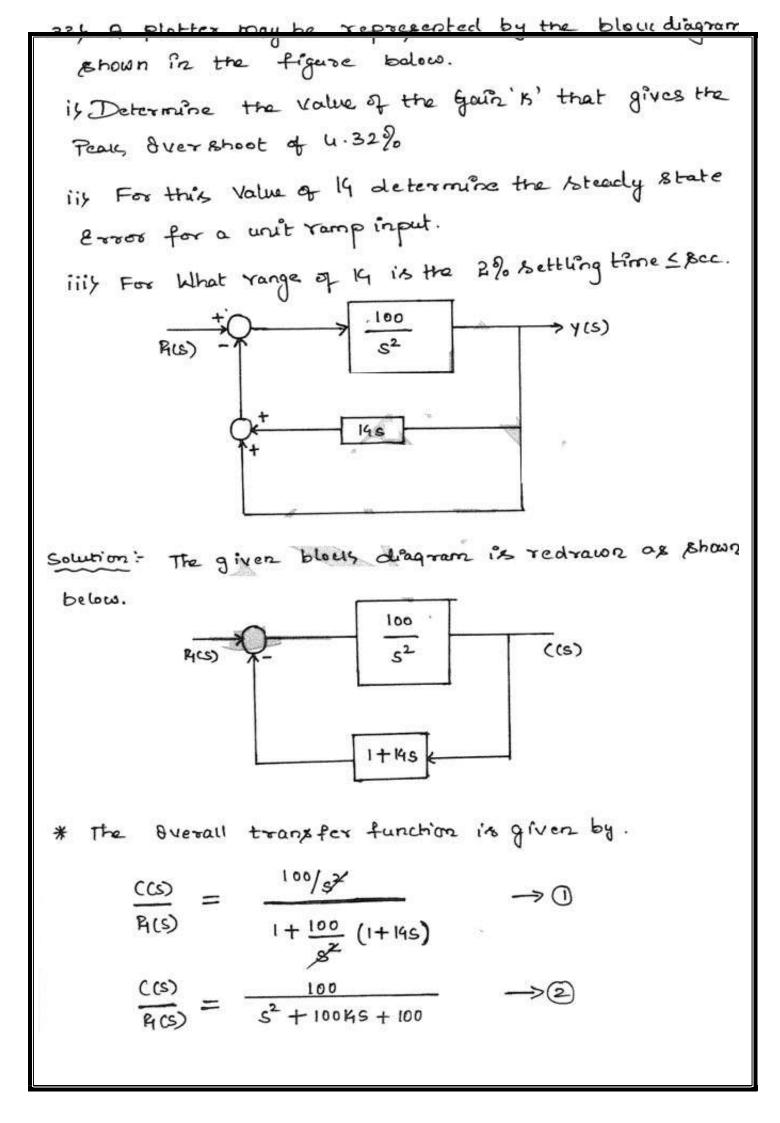
W.14.T

$$\therefore c_{kk} = \lim_{s \to 0} g(1) \left[\frac{(s+3)(s+4)(s+5) + 24 + (3s-3)(s+5)}{(s+3)(s+4)(s+5) + 24} \right]$$

$$\frac{(3 \times 4 \times 5) + 24 + (-3)(5)}{(3 \times 4 \times 5) + 24} = \frac{69}{84}$$

Steady state Error for a unit ramp if
$$P(S) = \frac{1}{S^2}$$

 $e_{\beta,\beta} = \lim_{S \to 0} S \cdot E(S)$
 $\therefore C_{\beta,\beta} = \lim_{S \to 0} S \cdot \frac{1}{S^2} \left[\frac{(S+3)(S+4)(S+S) + 24 + (3S-3)(S+S)}{(S+3)(S+4)(S+S) + 24} \right]$
 $e_{\beta,\beta} = \lim_{S \to 0} \frac{1}{S} \left[\frac{(S+3)(S+4)(S+S) + 24 + (3S-3)(S+S)}{(S+3)(S+4)(S+S) + 24} \right]$
 $e_{\beta,\beta} = \frac{1}{S \to 0} \text{ for } S \Rightarrow 0;$
214 A unity feed back (solvest system has
 $G(S) = \frac{14}{S(S+2)(S^2+2S+5)}$
(i) For a unit ramp input, it is degired $e_{\beta,\beta} \le 0.2$,
find K.
(11) Determine $e_{\beta,\beta}$ if input $r(t) = 2 + 4t + \frac{12}{2}$
Solution:
(ii) The velocity lenges (solvest) is given by
 $K_{14} = \lim_{S \to 0} S \cdot \frac{14}{g(S+2)(S^2+2S+5)}$
 $K_{19} = \frac{K_{1}}{S \cdot 5}$
 $K_{19} = \frac{K_{1}}{S \cdot 5}$



Comparing
$$S^{2} + 100 \text{ With } S^{2} + 28 \text{ Wn } S + Wn^{2}$$

 $Un^{2} = 100$
 $U_{n} = \sqrt{100}$
 $\boxed{U_{0} = 10}$
 $2.5 \text{ Un} = 100 \text{ H}$
 $\text{H} = \frac{2.5 \text{ Wn}}{100}$
 $\boxed{H^{2} = \frac{5 \text{ Wn}}{100}}$
 $\boxed{H^{2} = \frac{5 \text{ Wn}}{100}}$
 $griver grids mp = u \cdot 32.5\%$
 $g_{0} \text{ Mp} = e^{-5\pi/\sqrt{1-5}2\omega}$
 $g_{0} \text{ Mp} = e^{-5\pi/\sqrt{1-5}2\omega}$
 $(u \cdot 3a) = e^{-5\pi/\sqrt{1-5}2\omega}$
 $tata^{2}ng natural log oraboth & didex.$
 $ln\left(\frac{U \cdot 32}{100}\right) = ln\left(e^{-5\pi/\sqrt{1-5}2\omega}\right)$
 $-3.14.19 = -\frac{5\pi}{\sqrt{1-5}2\omega}$
 $gy Tatuág Squars on both & dides.$
 $(-3.14.19)^{2} = \left(-\frac{5\pi}{\sqrt{1-5}2}\right)^{2}$
 $9 \cdot 8716 = \frac{5^{2}\pi^{2}}{1-5^{2}}$

 $9.8716 (1-6^{2}) = 6^{2} \Pi^{2}$ $9.8716 - 9.87166^{2} = 6^{2} \Pi^{2}$ $9.8716 - 9.87166^{2} = 9.86966^{2}$ $9.8716 = 9.86966^{2} + 9.87166^{2}$ $9.8716 = 19.74126^{2}$ $8^{2} = \frac{9.8716}{19.7412}$ $8^{2} = \frac{9.8716}{19.7412}$ $8 = \sqrt{\frac{9.8716}{19.7412}}$ $8 = \sqrt{\frac{9.8716}{19.7412}}$

We Know that

$$4 = \frac{4W_0}{50}$$

$$14 = 0.707 *10$$

$$50$$

$$14 = 0.1414$$

il's To find velocity Error Constant

W.14.T. 140 = lim S G (S) H(S) S>0

From Equations (1) W.K.T $G(S)H(S) = \frac{100}{S^2}(1+148)$ $\therefore 140 = \lim_{s \to 0} g. \frac{100}{S^2}(1+0.1414S)$ $\therefore 140 = 0$ for $s \to 0$

235 The Open loop transfer function of a lootrol hyster?
With a Unity field back is
$$G(S) = \frac{10}{S(1+0.15)}$$
. Evaluate
Errors Beriax for the hystern. Determine the
Actually state Error of the hystern with the Population
 $r(t) = 1+2t+t^2$
Solution:
The Errors transfer function is given by.

$$\frac{E(S)}{R(S)} = \frac{1}{1+G(S)H(S)}$$

$$= \frac{1}{1+\frac{10}{S(1+0.15)}}$$

$$= \frac{1}{3(1+0.15)+10}$$

$$\frac{E(S)}{R(S)} = \frac{5(1+0.15)}{0.15^2+5+10}$$

$$= \frac{0.15^2+5}{0.15^2+5+10}$$

$$= \frac{0.15^2+5}{0.1(5^2+105)+10} \rightarrow 1$$

$$C_0 = \lim_{s \to 0} \frac{1}{1+G(S)H(S)} = \frac{S^2+10S}{s^2+105+100}$$

$$E(S) = \frac{1}{1+G(S)H(S)} = \frac{S^2+10S}{s^2+105+100}$$

$$\begin{aligned} \overset{\text{J.H-I}}{C_{1}} &= \bigcup_{s \to 0}^{lm} \frac{d}{ds} \left(\frac{1}{|1+l_{9}(s)|H(s)} \right) \\ C_{2} &= \frac{1}{d!} \bigcup_{s \to 0}^{lm} \frac{d^{2}}{ds^{2}} \left(\frac{1}{|1+l_{9}(s)|H(s)} \right) \\ \Rightarrow \frac{d}{ds} \left(\frac{1}{|1+l_{9}(s)|H(s)} \right) &= \frac{d}{ds} \left(\frac{s^{2} + 10s}{s^{2} + 10s + 100} \right) \\ &= \frac{(s^{2} + 10s + 100)(2s + 10) - (s^{2} + 10s)(2s + 10)}{(s^{2} + 10s + 100)^{2-}} \\ &= \frac{(2s + 10)(s^{2} + 10s + 100)^{2-}}{(s^{2} + 10s + 100)^{2-}} \\ \frac{d}{ds} \left(\frac{1}{(1+l_{9}(s)|H(s)} \right) &= \frac{100(2s + 10)}{(s^{2} + 10s + 100)^{2-}} \rightarrow \textcircled{3} \\ \\ \text{LIX-T} C_{1} &= \frac{lm}{s \to 0} \frac{100(2s + 10)}{(s^{2} + 10s + 100)^{2-}} \\ C_{1} &= 100 + \frac{10}{(s0^{2} + 10s + 100)^{2-}} \\ &= \frac{1}{10} \\ \hline \left(\frac{C_{1} = 0.1}{c_{1}} \right) \\ &\Rightarrow \frac{d^{2}}{ds^{2}} \left(\frac{1}{(1+l_{9}(s)|H(s)} \right) &= \frac{d}{ds} \left\{ \frac{d}{ds} \left(\frac{1}{(1+l_{9}(s)|H(s)} \right) \right\} \\ &= \frac{d}{ds} \left\{ \frac{100(as + 10)}{(s^{2} + 10s + 100)^{2}} \right\} \end{aligned}$$

$$\frac{d^{2}}{ds^{2}} \left(\frac{1}{1+c_{9}(s)H(s)} \right) = \frac{100}{\left(\frac{(s^{2}+10s+100)^{2}(2) - (2s+10) \cdot 2(s^{2}+10s+100)(2s+10)}{(s^{2}+10s+100)^{4}2} \right)}$$
$$\frac{d^{2}}{ds^{2}} \left(\frac{1}{1+c_{9}(s)H(s)} \right) = \frac{200 \left[s^{2}+10s+100 - (2s+10)^{2} \right]}{(s^{2}+10s+100)^{2}}$$

W. 4.T

$$C_{2} = \frac{1}{21} \quad \lim_{S \to 0} \frac{d^{2}}{ds^{2}} \left(\frac{1}{1 + 4(s) + 4s} \right)$$

$$C_{2} = \frac{1}{2} \quad \lim_{S \to 0} \frac{200 \left[S^{2} + 10S + 100 - 4S^{2} - 40S - 100 \right]}{(S^{2} + 10S + 100)^{2}}$$

$$C_{2} = \frac{1}{2} \cdot \left\{ \frac{200 \left[100 - 100 \right]}{(t00)^{2}} \right\} \quad \text{for } s = 0$$

$$C_{2} = 0$$

Error peries its given by:

$$e(t) = C_0 * ct) + C_1 *'(t) + C_2 *''(t) + \cdots + \cdots$$

$$r(t) = 1 + 2t + t^2$$

$$r'(t) = 2 + 2t$$

$$r''(t) = 2$$

$$r'''(t) = 0$$

$$r'''(t) = 0$$

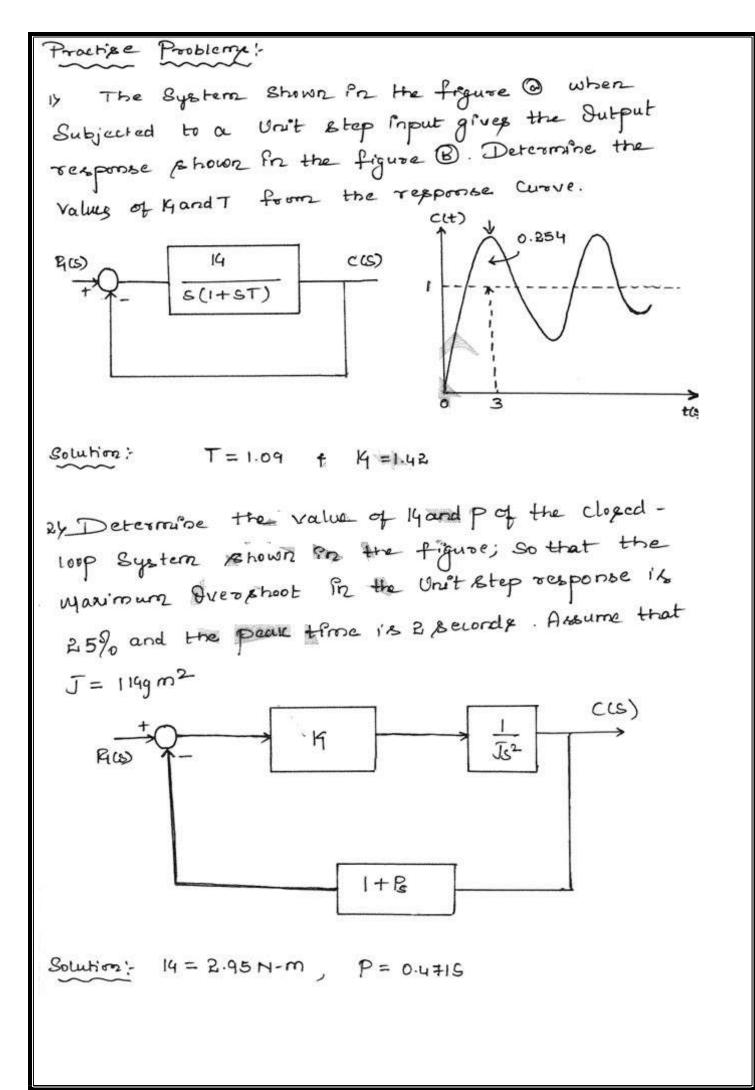
$$\therefore e(t) = 0 + 0 \cdot 1 (2 + 2t) + 0$$

$$e(t) = 0 \cdot 2 + 0 \cdot 2t$$

$$e(t) = 0 \cdot 2 [1 + t]$$

$$\therefore SSE = e_{\beta,\beta} = \lim_{t \to \infty} e(t)$$

$$e(t) = 0$$



3) For the Control Systems with Open-loop transfer
functions given balow, Explain what type of Input
Signal gives Tipes to a Constant steady - state
Error and Calculate there values.
(a)
$$G(s) = \frac{20}{(s+1)(s+4)}$$
 (b) $G(s) = \frac{10(s+4)}{s(s+1)(s+2)}$ (c) $G(s) = \frac{20}{s^2(s+1)(s+4)}$
Solution: as type -0 system
 $esc = \frac{1}{5}$
by type -1 system
 $esc = 0.85$
C5 type -2 System
 $esc = 0.85$
(c) type -2 System
 $esc = 0.78$
U) Consider a unity feed bouk System with a closed
transfer function $\frac{C(s)}{R(s)} = \frac{Ks+5}{s^2+as+5}$
Determine the Open-loop transfer function $G(s)$.
Show that the steady state Error with Unit ramp
input is given by $\frac{a-K}{5}$
Solution: $G(s) = \frac{Ks+5}{s(s+(a-K))}$
Gfvez unit nomp input. : $K_{12} = \frac{Ks}{s-20} = \frac{b}{a-K}$

8% The open loop transfer function of a feed back
Control System in given by

$$G(S) H(S) = \frac{K(S+1)}{S(1+KT)(1+25)}$$

Determine the Error Co-structurents and Error
due to the Unit positional input, Unit ramp Unit,
and Unit parabolic input, it K=10 and T=4;
Solution: $Kp = \infty$, $Kv = 10$, $Ka = 0$;
is Unit step input = $\frac{1}{1+Kp} = 0$
is Unit step input = $\frac{1}{1+Kp} = 0$
is Unit ramp input = $\frac{1}{1Ka} = \infty$
9% Find all the three domain Specifications for a Unity
feedback control System: Whose Open-toop transfer
function: $Wn = 5 \operatorname{rad/Sec}$; $Wd = 4 \operatorname{rad/Se}$; tr = 0.55 pccs,
 $tp = 0.785$ pccs, ghtp = 9.5%, Mp = 0.0947, ts = 1.23;
10% A Unity feed back Serve - driven instrument
has an Open-loop transfer function $G(S) = \frac{10}{S(S+2)}$
find,
a) The time domains for a Unit
step input.
by The natural frequency of Oscillations (won) and
damping-ratio (%).

© Maximum Overschoot and the peak time.
(a) Steady - State Error to an input (1+4t)
Solution:
(b)
$$\Rightarrow$$
 CC+D = 1 - 1.05e^t sin (3t + 71.34°)
(b) \Rightarrow Wn = 3.16 rad/sec, $g = 0.32$
(c) \Rightarrow Mp = 0.3460; $\%$ Mp = 34.60%; tp = 1.04 sec;
(a) \Rightarrow $e_{ss} = \lim_{s \to 0} s \in CS$) = 0.8

Assignment problems :-14 A closed - 100p (ontrol system is represented by the differential Equation. $\frac{d^2 c}{dt^2} + 4 \frac{d c}{dt} = 16 e$ Where e= - c is the Error Signal. Determine the undamped natural frequency, damping ratio, and Percentage maaimum overshoot for a unit step input. 24 A unity feed balls system has an Open-loop transfer G(S) = 5/(S(S+1)), Find the rise time, function Perceptage Overschoot, peak fine and Settling time for a step input of 10 units, also determine the Fran Quesphoot 34 A unity freed backs system is characterised by the Open - loop toansfor function. Determine the steady state Errors for Unit step, Unit ramp and Unit-acceleration input. 4% The Open loop transfer function of a unity feedback System is given by $G(S) = \frac{14}{S(1+ST)}$, where T and 4 are constants having positive values, By what factor the amplifier gain be reduced so that @ The peak Due school of unit step response of the

System is reduced from \$5% to 25%
(b) The damping ratio increases from 0.1 to 0.6.
5% For a Unity feed backs system whose Open-loop
transfers function is
$$G(S) = \frac{50}{(1+0.1S)(1+2S)}$$
, find the
Position, Velocity and acceleration for Constants.
6% Determine the force Co-2thicients and static forces
for Unity and non unity feed back system.
 $G(S) = \frac{1}{S(S+1)(S+10)}$; H(S) = S+2;

Hy A Certain feed bould Constrol bystem its described. by the following transfer function

$$G(S) = \frac{14}{s^2(s+20)(s+30)}$$
; H(S)=1;

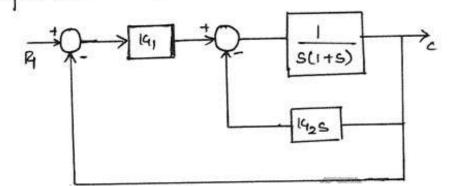
Determine steady state Errors Co-Ebbients and also determine the value of '14' to limit the Borors to counit, due to input.

By Determine the position, velouity and acceleration Error Constants for a unity feed back Control System Whose Open loop transfer function given by

$$g(s) = \frac{14}{s(s+4)(s+10)}$$

It 14=400, determine the Steady-state for for a unit ramp input.

94 A feed back system Employing Output rate damping is snown in figure at Find value of K1 and K2. So that Clope - loop system repembles a second Order System witz damping ratio Equal to 0.5 and frequency of damped Oscillations 9.5. rad/sec.



by with the above value of 19, and 192 find the percentage overproof when input is step input

ic's himat is the setting time for 2 percent to levance.

10% A second Order pervo system has unity feedback and an open loop transfer function.

500 G(s) =

ay Draw a blour diagram for the cloped loop transfer function.

by What is the characteristic Equation of the Bystem.

cy What is the value of natural frequency (Wr) and damping ratio (y).

de skietch the transient response for a unit step inpu

ey Obtain the value of percentage Overshoot and the Peak time.

fy What is the Setting time of the Rystem.

9) If the System 18 Subjected to a ramp input of 0.5 rad/s; what is the steady state brook? Introduction to PI, PD, and PID Controller

P- Controller:

- * The proportional Controller is a device that produces an Output Signal which is proportional to the input Signal.
- * The propositional Controller inoproves the steady state tracking accuracy, disturbance beignal rejection and, relative stability. It also decreases Scraibivity of the system to parameter Variations.
- * The disadvantage of a propositional Costroller is that it produces a constant steady state Error.

PI-Costooller:

- * The PI-Controller is a device that produces and Output Bignal Consisting of two terms - one. Proportional to input Bignal and the Other proportional to integral of input Bignal.
- * The Effect of a PI Cootmoller On the System performance is that it increases the Order of the System by One, which regults in the reductions of the Steady-state Error. But the System becomes less stable than the Original One.

* The transfer function of PI controller is given by 14p + 14° G (S) =

PD- Controllers:

- * The PD Controllers is a device that produces an Output Rignal Consisting of two terms - One propositional to Proput Bignal and the Other propositional to derivative of the Proput Signal.
- * The PD Costrollers Preseaged the damping of the System which regults is reducing the peace Overschool.
- * The Effect of PD Costrollers On the System Performance its to Prevenge the damping ratio of the System and so the peak Overshoot its reduced.
- * The transfers function of PD Cootrollers ite givenby

PID-Cootroller:

The PID Controllers is a device. which produces an Output Bignal Consisting of three terms - one.
Propositional to Consisting of three terms - one.
Propositional to Consisting al another one proportional to integral of Coput Bignal and the third one propositional to desirvative of Coput Bignal.
* The PID - Controllers Stabilizes the gain, reduces the Steady State Error and peak Overshoot of the Bysters. The transfer function for PID Controllers 18

given by Gc (s) = Kp + Kils + Kas

Theory queptions !-

- 1) Inlitte a usual notations. Devive an Expression for a unit step response of an underdamped second Order system.
- is Considering the response of a Second Order Underdampe System to a step Popul, derive the following, is peak time is Rise time is Maximum peak Oversmoot is setting time.
- 34 Drows the time repponse Lurve and Define the time domain specifications for a second Orders Control bystem for a unit step input
- by Define the following terms Transient reponse B Steady state response.
- 5) Devive the Output ockponse for First Order System with a relevant figures.
- by worke a cohoost note on ay PI Controller.
 - by PD Controller.
 - CY PID Cootroller.